



STATISTICAL ENERGY ANALYSIS  
COMPUTER PROGRAM

USER'S GUIDE

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## PREFACE

The work reported herein was performed for the National Aeronautics and Space Administrations's Marshall Space Flight Center under contract NAS8-33191, Supplemental Agreements 2 and 3. This effort involved the development of a computer program to perform Statistical Energy Analysis. This volume constitutes the final deliverable item under the contract. A card deck of the computer program was forwarded previously by MDAC letter A3-130-GWJ-1283.

Programming was accomplished by S. J. Nygaard, McDonnell Douglas Automation Company - Huntington Beach.

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## Section 1 INTRODUCTION

Significant high frequency random vibration environments are generated during the operation of aerospace vehicles within the atmosphere. To achieve optimum vehicle design, vibration and acoustic criteria are developed early in the vehicle development program and are updated periodically as the design matures.

High frequency random vibration response prediction does not lend itself well to classical structural dynamics and such predictions are usually made by extrapolation from existing data banks. This method gives excellent results when similar structures are involved. However, as similarity decreases and associated extrapolations become large, uncertainty over accuracy also increases.

The efforts of numerous investigators (Maidanik, Lyons, et al.) have examined a more general high frequency random vibration analysis approach - the so-called Statistical Energy Analysis method. The SEA method is able to accomplish high frequency prediction of arbitrary structural configurations and is therefore a significant improvement over extrapolation methods when little or no previous data exist. The SEA method also represents a great improvement over normal mode methods for high frequency random vibration prediction because of the greatly reduced computational complexity associated with the more general SEA model elements and the attendant improvement in analysis turnaround time.

SEA has been developed for complex structures by MDAC under contract to MSFC (Ref. 1 and 2) over the past three years. The past year's effort has created a general SEA computer program which is described in this manual. The manual contains (1) a summary of SEA theory, (2) example problems of SEA program application, (3) a description of the computer program, and (4) a complete program listing (Appendices).

## Section 2

### A SUMMARY OF STATISTICAL ENERGY ANALYSIS PRINCIPLES

Statistical Energy Analysis (SEA) is a powerful tool for estimating the high frequency vibration spectra of complex systems. The analysis method is based on the estimation of the power flow between idealized gross elements of a vibrating system. The method is statistical in that averaging assumptions are made with regard to distribution of energy within an element, distribution of resonant modes, and the coupling between elements. These assumptions greatly simplify the computational complexity associated with normal mode methods. These same assumptions impose the limitation that point response predictions cannot be made.

The assumptions on which the method rests and their implications can be quite rigorously stated as follows:

1. The total vibrating system can be partitioned into SEA elements (with suitable boundary conditions) whose modes approximate the modes of the original vibrating system.
2. The modes of the elements of a system contain all of the vibratory energy of the system.
3. The energy in one frequency band of a system element is equally distributed among the modes of that element occurring in the frequency band.
4. Only modes occurring within the same frequency band are coupled.
5. For two coupled elements, all of the modes occurring in one of the elements in one frequency band are equally coupled to each mode occurring in the same frequency band in the other element.

Assumption 1 contains the fundamental existence basis for SEA: the concept of partitionability. This concept implies that a coupled vibrating system with system modes can be approximated by two or more separately idealized vibrating elements, each with its own independent mode set. These sets are coupled only in the sense of having power flow to and from each set across the partition boundary (later referred to as the "joint"). The approximation to this model exists in most structures having reflective boundaries in the higher frequencies. For example, a skin/stringer structure has higher order skin panel modes that are nearly the same frequency and shape as an ideally supported panel because the stringer is a comparatively massive boundary causing reflection of flexural waves from the skin panel. An SEA plate element could logically be equal to the panel area bounded by stringers or frames. Such elements will then have to be coupled with joint elements in order to develop an SEA model which emulates the vibratory power flow of the real structure.

Assumption 3 is the most important simplifying assumption of SEA because it eliminates the necessity to calculate generalized modal forces and responses. The conditions implicit in this assumption are usually approximated by the higher order modes of a structure in a reasonable bandwidth, say 1/3 octave. One-third octave bands represent a reasonable compromise between the necessity to get a fairly large number of modes (>20) in the band for good statistics and the necessity to have some frequency response resolution in the vibration prediction. The number of modes in a unit bandwidth can be estimated for simple structural forms (such as beams, plates, etc.) using algebraic expressions for modal density such as those given in Section 4 of this report. Estimation of modal density in this way is a considerable simplification over normal mode methods.

Given SEA elements with the properties described above it is now necessary to join them to permit power flow between the modes of one element and the modes of another. This is done with a parameter called the coupling loss factor  $\phi$  and leads to assumptions 4 and 5. Assumption 4 is directly linked to assumption 2 and the further assumption of a linear process. Assumption 5 follows directly from assumption 3 as part of the simplification associated with a statistical rather than explicit description of modes.



With these properties and assumptions we can now write the SEA equation systems as follows:

$$\begin{bmatrix} \alpha_{11} & \alpha_{12} & \alpha_{13} & \dots & \alpha_{1j} \\ \alpha_{21} & \alpha_{22} & \alpha_{23} & \dots & \alpha_{2j} \\ \alpha_{31} & \alpha_{32} & \alpha_{33} & \dots & \alpha_{3j} \\ \vdots & \vdots & \vdots & \ddots & \vdots \\ \alpha_{i1} & \alpha_{i2} & \alpha_{i3} & \dots & \alpha_{iM} \end{bmatrix} \begin{bmatrix} E_1 \\ E_2 \\ E_3 \\ \vdots \\ E_i \end{bmatrix} = \begin{bmatrix} S_1 \\ S_2 \\ S_3 \\ \vdots \\ S_i \end{bmatrix} \quad (1)$$

where

M is the number of SEA elements

$$\alpha_{ij} = \begin{cases} -N_i \phi_{ij} & i \neq j \\ \omega \eta_j + \sum_{k=1}^M N_k \phi_{ik} & i = j \end{cases}$$

$N_i$  = number of modes resonant in element i

$\eta_j$  = element i loss factor

$\phi_{ij}$  = power transfer coefficient for coupling between modes in elements i and j

$\omega$  = center frequency of bandwidth

$E_i = m_i \frac{\langle \bar{a}_i^2 \rangle}{\omega^2}$  = total energy of element i,  $\langle \bar{a}^2 \rangle$  being the mean square acceleration

$S_i$  = external acoustic or mechanical excitation in the bandwidth of interest

Note that the matrix  $\alpha$  is square but not symmetric. The lack of symmetry arises from the nature of the term  $-N_i \phi_{ij}$ . Using the first two rows as an example the power balance equations are

$$(\omega \eta_1 + \sum_{k=1}^M N_k \phi_{1k}) E_1 - N_1 \phi_{12} E_2 - N_1 \sum_{k=3}^M \phi_{1k} E_k = S_1$$

$$-N_2 \phi_{12} + (\omega \eta_2 + \sum_{k=1}^M N_k \phi_{2k}) E_2 - N_2 \sum_{k=3}^M \phi_{2k} E_k = S_2$$

Note that the "symmetric" positions  $\alpha_{21}$  and  $\alpha_{12}$  actually carry the number of modes belonging to the row number only. This unsymmetric form preserves the power flow both to and from each element.

Each equation in the matrix states the following relationship for each element. The net power flow into element  $i$  ( $S_i$ ) equals the difference between (1) the power flow dissipated within the element ( $\omega n_i E_i$ ) plus the power flow lost to other elements across the joints

$$E_i \sum_{k=1}^M N_k \phi_{ik}$$

and (2) the power flow added to element  $i$  from all other elements

$$N_i \sum_{k=1}^M E_k \phi_{ik}, \quad i \neq k$$

It must be remembered that the coupling term  $\phi_{ik}$  has nothing to do with coupling modes; it only relates the fractional amount of energy resident in the modes of element  $i$  that flows to the modes of element  $k$ .

For all but the simplest of systems even the SEA equations can be laborious to evaluate by hand. The computer program described in the following sections performs all of the computations necessary to evaluate terms and solves the above system of equations for element energies and prints the results as vibration PSD or RMS levels for all elements. This program relieves the analyst of the necessity to have an extensive knowledge of how to compute SEA parameters and permits the use of SEA model sizes that would otherwise be intractable. The program is described in detail in Section 2; some SEA modeling examples and corresponding program inputs are described in Section 3.

### Section 3

#### EXAMPLES OF SEA PROGRAM APPLICATIONS

An SEA computer program has been developed that performs the organization of specific problem solutions for up to a 20-element system. A number of different element and joint types are available to describe a wide variety of structural forms in terms of an SEA model. Random acoustic or mechanical excitation can be applied in 1/3 octave bands to any arbitrary number of elements within a given model. The resulting equations are then solved giving the vibration response spectrum for each element. This section illustrates the use of SEA and the computer program with specific examples. The basic analysis procedure is divided into three steps:

1. Idealization
2. Parameter Generation
3. Problem Solution

The idealization step must be performed entirely by the user as it consists of modeling the physical structure in terms of available SEA program elements - a conceptual process. The second step is one of simply providing the proper data to the program which then carries out the third step. These steps will be illustrated below with specific examples, and the SEA program elements will be discussed in detail.

The idealization step is by far the most crucial step in the process. It is here that the art of engineering judgement must reach a well developed state, balancing the realities of the structural article to be analyzed with the capabilities and assumptions implicit in the SEA process to

obtain a useful engineering solution. Consider the model shown in Figure 3-1. This is the basic form of all SEA models. It consists of elements, denoted as boxes, which may be plates, beams, etc., and joints denoted by connecting lines which correspond to the physical interfaces at the selected partitions.

The model shows the articles included in each SEA element and the connection relationships between or among the elements. To gain a better understanding of what these elements and connections are it is helpful to now explain in detail the various types available in the computer program. Each of the elements is made up of one or more sub-elements, the first of which is the main sub-element. The sub-element system provides a convenient way to compute and include the modal density and mass properties of individual structural pieces that make up the element on a piece-by-piece basis. However, the sub-element with the most important property being modelled in that SEA element should always be the first or main sub-element. The joint properties must also be consistent with this sub-element as it is the only one that can be coupled to other SEA main elements. An example is given by element 1 of the model. This element has structure elements exposed to an acoustic field and those that are not. The most important property is the reception of acoustic excitation and transmission of vibration through its boundaries to other main elements. That portion must therefore be sub-element 1.

It is often the case that structures are made of different materials; this condition is taken into account when the program sets up the solution. The program automatically recomputes properties (thickness, density, etc.) to match the elastic modulus of element 1, sub-element 1. Furthermore, only sub-element 1 of each element can receive acoustic excitation; the others contribute only to the modal density and to the element mass. Sub-element 1 will always be used for the structure element that is the main piece of the given SEA element because all other properties except mass and modal density will be those of this sub-element.

The types of sub-elements available are beams, plates, cylinders, membrane, and reverberant room. In general, beams, plates and cylinders can

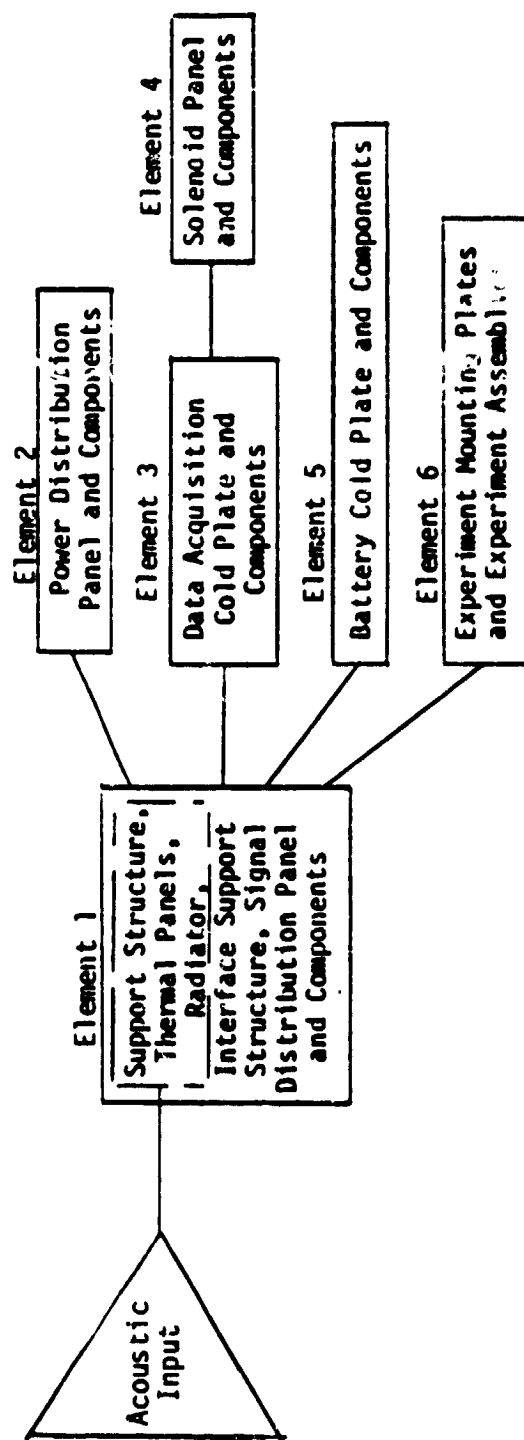


Figure 3-1. SEA Model Elements for Acoustic Test Configuration

be freely mixed in describing a structure. The membrane sub-element should only be used as an ancillary sub-element (i.e., sub-element number > 1). The room sub-element requires special treatment in that the loss factor must be developed from the reverberation time ( $n = 2.2/fT_{60}$ ) and the answers must be converted from  $q^2$  to pressure<sup>2</sup> using energy density relationships. The room sub-element must always be the only sub-element in that particular element.

Given these SEA elements, joint properties must now be developed to describe the power flow from the modes of one element to another. The program gives the user a choice of four types: (1) plate to plate, (2) beam to plate, (3) bolted joint, and (4) plate to acoustic. Each joint has two ends for accounting purposes and the word order used describes the A and B ends respectively. For example, in joint (2) the beam is always at the A end and plate is always at the B end. The joint loss factors are from the literature (cited in References 1 and 2) except for the bolted joint. This joint is a plate-to-plate joint with an additional insertion loss to account for internal losses in the joint due to fastener effects. The insertion loss is a load sheet input which provides to the user a more general purpose alternative to joints (1) and (2). Values of the insertion loss parameter for various fastener arrangements are not well defined, however, and must be the topic of continuing research. In the case of the Materials Experiment Assembly (MEA) analyzed during the last phase of this study, the empirically determined insertion loss factor was approximately 10 for each bolted joint.

With the basics of the program now given, two examples will be shown as a guide for program use. The first example will be a segment of skin stringer structure exposed to an acoustic field with an equipment panel on the opposite side. The second example will be a simplified MEA analysis.

The first case can be idealized with two SEA elements as shown in Figure 3-2. The relevant parameters are shown in the following table:

Skin Thickness	.040"
Segment Dimensions	18" x 72"
Stringer Spacing	10" O.C.
Stringer Dimensions	1-1/2" high x 1" wide with 3/4
Internal Frames	flanges, .063 thick (full hat section)
Panel	400 in <sup>2</sup> of .063 thick aluminum
	1" honeycomb with .020 face sheets
	17 x 30 with 16 lb of small equip-
	ment items mounted on its surface.
	Panel riveted to frames at four
	places (U-shaped channels) .063"
	thick x 1-1/2" high

The SEA element 1 shown in Figure 3-2 will consist of all structural elements in the table except the panel and its mounted equipment which is SEA element 2. From this information an elementary SEA analysis can be made. The load sheet entries are determined as follows and are shown in Figures 3-3 through 3-10 in the proper sequence.

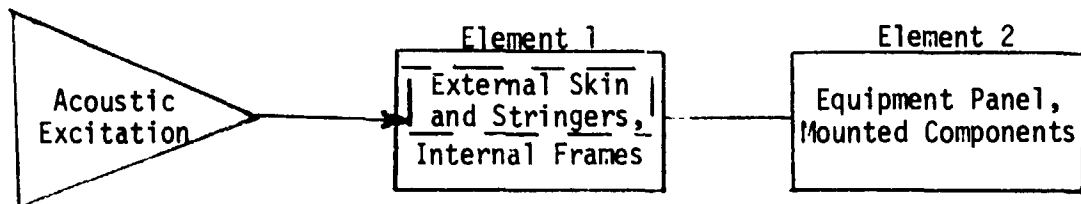


Figure 3-2. Two-Element SEA Model Example

Figure 3-3 shows the header card which basically describes the problem as consisting of two elements and eleven 1/3 octave bands from 250 to 2500 Hz and that the output will be a vibration PSD. Figure 3-4 describes element 1 as consisting of two sub-elements, the first of which is exposed to acoustic excitation, and describes the damping vs. frequency curve. Figure 3-5 shows the 1/3 octave sound pressure levels; Figures 3-6 and 3-7 describe the sub-element properties in detail.

- One required per case
- First card in sequence

1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18
2	1	1								2	5	0	.	0			P.S.D

<u>Card Columns</u>	<u>Value</u>	<u>Definition</u>
1-2	Integer	Total number of elements in model ( $2 \leq E \leq 20$ )
3-4	Integer	Total number of analysis frequencies ( $1 \leq F \leq 40$ ) (Analysis frequencies are spaced 1/3 octave apart)
5-14	Real	Lowest analysis frequency
15	Alpha	Units; M = metric (MKS), default = English (in, lb, sec)
16-18	Alpha	Output mode; RMS or PSD

**Figure 3-3. Card 1, Example Case 1**



- One required for each element ( $2 \leq E \leq 20$ )
- Must be in ascending numerical sequence

[illegible]

### Definition

<u>Card Column</u>	<u>Value</u>	<u>Definition</u>
1-2	Integer	Element number
3-4	Integer	Number of sub-elements (SE ≥ 1)
5	Alpha	Excitation type on this element (if any) A = acoustic    M = direct mechanical
6-8	Alpha	If CC5 = A, leave blank; defaults to dB re 20 μbar If CC5 = M, RMS = 1/3 octave RMS g's PSD = g <sup>2</sup> /Hz input at 1/3 octave centers
9-18	Real	Element loss factor constant(η <sub>o</sub> )
19-28	Real	Loss factor high frequency slope (s) } such that $\eta(f) = \eta_o \left( \frac{f}{f_o} \right)^s$
29-38	Real	Loss factor crossover frequency (f <sub>o</sub> ) }

**Figure 3-4. Card 2, Example Case 1**

° Must immediately follow element properties if LS#2, CC5 ≠ blank

- All entries are real numbers, and must be consistent with LS #2, CC 6-8 (units) and LS #1, CC 3-4 (no. of frequencies - 40 max)
- Make entries across in order of increasing frequency

[illegible]

**3-8**

**LOADSHEET (4): SUB-ELEMENT PROPERTIES, CARDS 1 AND 2**

- Must be consistent with Loadsheet 2, CC 3-4 (SE ≥ 1)
- Two cards (records) per sub-element
- Must follow associated element card and excitation card (if any)

1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18	19	20	21	22	23	24	25	26	27	28	29	30	31	32	33	34	35	36	37	38	39	40	41	42	43	44	45	46	47	48	49	50	51	52	53	54	55	56	57	58	59	60	61	62	63	64	65	66	67	68	69	70	71	72	73	74	75	76	77	78	79	80	81	82	83	84	85	86	87	88	89	90	91	92	93	94	95	96	97	98	99	100
18 2.617E-4 10.0E+6 0.040 1296. 0.33																																																																																																			

DATA ENTRY DESCRIPTION	
01	01
02	02
03	03
04	04
05	05
06	06
07	07
08	08
09	09
10	10
11	11
12	12
13	13
14	14
15	15
16	16
17	17
18	18
19	19
20	20
21	21
22	22
23	23
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86	86
87	87
88	88
89	89
90	90
91	91
92	92
93	93
94	94
95	95
96	96
97	97
98	98
99	99
100	100

<u>Card Columns</u>	<u>Value</u>	<u>Definition</u>
1-2	Integer	Sub-element number (Sub-element 1 is always the main sub-element)
3	Alpha	Sub-element type; B = beam, C = cylinder, M = membrane, P = plate R = room (acoustic element)
4-13	Real	Mass density (B, C, M, P, R)
14-23	Real	Elastic modulus (B, C, M, P)
24-33	Real	Thickness (B, C, M, P)
34-43	Real	Area (section if B; surface if M or P)
44-53	Real	Poisson's ratio (C, P)
54-63	Real	Length (B, C)
64-73	Real	Pressure (M only)
1	Logical	Replace F with T if stiffness increase is desired (B, C, P)
2-11	Real	Radius (C only)
12-21	Real	Volume (R only)
22-31	Real	Speed of sound - Sub-element 1 only and if $\begin{cases} (P) \\ (R) \end{cases}$ and CC5, LS2 = A, or unconditionally
32-41	Real	Added non-structural mass (B, C, M, P)

**Figure 3-6. Card 4, Example Case 1**

# LOADSHEET (4): SUB-ELEMENT PROPERTIES, CARDS 1 AND 2

- Must be consistent with Loadsheet 2, CC 3-4 (SE ≥ 1)
- Two cards (records) per sub-element
- Must follow associated element card and excitation card (if any)

1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18	19	20	21	22	23	24	25	26	27	28	29	30	31	32	33	34	35	36	37	38	39	40	41	42	43	44	45	46	47	48	49	50	51	52	53	54	55	56	57	58	59	60	61	62	63	64	65	66	67	68	69	70	71	72	73	74	75	76	77	78	79	80	81	82	83	84	85	86	87	88	89	90	91	92	93	94	95	96	97	98	99	100
24													2,617E-4													10.0E+4													C.063													508													0.33																																		

1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18	19	20	21	22	23	24	25	26	27	28	29	30	31	32	33	34	35	36	37	38	39	40	41	42	43	44	45	46	47	48	49	50	51	52	53	54	55	56	57	58	59	60	61	62	63	64	65	66	67	68	69	70	71	72	73	74	75	76	77	78	79	80	81	82	83	84	85	86	87	88	89	90	91	92	93	94	95	96	97	98	99	100
F																																																				1.0E-3																																															

## DATA ENTRY DESCRIPTION

Card	Columns	Value	Definition
Card 1	1-2	Integer	Sub-element number (Sub-element 1 is always the main sub-element)
	3	Alpha	Sub-element type; B = beam, C = cylinder, M = membrane, P = plate R = room (acoustic element)
	4-13	Real	Mass density (B, C, M, P, R)
	14-23	Real	Elastic modulus (B, C, M, P)
	24-33	Real	Thickness (B, C, M, P)
	34-43	Real	Area (section if B; surface if M or P)
	44-53	Real	Poisson's ratio (C, P)
	54-63	Real	Length (B, C)
	64-73	Real	Pressure (M only)
	1	Logical	Replace F with T if stiffness increase is desired (B, C, P)
Card 2	2-11	Real	Radius (C only)
	12-21	Real	Volume (R only)
	22-31	Real	Speed of sound - Sub-element 1 only and if $\begin{cases} (P) \\ (R) \end{cases}$ unconditionally
	32-41	Real	Added non-structural mass (B, C, M, P)

Figure 3-7. Card 5, Example Case 1

Sub-element 1 consists of the skin and stringer elements. The skin is used as the principal property because it is the major element being excited by the acoustic field. The stringers are added as a smeared mass because their internal resonant frequencies are very high and therefore would be expected only to load the skin in the frequency range of interest. Sub-element 2 (Figure 3-7) represents the properties of the internal frames and channels which mount the panel.

Figures 3-8 and 3-9 show the loadsheets for the inside equipment panel. Figure 3-8 defines the panel as Element 2, with one sub-element and constant damping vs. frequency. Figure 3-9 has the panel properties which have been equivalenced to an isotropic plate because the panel is a composite structure. Note that both the thickness and the density have been changed to be consistent. Any other equivalence could also be used, e.g. leave the thickness = 1.0 and the density and Young's modulus will be recalculated using

$$\frac{1}{T} \sum_i e_i T_i = \rho_{eff}$$

and 
$$E_{eff} = \frac{12E(1-\nu^2) \sum_i I_i}{bt^3}$$

There is no requirement that sub-elements be of similar materials, but all plate, beam and cylinder sub-element properties must always be isotropic equivalents.

The weight of the components is treated as a non-structural mass. If half or more of the panel area were covered with these components, the F should be changed to a T as shown on the sheet to account for the reduction of modal density associated with stiffening of the panel by the components as reported in Reference 2.

Figure 3-10 describes the joint properties between the skin and the plate. Note that an added insertion loss factor of 2 is used to account for rivet effects.

# LOADSHEET (2): SEA ELEMENT PROPERTIES CARD

- One required for each element ( $2 \leq E \leq 20$ )
- Must be in ascending numerical sequence

1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18	19	20	21	22	23	24	25	26	27	28	29	30	31	32	33	34	35	36	37	38	
2	1																																					

## DATA ENTRY DESCRIPTION

Card Column	Value	Definition
1-2	Integer	Element number
3-4	Integer	Number of sub-elements ( $SE \geq 1$ )
5	Alpha	Excitation type on this element (if any) A = acoustic M = direct mechanical
6-8	Alpha	If CC5 = A, leave blank; defaults to dB re 20 $\mu$ bar If CC5 = M, RMS = 1/3 octave RMS g's PSD = $g^2/Hz$ input at 1/3 octave centers
9-18	Real	Element loss factor constant( $\eta_0$ )
19-28	Real	Loss factor high frequency slope (s) such that $\eta(f) = \eta_0 \left(\frac{f}{f_0}\right)^s$
29-38	Real	Loss factor crossover frequency ( $f_0$ )

Figure 3-8. Card 6, Example Case 1

## LOADSHEET (4): SUB-ELEMENT PROPERTIES, CARDS 1 AND 2

- Must be consistent with Loadsheet 2, CC 3-4 (SE ≥ 1)
- Two cards (records) per sub-element
- Must follow associated element card and excitation card (if any)

[illegible][illegible]

<u>Card Columns</u>	<u>Value</u>	<u>Definition</u>
1-2	Integer	Sub-element number (Sub-element 1 is always the main sub-element)
3	Alpha	Sub-element type; B = beam, C = cylinder, M = membrane, P = plate R = room (acoustic element)
4-13	Real	Mass density (B, C, M, P, R)
14-23	Real	Elastic modulus (B, C, M, P)
24-33	Real	Thickness (B, C, M, P)
34-43	Real	Area (section if B; surface if M or P)
44-53	Real	Poisson's ratio (C, P)
54-63	Real	Length (B, C)
64-73	Real	Pressure (M only)
1	Logical	Replace F with T if stiffness increase is desired (B, C, P)
2-11	Real	Radius (C only)
12-21	Real	Volume (R only)
22-31	Real	Speed of sound - Sub-element 1 only and if $\begin{cases} (P) \\ (R) \end{cases}$ and CC5, LS2 = A, or unconditionally
32-41	Real	Added non-structural mass (B, C, M, P)

**Figure 3-9. Card 7, Example Case 1**

- Must follow all element and sub-element cards at end of deck.
- Must be consistent with elements being joined.

[illegible]

<u>Card Columns</u>	<u>Value</u>	<u>Definition</u>
1-2	Integer	Element number of A end of joint
3-4	Integer	Element number of B end of joint
5-6	Alpha	Joint type: PP = plate-to-plate, BP = beam-to-plate, BJ = bolted joint, PA = plate to acoustic
7-8	Integer	No. of sides exposed to acoustic input (1 or 2 for PA only)
9-18	Real	Joint length
19-28	Real	Thickness of A end of joint
29-38	Real	Thickness of B end of joint
39-48	Real	Acoustic space mass density (P/RT)
49-58	Real	Beam length (BP only)
59-68	Real	Energy reduction factor (BJ only)

**Figure 3-10. Card 8, Example Case 1**



These data are then assembled as a file which the program reads as an input.

The corresponding program output is shown in Figure 3-11 and consists of a labeled list of the input data, a display of the element modal densities, and a table of vibration responses. The adequacy of the SEA model with regard to assumptions 4 and 5 can be checked using the modal density tables of the figure. For example, element 1 contains more than 20 modes per 1/3 octave over the entire analysis range, whereas element 2 has far fewer. Since frequency response will be smoother with more modes per band, it is expected that the panel prediction will be poorer than the skin predictions. Specifically, the actual panel response may have some peaks which exceed the SEA prediction.

The SEA response prediction is shown in Table 3-1. The levels for element 1 are high compared to the criteria published in Reference 3, but one must remember that this estimate includes a space average over the skin. The stringer and frame vibration levels which are inputs to the panel are a factor of  $\sim 100$  less.

Simple changes can be made to improve the prediction. For example, an internal acoustic field can be put on the panel by also placing an A in card column 5 of Figure 3-8 and entering the appropriate table immediately after, as shown in Figure 3-12. The results of this modification are shown in Table 3-2. As can be seen, the panel vibration increases considerably and critical frequency behavior is evident around 625 Hz. It should be noted that SEA often overpredicts in the critical frequency region. This simple example illustrates some of the possibilities of SEA with the help of this computer program.

A more extensive example is that of the MEA done in the previous phase. The input for this six-element model (Fig. 3-1) is extensive and much too elaborate to be explained in detail here, but the breakdown of the sub-elements used is given in Table 3-3. The input listing is given in Appendix III and the output is shown in Table 3-4.

A mechanical vibration input may also be applied if the known vibration level is included as an additional element. Load sheet 2 card column 5

# STATISTICAL ENERGY ANALYSIS OF COMPLEX STRUCTURES

RECORD  
NUMBER

DATA READ FROM UNIT 3

```

1  NUMBER OF ELEMENTS = 2
   NUMBER OF ANALYSIS FREQUENCIES = 11
   FIRST ANALYSIS FREQUENCY = 2.50000E+02
   TYPE OF UNITS =
   TYPE OF OUTPUT = PSD
2  ELEMENT NUMBER = 1
   NUMBER OF SUB-ELEMENTS = 2
   TYPE OF EXCITATION = A
   TYPE OF MECHANICAL INPUT =
   DAMPING = 1.00000E-01
   SLOPE = -1.00000E+00
   STARTING FREQUENCY = 2.50000E+02
      SOUND PRESSURE LEVELS
3      1.47000E+02  1.47500E+02  1.46000E+02  1.40000E+02
      1.49500E+02  1.50000E+02  1.50000E+02  1.50000E+02
4      1.50000E+02  1.53000E+02  1.50000E+02  0.
      0. 0. 0. 0.
5  SUB-ELEMENT NUMBER = 1
   TYPE OF SUB-ELEMENT = P
   DENSITY = 2.61700E-04
   MODULUS OF ELASTICITY = 1.00000E+07
   THICKNESS = 4.00000E-02
   AREA = 1.23600E+03
   POISSONS RATIO = 3.30000E-01
   LENGTH = 0.
   PRESSURE = 0.
6  STIFFNESS REDUCTION REQUIRED = F
   RADIUS = 0.
   VOLUME = 0.
   SPEED OF SOUND IN ROOM MEDIUM = 1.34000E+04
   ADDED MASS = 1.30500E-02
7  SUB-ELEMENT NUMBER = 2
   TYPE OF SUB-ELEMENT = P
   DENSITY = 2.61700E-04
   MODULUS OF ELASTICITY = 1.00000E+07
   THICKNESS = 6.30000E-02
   AREA = 5.04000E+02
   POISSONS RATIO = 3.30000E-01
   LENGTH = 0.
   PRESSURE = 0.
8  STIFFNESS REDUCTION REQUIRED = F
   RADIUS = 0.
   VOLUME = 0.
   SPEED OF SOUND IN ROOM MEDIUM = 0.
   ADDED MASS = 0.

```

(Continued)

Figure 3-11. Two-Element SEA Program Output - External Acoustic Excitation

9 ELEMENT NUMBER = 2  
 NUMBER OF SUB-ELEMENTS = 1  
 TYPE OF EXCITATION =  
 TYPE OF MECHANICAL INPUT =  
 DAMPING = 5.00000E-02  
 SLOPE = 0.  
 STARTING FREQUENCY = 0.

10 SJB-ELEMENT NUMBER = 1  
 TYPE OF SJB-ELEMENT = P  
 DENSITY = 3.49000E-05  
 MODULUS OF ELASTICITY = 1.00000E+07  
 THICKNESS = 3.00000E-01  
 AREA = 5.10000E+02  
 POISSONS RATIO = 3.30000E-01  
 LENGTH = 0.  
 PRESSURE = 0.

11 STIFFNESS REDUCTION REQUIRED = F  
 RADIUS = 0.  
 VOLUME = 0.  
 SPEED OF SOUND IN ROOM MEDIUM = 0.  
 ADDED MASS = 4.15000E-02

CENTER FREQ(HZ)	MODAL DENSITY - MODES/(RAD/SEC)	
	ELEMENT 1	ELEMENT 2
250.00	5.38653E-02	8.26429E-04
312.50	5.38653E-02	8.26429E-04
400.00	5.38653E-02	8.26429E-04
500.00	5.38653E-02	8.26429E-04
625.00	5.38653E-02	8.26429E-04
787.50	5.38653E-02	8.26429E-04
1000.00	5.38653E-02	8.26429E-04
1250.00	5.38653E-02	8.26429E-04
1575.00	5.38653E-02	8.26429E-04
2000.00	5.38653E-02	8.26429E-04
2500.00	5.38653E-02	8.26429E-04

RECORD  
 NUMBER  
 -----

DATA READ FROM UNIT 3  
 -----

12 FIRST ELEMENT = 1  
 SECOND ELEMENT = 2  
 TYPE OF JOINT = 4J  
 NUMBER OF SIDES = 0  
 JOINT LENGTH = 3.60000E+01  
 THICKNESS OF FIRST ELEMENT = 4.00000E-02  
 THICKNESS OF SECOND ELEMENT = 3.00000E-01  
 ACOUSTIC SPACE DENSITY = 0.  
 BEAM LENGTH = 0.  
 INSERTION LOSS FACTOR = 2.00000E+00

Figure 3-11 (Continued)

Table 3-1

TWO-ELEMENT SEA VIBRATION PREDICTION  
EXTERNAL ACOUSTIC EXCITATION

Center Frequency (Hz)	PSD Levels ( $G^2/Hz$ )	
	Element 1	Element 2
250.0	4.90675E+01	3.38593E-01
312.5	3.59925E+01	2.37220E-01
400.0	2.53599E+01	1.58301E-01
500.0	2.11177E+01	1.25095E-01
625.0	1.53027E+01	8.85583E-02
787.5	1.17827E+01	6.21389E-02
1000.0	7.83661E+01	3.86783E-02
1250.0	5.44536E+00	2.51832E-02
1575.0	3.81675E+00	1.64492E-02
2000.0	2.72180E+00	1.08695E-02
2500.0	2.05294E+00	7.61286E-03

for that element would contain an M and the vibration spectrum description would follow. The only restriction is that this element may not also have an acoustic input as it will be eliminated in the solution because the energy level is already known.

The possible permutations and combinations of elements, sub-elements, and other factors which this computer program can create go far beyond the ability to document in this report. These few examples give some insight into the processes involved in the performance of SEA using this program. Although determination of some of the parameters such as damping and insertion loss is still a difficult and often obscure process which requires substantial future improvement, this computer program provides a significant step toward streamlining and simplifying SEA for the analyst.

- Must immediately follow element properties if LS#2, CC5 ≠ blank

- All entries are real numbers, and must be consistent with LS #2, CC 6-8 (units) and LS #1, CC 3-4 (no. of frequencies - 40 max)
- Make entries across in order of increasing frequency

[illegible]

3-20

Table 3-2  
TWO-ELEMENT SEA VIBRATION PREDICTION  
EXTERNAL AND INTERNAL ACOUSTIC EXCITATION

<u>Center Frequency (Hz)</u>	<u>PSD Levels (G<sup>2</sup>/Hz)</u>	
	<u>Element 1</u>	<u>Element 2</u>
250.0	4.96984E+01	9.69548E-01
312.5	3.67160E+01	1.04475E+00
400.0	2.62012E+01	1.21999E+00
500.0	2.26604E+01	2.29959E+00
625.0	2.05716E+01	7.59740E+00
787.5	1.40047E+01	3.98634E+00
1000.0	8.75443E+00	1.86390E+00
1250.0	5.75084E+00	7.03965E-01
1575.0	3.94377E+00	3.33085E-01
2000.0	2.76144E+00	1.22184E-01
2500.0	2.06547E+00	4.69292E-02

Table 3-3  
SUB-ELEMENT BREAKDOWN FOR MEA TEST CASE

ELEMENT 1		ELEMENT 2	
Sub-element		Sub-element	
1	Thermal Panels	1	Power Distribution Panel
2	Orbiter Interface Panels	2	Power Distribution Box
3	Support Structure	3	
4		4	
5			
6			
7			
8	Radiator		
9			
10	Interface Support Structure		
11			
12			
13	Signal Distributor Panel		
14	Signal Distribution Box		
15			
16			
17	Support Brace Assembly and Gusset		
18			
19			
20	Pressure Sensor and Voltage Regulator Panels		

ELEMENT 3	
Sub-element	
1	Data Acquisition Cold Plate
2	
3	

ELEMENT 4	
Sub-element	
1	Solenoid Panel

ELEMENT 5	
Sub-element	
1	Battery Cold Plate
2	
3	

ELEMENT 6	
Sub-element	
1	Experiment Mounting Plates
2	



Table 3-4  
SEA COMPUTER PROGRAM OUTPUT  
FOR MEA/ACOUSTIC EXCITATION CASE

CENTER FREQ(HZ)	PSD LEVELS (G**2/HZ)			
	ELEMENT 1	ELEMENT 2	ELEMENT 3	ELEMENT 4
31.50	1.37117E+01	6.40344E-02	1.23606E-02	7.74395E-02
39.39	1.39434E+01	5.33723E-02	1.16254E-02	7.03306E-02
50.40	1.35352E+01	5.01950E-02	1.03534E-02	6.00698E-02
63.00	1.37834E+01	4.58005E-02	9.75593E-03	5.43442E-02
78.75	1.40495E+01	4.16226E-02	9.20447E-03	4.90858E-02
99.23	1.41144E+01	3.74861E-02	8.53649E-03	4.33913E-02
126.00	1.39873E+01	3.30110E-02	7.79522E-03	3.75635E-02
157.50	1.13790E+01	2.40471E-02	5.87560E-03	2.68592E-02
198.45	1.02517E+01	1.93208E-02	4.89243E-03	2.11104E-02
252.00	7.25080E+00	1.22183E-02	3.20463E-03	1.30301E-02
315.00	4.73043E+00	8.57629E-03	2.19825E-03	9.29518E-03
393.75	3.10091E+00	6.04840E-03	1.51521E-03	6.65439E-03
504.00	1.95717E+00	4.13889E-03	1.01107E-03	4.62347E-03
630.00	8.21183E-01	1.36311E-03	4.46156E-04	2.11322E-03
787.50	3.47857E-01	8.51231E-04	1.98790E-04	9.74010E-04
992.25	9.28257E-02	2.44964E-04	5.59056E-05	2.33312E-04
1260.00	4.96000E-02	1.41506E-04	3.15427E-05	1.65277E-04
1575.00	2.77460E-02	3.51314E-05	1.85675E-05	1.00240E-04
1984.50	1.24869E-02	4.13079E-05	8.81054E-06	4.59941E-05
2520.00	5.74855E-03	2.05537E-05	4.28491E-06	2.45336E-05
3150.00	2.82043E-03	1.08428E-05	2.21322E-06	1.30057E-05
3937.50	1.48151E-03	6.12303E-06	1.22397E-06	7.37309E-06
5040.00	8.43835E-04	3.77799E-06	7.34086E-07	4.56355E-06

CENTER FREQ(HZ)	PSD LEVELS (G**2/HZ)	
	ELEMENT 5	ELEMENT 6
31.50	3.80958E-03	9.53944E-03
39.38	3.55841E-03	8.88172E-03
50.40	3.13632E-03	7.90207E-03
63.00	2.92061E-03	7.24503E-03
78.75	2.71710E-03	6.72258E-03
99.23	2.47860E-03	6.11710E-03
126.00	2.21901E-03	5.46337E-03
157.50	1.63920E-03	4.02753E-03
198.45	1.33442E-03	3.27234E-03
252.00	8.53722E-04	2.08991E-03
315.00	5.94794E-04	1.45769E-03
393.75	4.16159E-04	1.02111E-03
504.00	2.82109E-04	6.93161E-04
630.00	1.26179E-04	3.10445E-04
787.50	5.69414E-05	1.40294E-04
992.25	1.62123E-05	4.00061E-05
1260.00	9.25557E-06	2.28779E-05
1575.00	5.50341E-06	1.36259E-05
1984.50	2.63620E-06	6.53979E-06
2520.00	1.29324E-06	3.21405E-06
3150.00	6.72712E-07	1.67509E-06
3937.50	3.74282E-07	9.33860E-07
5040.00	2.26942E-07	5.67554E-07

## Section 4

### SEA COMPUTER PROGRAM DESCRIPTION AND USAGE

This section describes the computer code and the input requirements for problem solutions.

The program was written in ASCII Fortran V for the MSC UNIVAC 1108 with the EXEC8 operating system resident in the computer. The program is designed to be run in the demand/interactive mode using the system editor to assemble and sequence the load sheet information.

The SEA computer code uses a main program whose function is essentially to call the various subroutines in the correct order. First called are two subroutines, EPROP and JINPUT, which read user-supplied data. Then for each analysis frequency, three other subroutines (JPROP, EXCITE, and ANSWER) are called to perform the various calculations. The subroutine ANSWER calls the UNIVAC library subroutine GASSEM from SYS \$\*MATHSTAT\$. Finally, subroutine RITER is called to print the solution. The complete program listing is given in Appendix I and the program flow charts are given in Appendix II.

#### 4.1 SUBROUTINE DESCRIPTION

Subroutine EPROP reads user-supplied data giving element and sub-element properties and calls one of five subroutines (BEAM, MEMBR, PLATE, ROOM, or CYLIN) to calculate element modal densities. See Appendix I for the list of variables read by EPROP. EPROP first reads a record giving the number of elements, number of analysis frequencies, the first frequency, and the type of units (metric or English). EPROP checks that the number of elements and analysis frequencies is within the bounds allowed by the program, since memory is reserved with maximum values in mind. Then EPROP creates a table of analysis frequencies by multiplying each element of a

predefined 1/3 octave table by the first frequency. The predefined table consists of a series of frequencies, each of which is approximately one-third octave higher than the preceding one, and the first one of which is equal to 1. EPROP then reads element properties. If the type of excitation indicated on this read operation is acoustic, EPROP then reads a list of sound pressure levels for each analysis frequency for this element. Next, sub-element properties are read. Included in sub-element properties is the type of sub-element. Depending on whether this is a beam, membrane, plate, room or cylinder, EPROP calls the appropriate subroutine to calculate the modal density.

Subroutines BEAM, MEMBR, PLATE, ROOM and CYLIN calculate the modal density for a sub-element which is a beam, membrane, plate, room or cylinder, respectively. First, the part of the equation which is not frequency dependent is calculated. If stiffness reduction has been indicated, this partial value is multiplied by 1/ 2. Then for each analysis frequency, the rest of the modal density equation is computed and the result summed to element modal density. The sub-element mass is summed to the element mass. If this is the first sub-element for the given element, it is assumed to be the main sub-element and element properties other than mass are set equal to the properties of this sub-element. The following equations are used to calculate modal densities:

$$\text{Beam: } n(\omega) = \frac{L}{2\pi} \left( \omega \sqrt{\frac{Et^2}{12\rho}} \right)^{-1/2}$$

$$\text{Membrane: } n(\omega) = \frac{A\omega\rho t}{2\pi S}$$

$$\text{Plate: } n(\omega) = \frac{A}{4\pi} \left( \frac{Et^2}{12\rho(1-\gamma^2)} \right)^{-1/2}$$

$$\text{Room: } n(\omega) = \frac{V\omega^2}{2\pi^2 c^3}$$

Cylindrical shell:

$$n(\omega) = \begin{cases} \frac{A}{4\pi} \left( \frac{Et^2}{12\rho(1-\gamma^2)} \right)^{-1/2} & \text{if } \omega r \left( \frac{\rho}{E} \right)^{1/2} > 1 \\ \frac{A}{4\pi} \left( \frac{Et^2}{12\rho(1-\gamma^2)} \right)^{-1/2} \left( \omega r \left( \frac{\rho}{E} \right)^{1/2} \right)^{2/3} & \text{if } \omega r \left( \frac{\rho}{E} \right)^{1/2} < 1 \end{cases}$$

<u>Symbol</u>	<u>FORTRAN Name</u>	<u>Description</u>
n	N	Modal Density
L	L	Length
$\omega$	OMEGA	2 $\pi$ Times the Frequency
E	E	Modulus of Elasticity
t	T	Thickness
$\rho$	RHO	Density
A	A	Area
S	S	Pressure
$\gamma$	GAMMA	Poisson's Ratio
V	V	Volume
C	C	Speed of Sound

Subroutine JINPUT reads joint properties supplied by the user. See Appendix I for a list of the variables read by JINPUT. JINPUT checks that element numbers are within range (i.e., less than or equal to the number of elements in the system) and that no pair of element numbers is input more than once. JINPUT keeps a running total of the number of pairs input and continues reading until the end of file is encountered. JINPUT checks to see that at least one pair of elements was read.

Subroutine JPROP calculates coupling coefficients based on the data read by JINPUT, using one of the following equations, depending on the type of joint as read by JINPUT:

Plate to plate at right angles:

$$\phi_{12} = \frac{1.07L}{\pi A N_2} \left( \omega t_1 \left( \frac{E_1}{\rho_1 (1-\gamma^2)} \right)^{1/2} \right)^{1/2} \tau$$

$$A = n_1 2 t_1 \left( \frac{E_1}{12 \rho_1 (1-\gamma^2)} \right)$$

$$\tau = \begin{cases} \frac{8}{27} & \text{if } t_1 > \frac{t_2}{2} \\ \frac{t_1}{t_2} & \text{if } t_1 < \frac{t_2}{2} \end{cases}$$

Beam to plate (cantilevered):

$$\phi_{12} = \frac{2\pi f b}{N_2 4L}$$

Plate to acoustic space:

$$\phi_{12} = \left( \frac{4.33\pi c_2^4}{f\omega^2 V_2} \right) \left( \frac{\alpha \rho_1 \sigma}{\rho_A} \right)$$

Bolted or riveted joints:

- 1) Calculate PHI as if for a plate-to-plate rigid joint.
- 2) Reduce PHI by insertion loss factor.

<u>Symbol</u>	<u>FORTTRAN Name</u>	<u>Description</u>
$\phi$	PHI	Coupling Coefficient
L	JL	Joint Length
N	MODES	Number of Modes in Bandwidth
$\omega$	OMEGA	$2\pi$ Times the Frequency
t	T	Thickness
E	EE	Modulus of Elasticity
$\rho$	DENSE	Density
$\gamma$	EGAMMA	Poisson's Ratio
$\tau$	TAU	Thickness Ratio
n	N	Modal Density
f	FREQ	Frequency
b	BW	Beam Width
c	EC	Speed of Sound in Room Medium
V	VOL	Volume
a	NS	Number of Sides
$\sigma$	SIGMA	Radiation Efficiency
$\rho_A$	ASD	Acoustic Space Density

Function SIGF returns a value for the radiation efficiency of a panel. A single argument, X, is passed to SIGF. The value of X is the ratio of the analysis frequency to the critical frequency. An internal table of 16 values of the log of the radiation efficiency for  $0 \leq X \leq 4$  is maintained. The first value, 0, is the value of X for which SIGF is a minimum. When  $X = 0$ ,  $SIGF = -1.8$ . Each subsequent value of the internal table is the value of X for which SIGF increases by 0.2 over the previous value. The 13th value in the table is 1. Since this is the 12th value after the 1st,  $SIGF = 12 * 0.2 + (-1.8) = 0.6$  when  $X = 1$ . This is the maximum value of SIGF.  $SIGF = 0$  for  $X = 4$ , the final value of the table. The value of SIGF is calculated by finding the least value of the table that is greater than X. This value and the preceding one give two values of SIGF that differ by 0.2. Linear interpolation is then used to find the actual value of SIGF.

Subroutine EXCITE calculates acoustic and mechanical energy inputs for the elements of the system. These values are initially set to 0. Then for each element, the acoustic or mechanical energy input is calculated

according to the following equations, depending on the type of excitation as read by EPROP:

$$\text{Acoustic: } S = \frac{0.66\pi c^2 A^2 \langle \bar{p}^2 \rangle}{\omega^3 m} \sigma N$$

$$\langle \bar{p}^2 \rangle = 10^{\text{SPL}/10} (8.41 \times 10^{-18})$$

$$\text{Mechanical in } g_{\text{rms}}: E = \frac{m}{\omega^2} g^2 g_{\text{rms}}^2$$

$$\text{Mechanical in PSD Levels: } E = \frac{m}{\omega^2} \text{PSD } g^2 \frac{f}{4.33}$$

<u>Symbol</u>	<u>FORTRAN Name</u>	<u>Description</u>
S	S	Acoustic Energy Input
c	EC	Speed of Sound in Room Medium
A	A	Surface Area Exposed to Sound Field
$\sigma$	SIGMA	Radiation Efficiency
N	MODES	Number of Modes in Bandwidth
$\omega$	OMEGA	2 $\pi$ Times the Frequency
m	MSUB	Mass
SPL	SPL	Sound Pressure Level
E	E	Element Energy Level
$g_{\text{rms}}$	MECH	Mechanical Input
g	G	Gravitational Constant
PSD	MECH	Mechanical Input
f	FREQ	Frequency

Subroutine ANSWER solves the SEA system of equations for element energy levels. First, element damping is determined. If the damping is constant, it is equal to the value read by EPROP. Otherwise the following equation is used:

$$\eta_2 = \eta_f \left( \frac{f}{f_s} \right)^s$$

<u>Symbol</u>	<u>FORTRAN Name</u>	<u>Description</u>
$\eta_2$	ETA2	Element Damping
$\eta_f$	ETA	Constant Level of Damping
$f_s$	SFREQ	Start Frequency
s	SLOPE	Slope

Next, the elements of the alpha matrix of equation 1, Section 1, are calculated.

If there are any elements for which the energy levels (the E matrix) are already known, these are eliminated from the equation as shown by the following example: Suppose the system has six elements and the second and fifth have known energy levels. Then the equation becomes:

$$\begin{vmatrix} \alpha_{11} & \alpha_{13} & \alpha_{14} & \alpha_{16} \\ \alpha_{31} & \alpha_{33} & \alpha_{34} & \alpha_{36} \\ \alpha_{41} & \alpha_{43} & \alpha_{44} & \alpha_{46} \\ \alpha_{61} & \alpha_{63} & \alpha_{64} & \alpha_{66} \end{vmatrix} \begin{Bmatrix} E_1 \\ E_3 \\ E_4 \\ E_6 \end{Bmatrix} = \begin{Bmatrix} S_1 - \alpha_{12}E_2 - \alpha_{15}E_5 \\ S_3 - \alpha_{32}E_2 - \alpha_{35}E_5 \\ S_4 - \alpha_{42}E_2 - \alpha_{45}E_5 \\ S_6 - \alpha_{62}E_2 - \alpha_{65}E_5 \end{Bmatrix}$$

Since (for example)  $S_1 = \alpha_{11}E_1 + \alpha_{12}E_2 + \alpha_{13}E_3 + \alpha_{14}E_4 + \alpha_{15}E_5 + \alpha_{16}E_6$ , it can be seen that this has the same solution as the original equation. ANSWER recalculates the values of the S matrix and calls SOLVE to eliminate the unnecessary rows and columns from the matrices and find the solution. The solution is then used to calculate the average acceleration with the formula

$$\langle \bar{a} \rangle = E \frac{\omega^2}{m}$$

<u>Symbol</u>	<u>FORTRAN Name</u>	<u>Description</u>
$\langle \bar{a} \rangle$	ABAR	Average Acceleration
E	E	Element Energy Levels
$\omega$	OMEGA	2 $\pi$ Times the Frequency
m	MASS	Mass

Subroutine SOLVE solves the SEA system of equations. It copies the alpha and S matrices to new matrices, leaving out those rows and columns which were to be eliminated. It then calls MATHSTAT library subroutine GASSEM to solve the equation. SOLVE then puts the solution in the element energy array E, and subroutine RITER prints the solution.

## 4.2 PROGRAM USAGE

### 4.2.1 Deck Setup and Sequence

At present, the source program resides on element C of file S1, so that it is necessary to compile and collect it before execution. The program

reads the SEA load sheet data on logical unit 3, which must be created by the user with the text editor or data processor, using the Q option so that file 3 is in ASCII code. The following sequence of control statements illustrates the creation of file 3 and the execution of the program:

```
@RUN ...
@ASG,C 3.
@ED,IQ 3.
    Statements creating file 3
EXIT
@ASG,A S1
@FTN,N S1.C,REL
@MAP,IN SYM,ABS
LIB SYS$*MATHSTAT$.
LIB SYS$*MSFC$.
LIB SYS$*MSFC$.
END
@XQT ABS
@FIN
```

#### 4.2.2 Input - Drum/Disk

The only input for the program is on logical unit 3, which contains the user's data. This file consists of the following five types of records:

1. Initial information used to process the other records.
2. Element properties.
3. Sound pressure levels or mechanical inputs.
4. Sub-element properties.
5. Joint properties.

The arrangement of these records and their data elements is shown on the following five loadsheet pages (Figures 4-1 through 4-5).



# LOADSHEET (1): HEADER CARD

- One required per case
- First card in sequence

1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18

## DATA ENTRY DESCRIPTION

Card Columns	Value	Definition
1-2	Integer	Total number of elements in model ( $2 \leq E \leq 20$ )
3-4	Integer	Total number of analysis frequencies ( $1 \leq F \leq 40$ ) (Analysis frequencies are spaced $1/3$ octave apart)
5-14	Real	Lowest analysis frequency
15	Alpha	Units; H = metric (MKS), default = English (in, lb, sec)
16-18	Alpha	Output mode; RMS or PSD

Figure 4-1. Loadsheets 1

# LOADSHEET (2): SEA ELEMENT PROPERTIES CARD

- One required for each element ( $2 \leq E \leq 20$ )
- Must be in ascending numerical sequence

1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18	19	20	21	22	23	24	25	26	27	28	29	30	31	32	33	34	35	36	37	38	39	40	41	42	43	44	45	46	47	48		

## DATA ENTRY DESCRIPTION

Card Column	Value	Definition
1-2	Integer	Element number
3-4	Integer	Number of sub-elements ( $SE \geq 1$ )
5	Alpha	Excitation type on this element (if any) A = acoustic M = direct mechanical
6-8	Alpha	If CC5 = A, leave blank; defaults to dB re 20 $\mu$ bar If CC5 = M, RMS = 1/3 octave RMS g's PSD = $g^2/Hz$ input at 1/3 octave centers
9-18	Real	If CC5 = A, input surface area exposed to acoustic excitation (consistent units)
19-28	Real	Element loss factor constant( $\eta_0$ )
29-38	Real	Loss factor high frequency slope (s)
39-48	Real	Loss factor crossover frequency ( $f_0$ )

such that  $\eta(f) = \eta_0 \left( \frac{f}{f_0} \right)^s$

Figure 4-2. Loadsheet 2



# LOADSHEET (4): SUB-ELEMENT PROPERTIES, CARDS 1 AND 2

- Must be consistent with Loadsheet 2, CC 3-4 (SE ≥ 1)
- Two cards (records) per sub-element
- Must follow associated element card and excitation card (if any)

1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18	19	20	21	22	23	24	25	26	27	28	29	30	31	32	33	34	35	36	37	38	39	40	41	42	43	44	45	46	47	48	49	50	51	52	53	54	55	56	57	58	59	60	61	62	63	64	65	66	67	68	69	70	71	72	73	74	75	76	77	78	79	80	81	82	83	84	85	86	87	88	89	90	91	92	93	94	95	96	97	98	99	100
1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18	19	20	21	22	23	24	25	26	27	28	29	30	31	32	33	34	35	36	37	38	39	40	41	42	43	44	45	46	47	48	49	50	51	52	53	54	55	56	57	58	59	60	61	62	63	64	65	66	67	68	69	70	71	72	73	74	75	76	77	78	79	80	81	82	83	84	85	86	87	88	89	90	91	92	93	94	95	96	97	98	99	100

## DATA ENTRY DESCRIPTION

<u>Card Columns</u>	<u>Value</u>	<u>Definition</u>
Card 1	1-2	Sub-element number
	3	Sub-element type; B = beam, C = cylinder, M = membrane, P = plate R = room (acoustic element)
	4-13	Mass density (B, C, M, P, R)
	14-23	Elastic modulus (B, C, M, P)
	24-33	Thickness (B, C, M, P)
	34-43	Area (section if B; surface if M or P)
	44-53	Poisson's ratio (C, P)
	54-63	Length (B, C)
	64-73	Pressure (M only)
Card 2	1	Replace F with T if stiffness increase is desired (B, C, P)
	2-11	Radius (C only)
	12-21	Volume (R only)
	22-31	Speed of sound (R, or 1f LS #2, CC5 = A)
	32-41	Added non-structural mass (B, C, M, P)

Figure 4-4. Loadsheet 4

LOADSHEET (5): JOINT PROPERTIES

- Must follow all element and sub-element cards at end of deck.
- Must be consistent with elements being joined.

1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18	19	20	21	22	23	24	25	26	27	28	29	30	31	32	33	34	35	36	37	38	39	40	41	42	43	44	45	46	47	48	49	50	51	52	53	54	55	56	57	58	59	60	61	62	63	64	65	66	67	68	69	70	71	72	73	74	75	76	77	78	79	80	81	82	83	84	85	86	87	88	89	90	91	92	93	94	95	96	97	98	99	00
---	---	---	---	---	---	---	---	---	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----

DATA ENTRY DESCRIPTION

Card Columns	Value	Definition
1-2	Integer	Element number of A end of joint
3-4	Integer	Element number of B end of joint
5-6	Alpha	Joint type: PP = plate-to-plate, BP = beam-to-plate, BJ = bolted joint, PA = plate to acoustic
7-8	Integer	No. of sides exposed to acoustic input (1 or 2)
9-18	Real	Joint length
19-28	Real	Thickness of A end of joint
29-38	Real	Thickness of B end of joint
39-48	Real	Acoustic space mass density (P/RT)
49-58	Real	Beam length (BP only)
59-68	Real	Bolt spacing (BJ only)

Figure 4-5. Loadsheets 5

#### 4.2.3 Output - Printout

The output consists of a list of input parameters, calculated modal densities, and calculated PSD levels (or  $G_{rms}$ ), if this was specified on input by the user) for each element for each analysis frequency, arranged in five columns below a heading. The first column is the frequencies, the other four columns list the PSD levels for elements 1 to 4. This is followed by five more columns below a new heading and containing the frequencies and PSD levels for elements 5 to 8. This is repeated until all the elements have been listed.

#### 4.2.4 Program Diagnostic Messages

The program can produce the error messages listed below. Suggestions for their cause and correction are also given. Lower case x's indicate values (usually, but not always, integers) that depend on the particular error.

\*\*\* ERROR \*\*\* THE INPUT FILE IS EMPTY. THE END OF FILE WAS ENCOUNTERED WHILE TRYING TO READ THE FIRST INPUT RECORD. THIS ERROR WAS DISCOVERED BY SUBROUTINE EPROP.

\*\*\* ERROR \*\*\* WHILE TRYING TO READ ELEMENT PROPERTIES, THE END OF FILE WAS ENCOUNTERED BEFORE INPUT RECORD xxxx. THIS ERROR WAS DISCOVERED BY SUBROUTINE EPROP.

\*\*\* ERROR \*\*\* WHILE TRYING TO READ SUB-ELEMENT PROPERTIES, THE END OF FILE WAS ENCOUNTERED BEFORE INPUT RECORD xxxx. THIS ERROR WAS DISCOVERED BY SUBROUTINE EPROP.

\*\*\* ERROR \*\*\* WHILE TRYING TO READ SOUND PRESSURE LEVELS, THE END OF FILE WAS ENCOUNTERED BEFORE INPUT RECORD xxxx. THIS ERROR WAS DISCOVERED BY SUBROUTINE EPROP.

\*\*\* ERROR \*\*\* WHILE TRYING TO READ MECHANICAL INPUTS, THE END OF FILE WAS ENCOUNTERED BEFORE INPUT RECORD xxxx. THIS ERROR WAS DISCOVERED BY SUBROUTINE EPROP.

\*\*\* ERROR \*\*\* THE END OF FILE WAS REACHED BEFORE ANY INFORMATION ABOUT JOINT PROPERTIES WAS READ. THIS ERROR WAS DISCOVERED BY SUBROUTINE JINPUT.

These errors are the result of an incomplete data file. The data file cannot end before joint properties for at least one pair of elements are read. The message indicates what kind of data the program was looking for when the end of file was encountered.

\*\*\* ERROR \*\*\* A FORTRAN ERROR OCCURRED WHILE TRYING TO READ ELEMENT PROPERTIES ON INPUT RECORD xxxx. THIS ERROR WAS DISCOVERED BY SUBROUTINE EPROP.

\*\*\* ERROR \*\*\* A FORTRAN ERROR OCCURRED WHILE TRYING TO READ SUB-ELEMENT PROPERTIES ON INPUT RECORDS xxxx AND xxxx. THIS ERROR WAS DISCOVERED BY SUBROUTINE EPROP.

\*\*\* ERROR \*\*\* A FORTRAN ERROR OCCURRED WHILE TRYING TO READ THE FIRST INPUT RECORD. THIS ERROR WAS DISCOVERED BY SUBROUTINE EPROP.

\*\*\* ERROR \*\*\* A FORTRAN ERROR OCCURRED WHILE TRYING TO READ SOUND PRESSURE LEVELS ON OR BEFORE INPUT RECORD xxxx. THIS ERROR WAS DISCOVERED BY SUBROUTINE EPROP.

\*\*\* ERROR \*\*\* A FORTRAN ERROR OCCURRED WHILE TRYING TO READ MECHANICAL INPUTS ON OR BEFORE INPUT RECORD xxxx. THIS ERROR WAS DISCOVERED BY SUBROUTINE EPROP.

\*\*\* ERROR \*\*\* WHILE ATTEMPTING TO READ JOINT PROPERTIES, A FORTRAN ERROR OCCURRED ON INPUT RECORD xxxx. THIS ERROR WAS DISCOVERED BY SUBROUTINE JINPUT.

A FORTRAN error is usually the result of invalid characters appearing in a data field; for example, alphabetic characters appearing where the program expects to read an integer. This is likely to occur if data RECORDS are missing or out of order; for example, if an element properties record says there are four sub-elements for that element, but records

for only three sub-elements are present. A FORTRAN error on the first record may indicate that the data file was not given the name 3. If the external file name is not 3, the internal file name should be made 3 by a USE statement.

\*\*\* ERROR \*\*\* THE NUMBER OF ELEMENTS WAS GIVEN AS xx. IT MUST BE BETWEEN 2 AND 20 INCLUSIVE. THIS ERROR WAS DISCOVERED BY SUBROUTINE EPROP.

\*\*\* ERROR \*\*\* THE NUMBER OF ANALYSIS FREQUENCIES WAS GIVEN AS xx. IT MUST BE BETWEEN 1 AND 40 INCLUSIVE. THIS ERROR WAS DISCOVERED BY SUBROUTINE EPROP.

\*\*\* ERROR \*\*\* THE TYPE OF MECHANICAL INPUT GIVEN FOR ELEMENT xx ON INPUT RECORD xxxx IS xxx. THIS IS NOT A VALID TYPE. THE TYPE MUST BE RMS OR PSD. THIS ERROR WAS DISCOVERED BY SUBROUTINE EPROP.

\*\*\* ERROR \*\*\* TYPE = x ON INPUT RECORD xxxx IS AN INVALID TYPE. TYPE MUST BE B, M, P, C, OR R. THIS ERROR WAS DISCOVERED BY SUBROUTINE EPROP.

\*\*\* ERROR \*\*\* THE TYPE OF JOINT GIVEN FOR ELEMENT PAIR xx AND xx ON INPUT RECORD xxxx IS xx. THIS IS NOT A VALID TYPE. THE TYPE MUST BE PP, BP, BJ, OR PA. THIS ERROR WAS DISCOVERED BY SUBROUTINE JPROP.

\*\*\* ERROR \*\*\* THE DETERMINANT OF THE SEA EQUATION MATRIX IS 0. HENCE THERE IS NO SOLUTION. THIS ERROR WILL CAUSE THE PROGRAM TO ABORT. THIS ERROR WAS DISCOVERED BY SUBROUTINE SOLVE.

These messages are self explanatory. If any of the errors listed thus far occur, the subroutine in which they occur will continue processing. Before control is transferred to another subroutine, however, the following message is printed and the program is aborted through a CALL FERR statement:

BECAUSE OF THE ERRORS LISTED ABOVE, THE SEA PROGRAM WILL ABORT.



The following errors will not cause the program to abort, but may put the results in error:

\*\*\* WARNING \*\*\* ON INPUT RECORD xxxx THE ELEMENT NUMBER WAS GIVEN AS xx, WHICH IS OUT OF ORDER. IT HAS BEEN CHANGED TO xx. THIS ERROR WAS DISCOVERED BY SUBROUTINE EPROP.

Element properties are stored in arrays in the order in which they are read. If elements are referenced in the joint properties section in any other order, that is, when they are referenced as one element of a pair, the results are almost certainly erroneous.

\*\*\* WARNING \*\*\* ON INPUT RECORD xxxx THE SUB-ELEMENT NUMBER WAS GIVEN AS xx, WHICH IS OUT OF ORDER. IT SHOULD BE xx. THIS ERROR WAS DISCOVERED BY SUBROUTINE EPROP.

Sub-element properties are not stored in arrays so this is not likely to result in an error. If other errors occur, this message probably indicates that there are missing or extraneous records.

\*\*\* WARNING \*\*\* ON INPUT RECORD xxxx, ONE OR BOTH MEMBERS OF THE ELEMENT PAIR xx AND xx WAS EITHER LESS THAN 1 OR GREATER THAN xx, THE TOTAL NUMBER OF ELEMENTS. THIS PAIR WILL BE IGNORED. THIS ERROR WAS DISCOVERED BY SUBROUTINE JINPUT.

\*\*\* WARNING \*\*\* THE ELEMENT PAIR xx AND xx ON INPUT RECORD xxxx WAS PREVIOUSLY READ ON INPUT RECORD xxxx. THE FIRST VALUES WILL BE USED. THIS ERROR WAS DISCOVERED BY SUBROUTINE JINPUT.

\*\*\* WARNING \*\*\* ON INPUT RECORD xxxx, BOTH ELEMENT NUMBERS WERE GIVEN AS xx. THEY MUST BE DIFFERENT. THIS RECORD WILL BE IGNORED. THIS ERROR WAS DISCOVERED BY SUBROUTINE JINPUT.

These errors result in the indicated record being ignored. Whether or not the results are erroneous depends on whether the indicated record is necessary to the results.

In addition, data which causes overflow, negative arguments to the square root functions, etc., will produce ASCII FORTRAN diagnostics.

Section 5  
REFERENCES

1. R. W. Trudell, L. I. Yano. Statistical Energy Analysis of Complex Structures - Phase II Final Report. McDonnell Douglas Astronautics Company - Huntington Beach report MDC G9203, September 1980.
2. R. F. Davis. Statistical Energy Analysis Reponse Prediction Methods for Structural Systems - Final Report. McDonnell Douglas Astronautics Company - Huntington Beach report MDC G8150, October 1979.
3. Acoustic, Shock and Vibration Criteria for Design Evaluation/Qualification and Formal Qualification Test Programs - Model Saturn S-IVB-IB/V. Douglas Missile & Space Systems Division report DAC 56624, June 1968.

## Appendix I

### SEA PROGRAM FORTRAN LISTING



```

1  C ..... SEA PROGRAM .....
2  C
3  C
4  C
5  C
6  C
7  C
8  C
9  C
10 C
11 C
12 C
13 C
14 C
15 C
16 C
17 C
18 C
19 C
20 C
21 C
22 C
23 C
24 C
25 C
26 C
27 C
28 C
29 C
30 C
31 C
32 C
33 C
34 C
35 C
36 C
37 C
38 C
39 C
40 C
41 C
42 C
43 C
44 C
45 C
46 C
47 C
48 C
49 C
50 C
51 C
52 C
53 C
54 C
55 C
56 C
57 C
58 C

```

STATISTICAL ENERGY ANALYSIS OF COMPLEX STRUCTURES

C INPUT IS READ FROM UNIT 3 IN ASCII CODE

C OUTPUT IS WRITTEN TO UNIT 6, THE PRINTER (OR TERMINAL IN DEMAND MODE)

C THE FOLLOWING VARIABLES ARE USED IN THIS PROGRAM UNIT:

NAME	TYPE	DESCRIPTION
AF	INT	ANALYSIS FREQUENCY ORIGNAL (FROM 1 TO NUMAF)
FREQ	REAL	FREQUENCY
FTAB(48)	REAL	TABLE OF ANALYSIS FREQUENCIES
NUMAF	INT	NUMBER OF ANALYSIS FREQUENCIES
OMEGA	REAL	2*PI*FREQUENCY

C THE FOLLOWING COMMON BLOCKS ARE USED:

COMMON	OTHER PROGRAM UNITS USING THIS COMMON BLOCK
CB1	EPROP, BEAM, MEMBR, PLATE, ROOM, CYLIN, RITER, BLOCK DATA
CB2	EPROP, JINPUT, BLOCK DATA
CB3	JPROP, EXCITE, ANSWER

INTEGER AN

LOGICAL ERROR

COMMON /CB1/ FTAB(48), NUMAF, FREQ1

COMMON /CB2/ ERROR

COMMON /CB3/ FREQ, AF, OMEGA

DATA PI /3.141592/

2C FORMAT (10RECAUSE OF THE ERRORS LISTED ABOVE, THE SEA PROGRAM,

• I WILL ABORT. I)

BEHIND 3

C CALL EPROP TO READ ELEMENT PROPERTIES INPUT

CALL EPROP (45)

C CALL JINPUT TO READ JOINT PROPERTIES INPUT

CALL JINPUT

C IF AN ERROR HAS OCCURRED ON INPUT, WRITE A MESSAGE AND

C TERMINATE THE PROGRAM.

5 IF (.NOT. ERROR) GO TO 10

WRITE (6,30)

CALL FERR

C DO FOR EACH ANALYSIS FREQUENCY

10 DO 20 AF = 1, NUMAF

C DEFINE THE CURRENT ANALYSIS FREQUENCY

FREQ = FTAB(AF)

OMEGA = 2. \* PI \* FREQ

C CALL JPROP TO CALCULATE JOINT PROPERTIES

CALL JPROP

C CALL EXCITE TO DETERMINE ENERGY SOURCES FROM EXCITATION INPUTS

CALL EXCITE

C CALL ANSWER TO SOLVE THE SEA EQUATIONS FOR ELEMENT ENERGIES

CALL ANSWER

2C CONTINUE

C CALL RITER TO WRITE OUT THE ANSWER

MAIN PROGRAM

PAGE 2

CALL RITER  
STOP  
ENC

57  
58  
59

MAIN 59  
MAIN 60  
MAIN 61

```

60 SUBROUTINE FPROP(*)
61 C THIS SUBROUTINE READS ELEMENT AND SUB-ELEMENT PROPERTIES FROM UNIT
62 C 3 AND CALLS THE APPROPRIATE SUBROUTINE, DEPENDING ON THE TYPE OF
63 C SUB-ELEMENT, TO CALCULATE ELEMENT MODAL DENSITIES. EPROP IS CALLED
64 C FROM THE MAIN PROGRAM.
65 C
66 C THE FOLLOWING VARIABLES ARE USED IN THIS PROGRAM UNIT:
67 C NAME TYPE DESCRIPTION
68 C ----
69 C A REAL AREA
70 C C REAL SPEED OF SOUND IN ROOM MEDIUM
71 C E REAL MODULUS OF ELASTICITY
72 C ELNUM INT ELEMENT NUMBER
73 C ETAC(20) REAL DAMPING
74 C ETYFC(20) CHAR TYPE OF EXCITATION
75 C FREQ1 REAL FIRST ANALYSIS FREQUENCY
76 C FTAP(40) REAL TABLE OF ANALYSIS FREQUENCIES
77 C G REAL GRAVITATIONAL CONSTANT
78 C GAMP REAL POISSON'S RATIO
79 C G1 REAL GRAVITATIONAL CONSTANT IN METRIC UNITS
80 C IHS1 INT NUMBER OF INPUT RECORDS READ
81 C L REAL LENGTH
82 C M REAL MASS
83 C MAIN LOG TRUE IF SUB-ELEMENT IS MAIN SUB-ELEMENT
84 C MECH(20,40) REAL MECHANICAL INPUT
85 C MYTPE(20) CHAR TYPE OF MECHANICAL INPUT
86 C NUMAF INT NUMBER OF ANALYSIS FREQUENCIES
87 C NUMEL INT NUMBER OF ELEMENTS
88 C NLSLR INT NUMBER OF SUB-ELEMENTS
89 C OTYPE CHAR TYPE OF OUTPUT
90 C R REAL RADIUS
91 C RHO REAL DENSITY
92 C S REAL PRESSURE
93 C SFREQ(20) REAL STARTING FREQUENCY
94 C SLOPE(20) REAL SLOPE
95 C SEL(20,40) REAL SOUND PRESSURE LEVEL
96 C STIFF LOG TRUE IS STIFFNESS REDUCTION REQUIRED
97 C SUBNUM INT SUB-ELEMENT NUMBER
98 C T REAL THICKNESS
99 C TYPE CHAR TYPE OF SUB-ELEMENT
100 C UITS CHAR TYPE IF METRIC, OTHERWISE ENGLISH UNITS
101 C V REAL VOLUME
102 C
103 C THE FOLLOWING COMMON BLOCKS ARE USED:
104 C BLOCK OTHER PROGRAM UNITS USING THE COMMON BLOCK
105 C -----
106 C C01 MAIN,RRAP,MEHRR,PLATE,ROOM,CYLIN,PITER,BLOCK DATA
107 C C02 MAIN,JTAPUT,BLOCK DATA
108 C C03 MEHRR,MEHRR,PLATE,ROOM,CYLIN
109 C C04 CAM,MEHRR,PLATE,ROOM,CYLIN,PFPROP,EXCITE,ANSWER
110 C C05 SLATS,EXCITE
111 C C06 JINPUT,JPROP,EXCITE,ANSWER,SOLVE,PITER,BLOCK DATA
112 C C07 ANSWER,PITER,BLOCK DATA
113 C C08 HFAM,MEHRR,PLATE,ROOM,CYLIN,JPROP,EXCITE,BLOCK DATA
114 C C09 JINPUT,BLOCK DATA
115

```

```

EPROP 2
EPROP 3
EPROP 4
EPROP 5
EPROP 6
EPROP 7
EPROP 8
EPROP 9
EPROP 10
EPROP 11
EPROP 13
EPROP 14
EPROP 15
EPROP 16
EPROP 17
EPROP 18
EPROP 19
EPROP 20
EPROP 21
EPROP 22
EPROP 23
EPROP 24
AUG26 1
EPROP 25
EPROP 26
EPROP 27
EPROP 28
EPROP 29
EPROP 30
MAY14 1
EPROP 31
EPROP 32
EPROP 33
EPROP 34
EPROP 35
EPROP 36
EPROP 37
AUG26 2
EPROP 38
EPROP 39
EPROP 40
EPROP 41
EPROP 42
EPROP 43
EPROP 44
EPROP 45
EPROP 46
EPROP 47
EPROP 48
EPROP 49
EPROP 50
OCT29 1
EPROP 52
EPROP 53
EPROP 54
EPROP 55

```

```

116 CHARACTER TYPE,ETYPF,MTYPE*3,UNITS,OTYPE*3
117 INTEGER ELAUM,SURNUM
118 REAL MECH*4,M*4,N
119 LOGICAL ERROR,STIFF,MAIN
120 COMMON /CB1/ FTAR(40),NUMAF,FREQ1
121 COMMON /CB2/ ERROR
122 COMMON /CB4/ ELNUM,T,E,GAMMA,RMO,M,STIFF,L,A,V,C,R,S,MAIN
123 COMMON /CB5/ THICK(20),AREA(20),DENSE(20),VOL(20),EE(20),
124 * EGAMMA(20),EC(20)
125 COMMON /CB6/ SPL(20,40),MECH(20,40),ANM(20),ETYPE(20),MTYPE(20)
126 COMMON /CB7/ NUMEL,G
127 COMMON /CB8/ SLOPE(20),SFREQ(20),ETA(20),OTYPE
128 COMMON /CB9/ N(20,40)
129 COMMON /CB10/ INPUT
130 DATA G1 /9.81/
131
132 300 FORMAT (2I2,F10.2,A1,A3)
133 310 FORMAT (2I2,A1,A3,F10.2)
134 320 FORMAT (I2,A1,F10.2,F10.2,F10.2)
135 330 FORMAT (I0... ERROR ... TYPE = I,A,I ON INPUT RECORD I,
136 1 I,A,I IS AN INVALID TYPE,I/I TYPE MUST BE B, M, P, C, OR R,I/
137 1 THIS ERROR WAS DISCOVERED BY SUBROUTINE EPROP,I/
138 340 FORMAT (I0... ERROR ... A FORTRAN ERROR OCCURRED WHILE I,
139 1 TRYING TO READ,I/ SUB-ELEMENT PROPERTIES ON INPUT RECORDS I,
140 2 I,A,I AND I,I,A,I,I/
141 1 THIS ERROR WAS DISCOVERED BY SUBROUTINE EPROP,I/
142 350 FORMAT (I0... ERROR ... A FORTRAN ERROR OCCURRED WHILE I,
143 1 TRYING TO READ,I/ ELEMENT PROPERTIES ON INPUT RECORD I,I,A,I,I/EPROP I,
144 1 THIS ERROR WAS DISCOVERED BY SUBROUTINE EPROP,I/
145 360 FORMAT (I0... ERROR ... A FORTRAN ERROR OCCURRED WHILE I,
146 1 TRYING TO READ,I/ THE FIRST INPUT RECORD,I/
147 1 THIS ERROR WAS DISCOVERED BY SUBROUTINE EPROP,I/
148 370 FORMAT (I0... ERROR ... THE INPUT FILE IS EMPTY. THE I,
149 1 END OF FILE WAS ENCOUNTERED,I/ WHILE TRYING TO READ THE I,
150 2 FIRST INPUT RECORD,I/
151 1 THIS ERROR WAS DISCOVERED BY SUBROUTINE EPROP,I/
152 390 FORMAT (I0... ERROR ... WHILE TRYING TO READ ELEMENT I,
153 * PROPERTIES,I/
154 1 THE END OF FILE WAS ENCOUNTERED BEFORE INPUT RECORD I,
155 2 I,A,I,I/
156 1 THIS ERROR WAS DISCOVERED BY SUBROUTINE EPROP,I/
157 390 FORMAT (I0... ERROR ... WHILE TRYING TO READ SUB-ELEMENT,I,
158 * PROPERTIES,I/AT THE END OF FILE WAS ENCOUNTERED BEFORE I,
159 2 INPUT RECORD I,I,A,I,I/
160 1 THIS ERROR WAS DISCOVERED BY SUBROUTINE EPROP,I/
161 400 FORMAT (A10.2)
162 410 FORMAT (I0... ERROR ... A FORTRAN ERROR OCCURRED WHILE I,
163 1 TRYING TO READ,I/ SOUND PRESSURE LEVELS ON OR BEFORE INPUT I,EPROP I,
164 2 RECORD I,I,A,I,I/
165 1 THIS ERROR WAS DISCOVERED BY SUBROUTINE EPROP,I/
166 420 FORMAT (I0... ERROR ... WHILE TRYING TO READ SOUND I,
167 * PRESSURE LEVELS,I/ THE END OF FILE WAS ENCOUNTERED I,
168 2 BEFORE INPUT RECORD I,I,A,I,I/
169 1 THIS ERROR WAS DISCOVERED BY SUBROUTINE EPROP,I/
170 430 FORMAT (I0... WARNING ... ON INPUT RECORD I,I,A,I THE I,
171

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EPROP 56
EPROP 57
MAY13 5
EPROP 59
EPROP 60
EPROP 61
EPROP 62
EPROP 63
MAY13 6
EPROP 65
EPROP 66
UNVAC? 1
EPROP 68
UNIVAC 2
EPROP 70
EPROP 71
EPROP 72
EPROP 73
OCT28 3
EPROP 75
EPROP 76
EPROP 77
EPROP 78
EPROP 79
EPROP 80
MAY13 11
EPROP 82
EPROP 83
EPROP 84
EPROP 85
EPROP 86
EPROP 87
EPROP 88
MAY13 12
MAY13 13
EPROP 91
EPROP 92
MAY13 14
MAY13 15
EPROP 94
EPROP 95
EPROP 96
EPROP 97
EPROP 98
EPROP 99
EPROP100
EPROP101
EPROP102
EPROP104
EPROP105
EPROP106
EPROP107
EPROP108
EPROP109
EPROP110
EPROP111

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172 1 TELEMET NUMBER WAS GIVEN// AS 1,12,1, WHICH IS OUT OF 1,
173 2 1ORDER, IT WAS BEEN CHANGED TO 1,12,1,1//
174 3 1 THIS ERROR WAS DISCOVERED BY SUBROUTINE EPROP.1)
175 44C FORMAT (10...WARNING ... ON INPUT RECORD 1,14,1 THE 1,
176 1 1SUB-ELEMENT NUMBER WAS GIVEN// AS 1,12,1, WHICH IS OUT OF 1,
177 2 1ORDER, IT SHOULD BE 1,12,1,1//
178 3 1 THIS ERROR WAS DISCOVERED BY SUBROUTINE EPROP.1)
179 44C FORMAT (10...ERROR ... THE ALPHA OF ELEMENTS 1,
180 1 1WAS GIVEN AS 1,12,1, IT MUST BE// BETWEEN 2 AND 20, 1,
181 2 1INCLUSIVE.//
182 3 1 THIS ERROR WAS DISCOVERED BY SUBROUTINE EPROP.1)
183 44C FORMAT (10...ERROR ... THE NUMBER OF ANALYSIS FREQUENCIES 1,
184 1 1WAS GIVEN AS 1,12,1, IT MUST BE// BETWEEN 1 AND 40, 1,
185 2 1INCLUSIVE.//
186 3 1 THIS ERROR WAS DISCOVERED BY SUBROUTINE EPROP.1)
187 470 FORMAT (10...ERROR ... A FORTRAN ERROR OCCURRED WHILE 1,
188 1 1 TRYING TO READ// MECHANICAL INPUTS ON OR BEFORE INPUT//,
189 2 1RECORD 1,14,1,1//
190 3 1 THIS ERROR WAS DISCOVERED BY SUBROUTINE EPROP.1)
191 48C FORMAT (10...ERROR ... WHILE TRYING TO READ 1,
192 1 1 MECHANICAL INPUTS,// THE END OF FILE WAS ENCOUNTERED 1,
193 2 1BEFORE INPUT RECORD 1,14,1,1//
194 3 1 THIS ERROR WAS DISCOVERED BY SUBROUTINE EPROP.1)
195 490 FORMAT (10...ERROR ... THE TYPE OF MECHANICAL INPUT 1,
196 1 1 GIVEN FOR ELEMENT 1,12// ON INPUT RECORD 1,14,1 IS 1,
197 2 1A3,1. THIS IS NOT A VALID TYPE. THE TYPE MUST//
198 3 1 BE R4,8 OR P5B,1//
199 3 1 THIS ERROR WAS DISCOVERED BY SUBROUTINE EPROP.1)
200 500 FORMAT (//5X,CENTER//,15X,MODAL DENSITY - MODES/(RAD/SEC)//
201 1 1AX,1FREQUENCY,4,15X,1ELEMENT 1,12:))
202 51C FORMAT (12,2,4,1PE15.5E2))
203 52C FORMAT (1M1 // 11X, 1STATISTICAL ENERGY ANALYSIS OF COMPLEX 1,
204 1 1STRUCTURES // 1 RECORD// 1 NUMBER, 13X, 1DATA READ 1,
205 2 1FROM UNIT 3// 1X, 6,1X,1, 13X, 21,1X,1) / 1
206 530 FORMAT (5X, 11X, 4X, 1NUMBER OF ELEMENTS = 1, 12 / 10X,
207 1 1NUMBER OF ANALYSIS FREQUENCIES = 1, 12 / 10X, 1FIRST 1,
208 2 1ANALYSIS FREQUENCY = 1, 1PE12.5E2 / 10X, 1TYPE OF UNITS = 1,
209 3 1A1 / 10X, 1TYPE OF OUTPUT = 1, 4X)
210 540 FORMAT (10, 4X, 1ELEMENT NUMBER = 1, 12 / 10X, 1NUMBER OF 1,
211 1 1SUB-ELEMENTS = 1, 12 / 10X, 1TYPE OF EXCITATION = 1, 4X, 1A1 /
212 2 10X, 1TYPE OF MECHANICAL INPUT = 1, 4X, 10X,
213 4 1DAMPING = 1, 1PE12.5E2 / 10X, 1SLOPE = 1, 1PE12.5E2 /
214 5 10X, 1STARTING FREQUENCY = 1, 1PE12.5E2)
215 55C FORMAT (20X, 1SOUND PRESSURE LEVEL// 1, 16, 7X,
216 1 1A12X, 1PE12.5E2) / 13X, 4,1X, 1PE12.5E2))
217 560 FORMAT (20X, 1MECHANICAL INPUT// 1, 16, 7X,
218 1 1A12X, 1PE12.5E2) / 12X, 4,1X, 1PE12.5E2))
219 57C FORMAT (16, 9X, 1SUB-ELEMENT NUMBER = 1, 12 / 15X, 1TYPE OF 1,
220 1 1SUB-ELEMENT = 1, 4X, 15X, 1DENSITY = 1, 1PE12.5E2 / 15X,
221 2 1MODULUS OF ELASTICITY = 1, 1PE12.5E2 / 15X, 1THICKNESS = 1,
222 3 1PE12.5E2 / 15X, 1AREA = 1, 1PE12.5E2 / 15X, 1POISSONS 1,
223 4 1RATIO = 1, 1PE12.5E2 / 15X, 1LENGTH = 1, 1PE12.5E2 / 15X,
224 5 1PRESSURE = 1, 1PE12.5E2 / 16, 9X, 1STIFFNESS REDUCTION 1,
225 6 1REQUIRED = 1, 11 / 15X, 1RADIUS = 1, 1PE12.5E2 / 15X,
226 7 1VOLUME = 1, 1PE12.5E2 / 15X, 1SPEED OF SOUND IN ROOM 1,
227 8 1VELOCITY = 1, 1PE12.5E2 / 15X, 1ACCO PASSED = 1, 1PE12.5E2)

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220 C WRITE THE HEADING FOR THE OUTPUT
221 WRITE (6,520)
222 C READ HOW MANY ELEMENTS AND ANALYSIS FREQUENCIES THERE ARE, THE
223 C FIRST FREQUENCY, THE SYSTEM OF UNITS, AND THE TYPE OF OUTPUT
224 READ (3,380,ERR=210,END=220)NUMEL,NUMAF,FREQ1,UNITS,OTYPE
225 C WRITE THE FIRST RECORD TO OUTPUT
226 WRITE (6,520) NUMEL, NUMAF, FREQ1, UNITS, OTYPE
227 C CHECK TO SEE THAT THE NUMBER OF ELEMENTS AND ANALYSIS
228 C FREQUENCIES IS WITHIN RANGE
229 IF (NUMEL .LT. 2 .OR. NUMEL .GT. 20) GO TO 260
230 IF (NUMAF .LT. 1 .OR. NUMAF .GT. 40) GO TO 270
231 C IF THE SYSTEM OF UNITS IS METRIC, CONVERT THE GRAVITATIONAL
232 C CONSTANT TO METRIC UNITS
233 IF (UNITS .EQ. 'MT') G = 61
234 C PUT THE VALUES OF THE ANALYSIS FREQUENCIES IN THE FREQUENCY TABLE
235 DO 10 I = 1,NUMAF
236 FTAB(I) = FTAB(I) + FREQ1
237 10 CONTINUE
238 C DO FOR EACH ELEMENT
239 DO 200 I = 1,NUMEL
240 C INCREMENT INPUT
241 INPUT = INPUT + 1
242 C READ THE ELEMENT PROPERTIES
243 READ (2,310,ERR=120,END=230) ELNUM,NUMSUR,ETYPE(I),MTYPE(I),
244 * ETAB(I),SLOPE(I),SFREQ(I)
245 C WRITE THE ELEMENT PROPERTIES TO OUTPUT
246 WRITE (6,540) INPUT, ELNUM, NUMSUR, ETYPE(I), MTYPE(I),
247 * ETAB(I),SLOPE(I),SFREQ(I)
248 C IF THE ELEMENT NUMBER IS OUT OF ORDER, WRITE A WARNING
249 C MESSAGE
250 IF (ELNUM .EQ. 1) GO TO 12
251 WRITE (6,430) INPUT,ELNUM,I
252 ELNUM = I
253 C IF THE TYPE OF ELEMENT IS MECHANICAL BUT THE TYPE OF MECHANICAL
254 C INPUT IS NEITHER RMS NOR PSD, WRITE A MESSAGE AND SET THE
255 C ERROR FLAG
256 12 IF (ETYPE(I) .NE. 'MT' .OR. MTYPE(I) .EQ. 'RMS' .OR.
257 * MTYPE(I) .EQ. 'PSD') GO TO 15
258 WRITE (6,490) ELNUM,INPUT,MTYPE
259 ERROR = .TRUE.
260 C IF THE TYPE OF EXCITATION IS ACOUSTIC
261 15 IF (ETYPE(I) .NE. 'AT') GO TO 18
262 C THEN INCREMENT INPUT
263 INPUT1 = INPUT
264 LINES = (NUMAF + 7) / 8
265 INPUT = INPUT + LINES
266 C AND READ THE SOUND PRESSURE LEVEL FOR EACH FREQUENCY
267 READ (2,400,ERR=130,END=250) (SPL(I,J),J=1,NUMAF)
268 C WRITE THE SOUND PRESSURE LEVELS TO OUTPUT
269 WRITE (6,550) (INPUT1 + K, (SPL(I, 8 + (K - 1) * J),
270 * J = 1, 8), K = 1, LINES)
271 C ELSE IF THE TYPE OF EXCITATION IS MECHANICAL,
272 18 IF (ETYPE(I) .NE. 'MT') GO TO 28
273 C THEN INCREMENT INPUT
274 INPUT1 = INPUT
275 LINES = (NUMAF + 7) / 8
276 INPUT = INPUT + LINES
277 C AND READ THE SOUND PRESSURE LEVEL FOR EACH FREQUENCY
278 READ (2,400,ERR=130,END=250) (SPL(I,J),J=1,NUMAF)
279 C WRITE THE SOUND PRESSURE LEVELS TO OUTPUT
280 WRITE (6,550) (INPUT1 + K, (SPL(I, 8 + (K - 1) * J),
281 * J = 1, 8), K = 1, LINES)
282 C ELSE IF THE TYPE OF EXCITATION IS MECHANICAL,
283 18 IF (ETYPE(I) .NE. 'MT') GO TO 28
284 C THEN INCREMENT INPUT
285 INPUT1 = INPUT
286 LINES = (NUMAF + 7) / 8
287 INPUT = INPUT + LINES

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AUG26 29  
 AUG26 30  
 EPROP143  
 EPROP144  
 EPROP145  
 AUG26 31  
 AUG26 32  
 EPROP146  
 EPROP147  
 EPROP148  
 EPROP149  
 EPROP150  
 EPROP151  
 EPROP152  
 MAY13 18  
 EPROP154  
 EPROP155  
 EPROP156  
 EPROP157  
 EPROP158  
 EPROP159  
 EPROP160  
 EPROP161  
 MAY13 19  
 OCT28 4  
 AUG26 33  
 AUG26 34  
 NOV4 3  
 EPROP164  
 EPROP165  
 EPROP166  
 EPROP167  
 EPROP168  
 EPROP169  
 EPROP170  
 EPROP171  
 EPROP172  
 EPROP173  
 EPROP174  
 EPROP175  
 MAY13 20  
 EPROP177  
 EPROP178  
 AUG26 36  
 AUG26 37  
 AUG26 38  
 EPROP180  
 MAY13 21  
 AUG26 39  
 AUG26 40  
 AUG26 41  
 EPROP182  
 EPROP183  
 EPROP184  
 AUG26 42  
 AUG26 43



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340      GO TO 100
341      C ELSE IF THE TYPE OF SUB-ELEMENT IS NONE OF THE ABOVE, WRITE AN
342      C ERROR MESSAGE AND SET THE ERROR FLAG TO TERMINATE THE PROGRAM
343      7C      WRITE (6,330) TYPE,INPUT-1
344      ERROR = .TRUE.
345      GO TO 100
346
347      C END IF
348
349      C IF AN ERROR WAS ENCOUNTERED WHILE READING SUB-ELEMENT PROPERTIES,
350      C WRITE A MESSAGE AND SET THE ERROR FLAG
351      8C      WRITE (6,340) INPUT-1,INPUT
352      ERROR = .TRUE.
353      GO TO 100
354
355      10C      CONTINUE
356      GO TO 200
357
358      C IF AN ERROR WAS ENCOUNTERED WHILE READING ELEMENT PROPERTIES,
359      C WRITE A MESSAGE AND SET THE ERROR FLAG
360      12C      WRITE (6,350) INPUT
361      ERROR = .TRUE.
362      GO TO 200
363
364      C IF AN ERROR WAS ENCOUNTERED WHILE READING SOUND PRESSURE LEVELS,
365      C WRITE A MESSAGE AND SET THE ERROR FLAG
366      13C      WRITE (6,410) INPUT
367      ERROR = .TRUE.
368      GO TO 20
369
370      C IF AN ERROR WAS ENCOUNTERED WHILE READING MECHANICAL INPUTS,
371      C WRITE A MESSAGE AND SET THE ERROR FLAG
372      14C      WRITE (6,470) INPUT
373      ERROR = .TRUE.
374      GO TO 20
375
376      20C      CONTINUE
377      DO 207 K = 1, NMELE, 4
378          KPLUS3 = MIN(NMELE, K + 3)
379          WRITE (6,500) (I,I=N,KPLUS3)
380          DO 207 J2 = 1, NMAF
381              WRITE(6,510) F1AB(J2),(N(J,42),J=N,KPLUS3)
382
383      207      CONTINUE
384      IF (ERROR) RETURN 1
385      RETURN
386
387      C IF AN ERROR OCCURRED WHILE READING THE FIRST INPUT RECORD,
388      C WRITE A MESSAGE AND SET THE ERROR FLAG
389      21C      WRITE (6,360)
390      GO TO 285
391
392      C IF AN END OF FILE WAS ENCOUNTERED WHILE TRYING TO READ THE
393      C FIRST INPUT RECORD, WRITE A MESSAGE AND SET THE ERROR FLAG
394      22C      WRITE (6,370)
395      GO TO 285
396
397      C IF AN END OF FILE WAS ENCOUNTERED WHILE READING ELEMENT PROPERTIES,
398      C WRITE A MESSAGE AND SET THE ERROR FLAG
399      23C      WRITE (6,380) IN:UT
400      GO TO 285
401
402      C IF AN END OF FILE WAS ENCOUNTERED WHILE READING SUB-ELEMENT
403      C PROPERTIES, WRITE A MESSAGE AND SET THE ERROR FLAG
404      24C      WRITE (6,350) INPUT
405      GO TO 285
406
407      C IF AN END OF FILE WAS ENCOUNTERED WHILE READING SOUND PRESSURE
408      C LEVELS, WRITE A MESSAGE AND SET THE ERROR FLAG
409      25C      WRITE (6,420) INPUT

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EPROP222  
 EPROP223  
 EPROP224  
 EPROP225  
 EPROP226  
 EPROP227  
 EPROP228  
 EPROP229  
 EPROP230  
 EPROP231  
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 EPROP233  
 EPROP234  
 EPROP235  
 EPROP236  
 EPROP237  
 EPROP238  
 EPROP239  
 EPROP240  
 EPROP241  
 EPROP242  
 EPROP243  
 EPROP244  
 EPROP245  
 EPROP246  
 EPROP247  
 EPROP248  
 EPROP249  
 EPROP250  
 EPROP251  
 JUNE25 6  
 JUNE25 7  
 AUG12J 1  
 AUG24 4  
 EPROP257  
 EPROP258  
 EPROP259  
 EPROP260  
 EPROP261  
 EPROP262  
 JUNE25 9  
 EPROP265  
 EPROP266  
 EPROP267  
 EPROP268  
 EPROP269  
 EPROP270  
 EPROP271  
 EPROP272  
 EPROP273  
 EPROP274  
 EPROP275  
 EPROP276  
 EPROP277  
 EPROP278  
 EPROP279

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396      GO TO 285
397      C IF THE NUMBER OF ELEMENTS IS OUT OF RANGE, WRITE A
398      C MESSAGE AND SET THE ERROR FLAG
399      260 WRITE (6,450) NUMEL
400      GO TO 285
401      C IF THE NUMBER OF ANALYSIS FREQUENCIES IS OUT OF RANGE, WRITE A
402      C MESSAGE AND SET THE ERROR FLAG
403      270 WRITE (6,460) NUMAF
404      GO TO 285
405      C IF AN END OF FILE WAS ENCOUNTERED WHILE READING MECHANICAL
406      C IMPLTS, WRITE A MESSAGE AND SET THE ERROR FLAG
407      280 WRITE (6,480) INPUT
408      285 ERROR = .TRUE.
409      RETURN 1
410      ENDC
EPROP280
EPROP281
EPROP282
EPROP283
EPROP284
EPROP285
EPROP286
EPROP287
EPROP288
EPROP289
EPROP290
EPROP291
EPROP292
EPROP293
EPROP294

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411 SUPERCLINE PFAM
412 C THIS SUBROUTINE CALCULATES THE MODAL DENSITY FOR A SUR-ELEMENT
413 C WHICH IS A BEAM AND SUMS THIS VALUE TO THE ELEMENT MODAL
414 C DENSITY. PFAM IS CALLED FROM EPROP.
415 C
416 C THE FOLLOWING VARIABLES ARE USED IN THIS PROGRAM UNIT:
417 C NAME TYPE DESCRIPTION
418 C ----
419 C A REAL SUR-ELEMENT AREA
420 C AREA(20) REAL ELEMENT AREA
421 C C REAL SPEED OF SOUND IN SUR-ELEMENT ROOM MEDIUM
422 C DENSE(20) REAL ELEMENT DENSITY
423 C F REAL SUB-ELEMENT MODULUS OF ELASTICITY
424 C FC(20) REAL SUB-ELEMENT MODULUS OF ELASTICITY
425 C EC(20) REAL SPEED OF SOUND IN ELEMENT ROOM MEDIUM
426 C EGAPPA(20) REAL POISSON'S RATIO FOR ELEMENT
427 C ENUF INT ELEMENT NUMBER
428 C FTAB(40) REAL TABLE OF ANALYSIS FREQUENCIES
429 C GAMPA REAL POISSON'S RATIO FOR SUB-ELEMENT
430 C L REAL LENGTH
431 C M REAL SUB-ELEMENT ADDED MASS (NON-STRUCTURAL)
432 C MASS(20) REAL ELEMENT MASS
433 C MSUB REAL SUB-ELEMENT TOTAL MASS
434 C N(20,40) REAL ELEMENT MODAL DENSITY
435 C NUFAF INT NUMBER OF ANALYSIS FREQUENCIES
436 C OMEGA REAL 2*PI*FREQUENCY
437 C PSUBPD REAL PARTIAL SUB-ELEMENT MODAL DENSITY
438 C P REAL RADIUS
439 C RMO REAL SUB-ELEMENT DENSITY
440 C S REAL PRESSURE
441 C SUBPD REAL SUB-ELEMENT MODAL DENSITY
442 C T REAL SUB-ELEMENT THICKNESS
443 C THICK(20) REAL ELEMENT THICKNESS
444 C V REAL SUB-ELEMENT VOLUME
445 C VOL(20) REAL ELEMENT VOLUME
446 C
447 C
448 C THE FOLLOWING COMMON BLOCKS ARE USED:
449 C BLOCK OTHER PROGRAM UNITS USING THIS COMMON BLOCK
450 C -----
451 C CP1 MAIN,EPROP,MEPR,PLATE,ROOM,CYLIN,RIET, BLOCK DATA
452 C CB4 EPROP,MEPR,PLATE,ROOM,CYLIN
453 C CB5 EPROP,MEPR,PLATE,ROOM,CYLIN,JPROP,EXCITE,ANSWER
454 C CB3 EPROP,MEPR,PLATE,ROOM,CYLIN,JPROP,EXCITE,BLOCK DATA
455 C CB11 MEPR,PLATE,ROOM,CYLIN,ANSWER,BLOCK DATA
456 C
457 C
458 C INTEGER ELNUM
459 C REAL M1,M2,MASS,PSUB
460 C LOGICAL STIFF,MAIN
461 C COMMON /CB1/ FTAB(40),NUMAF,FREQ1
462 C COMMON /CB4/ ENUF,TA,F,GAMPA,RMO,M,STIFF,L,A,V,C,R,S,MAIN
463 C COMMON /CB5/ THICK(20),AREA(20),DENSE(20),VOL(20),EC(20),
464 C EGAPPA(20),EC(20)
465 C COMMON /CB9/ M(20,40)
466 C COMMON /CB11/ MASS(20)

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467 DATA PI /3.141592/
468 C CALCULATE THE PART OF THE SUB-ELEMENT MODAL DENSITY THAT IS
469 C NOT FREQUENCY DEPENDENT
470 FSLRPO = L / (2. * PI) * SORT(SCRT(12. * RHO / E) / T)
471 C IF STIFFNESS REDUCTION IS REQUIRED, MULTIPLY BY SORT(.5)
472 IF (STIFF) PSURMD = SORT(.5) * PSURMD
473 C DO FOR EACH ANALYSIS FREQUENCY
474 DO 20 I = 1,NUMAF
475 OMEGA = 2. * PI * FIAB(I)
476 C CALCULATE THE SUB-ELEMENT MODAL DENSITY
477 SUBMD = PSURMD / SORT(OMEGA)
478 C SUM THE SUB-ELEMENT MODAL DENSITY TO THE ELEMENT MODAL DENSITY
479 M(ELNUM,I) = M(ELNUM,I) + SUBMD
480 20 CONTINUE
481 C SUM THE SUB-ELEMENT MASS TO THE ELEMENT MASS
482 PSUM = RHO * A * L * M
483 MASS(ELNUM) = MASS(ELNUM) + PSUM
484 C IF THIS IS NOT THE MAIN SUB-ELEMENT, RETURN TO EPROP
485 IF (.NOT. MAIN) RETURN
486 C ELSE PUT THE VALUES OF THE FOLLOWING SUB-ELEMENT VARIABLES INTO
487 C THE CORRESPONDING ELEMENT ARRAYS
488 EE(ELNUM) = E
489 THICK(ELNUM) = T
490 DEASE(ELNUM) = RHO
491 AREA(ELNUM) = A
492 VOL(ELNUM) = V
493 EGAMMA(ELNUM) = GAMMA
494 EG(ELNUM) = C
495 C SET MAIN TO INDICATE THAT ANY FOLLOWING SUB-ELEMENTS ARE NOT
496 C THE MAIN SUB-ELEMENT
497 MAIN = .FALSE.
498 RETURN
499 ENC

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BEAM 57
BEAM 58
BEAM 59
MAY13 29
BEAM 61
BEAM 62
BEAM 63
BEAM 64
BEAM 65
BEAM 66
BEAM 67
BEAM 68
MAY13 30
BEAM 70
BEAM 71
OCT16 4
OCT16 5
BEAM 73
BEAM 74
BEAM 75
BEAM 76
BEAM 77
BEAM 78
BEAM 79
BEAM 80
BEAM 81
BEAM 82
BEAM 83
BEAM 84
JUNE2510
BEAM 86
BEAM 87
BEAM 98

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500 CURROUTINE MEMBR
501 C THIS SUBROUTINE CALCULATES THE MODAL DENSITY FOR A SUB-ELEMENT
502 C WHICH IS A MEMBRANE AND SUMS THIS VALUE TO THE ELEMENT MODAL
503 C DENSITY. MEMBR IS CALLED FROM EPROP.
504 C
505 C THE FOLLOWING VARIABLES ARE USED IN THIS PROGRAM UNIT:
506 C NAME TYPE DESCRIPTION
507 C ----
508 C A REAL SUB-ELEMENT AREA
509 C AREA(20) REAL ELEMENT AREA
510 C C REAL SPEED OF SOUND IN SUB-ELEMENT ROOM MEDIUM
511 C DENSE(20) REAL ELEMENT DENSITY
512 C E REAL SUB-ELEMENT MODULUS OF ELASTICITY
513 C FC(20) REAL SPEED OF SOUND IN ELEMENT ROOM MEDIUM
514 C EE(20) REAL ELEMENT MODULUS OF ELASTICITY
515 C EGAMMA(20) REAL POISSON'S RATIO FOR ELEMENT
516 C ELNUM INT ELEMENT NUMBER
517 C FTAB(40) REAL TABLE OF ANALYSIS FREQUENCIES
518 C GAMMA REAL POISSON'S RATIO FOR SUB-ELEMENT
519 C L REAL LENGTH
520 C M REAL SUB-ELEMENT ADDED MASS (NON-STRUCTURAL)
521 C MASS(20) REAL ELEMENT MASS
522 C PSUB REAL SUB-ELEMENT TOTAL MASS
523 C N(20,40) REAL ELEMENT MODAL DENSITY
524 C NUMAF INT NUMBER OF ANALYSIS FREQUENCIES
525 C OMEGA REAL 2*PI*FREQUENCY
526 C PSURPB REAL PARTIAL SUB-ELEMENT MODAL DENSITY
527 C R REAL RADIUS
528 C RMO REAL SUB-ELEMENT DENSITY
529 C S REAL PRESSURE
530 C SUPMO REAL SUB-ELEMENT MODAL DENSITY
531 C T REAL SUB-ELEMENT THICKNESS
532 C THICK(20) REAL ELEMENT THICKNESS
533 C V REAL SUB-ELEMENT VOLUME
534 C VOL(20) REAL ELEMENT VOLUME
535 C
536 C
537 C THE FOLLOWING COMMON BLOCKS ARE USED:
538 C BLOCK OTHER PROGRAM UNITS USING THIS COMMON BLOCK
539 C -----
540 C CB1 MAIN,EPROP,BEAM,PLATE,ROOM,CYLIN,ITER,BLOCK DATA
541 C CB4 EPROP,BEAM,PLATE,ROOM,CYLIN
542 C CB5 EPROP,BEAM,PLATE,ROOM,CYLIN,JPROP,EXCITE,ANSWER
543 C CB9 EPROP,BEAM,PLATE,ROOM,CYLIN,JPROP,EXCITE,BLOCK DATA
544 C CB11 BEAM,PLATE,ROOM,CYLIN,ANSWER,BLOCK DATA
545 C
546 C
547 INTEGER ELNUM
548 REAL NAL,M,MASS,MSUB
549 LOGICAL STIFF,MAIN
550 COMMON /CB1/ FTAB(40),ALMAF,FRECI
551 COMMON /CB4/ ELNUM,T,E,GAMMA,RMO,M,STIFF,L,A,V,C,R,S,MAIN
552 COMMON /CB5/ THICK(20),AREA(20),DENSE(20),VOL(20),EE(20),
553 , EGAMMA(20),EC(20)
554 COMMON /CB9/ N(20,40)
555 COMMON /CB11/ MASS(20)

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MEMBR 2  
MEMBR 3  
MEMBR 4  
MEMBR 5  
MEMBR 6  
MEMBR 7  
MEMBR 8  
MEMBR 9  
MEMBR 10  
MEMBR 11  
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MEMBR 13  
MEMBR 14  
MEMBR 15  
MEMBR 16  
MEMBR 17  
MEMBR 18  
MEMBR 19  
MEMBR 20  
MEMBR 21  
OCT16 6  
MEMBR 23  
OCT16 7  
MEMBR 24  
MEMBR 25  
MEMBR 26  
MEMBR 27  
MEMBR 28  
MEMBR 29  
MEMBR 30  
MEMBR 31  
MEMBR 32  
MEMBR 33  
MEMBR 34  
MEMBR 35  
MEMBR 36  
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MEMBR 38  
MEMBR 39  
MEMBR 40  
MEMBR 41  
MEMBR 42  
MEMBR 43  
MEMBR 44  
MEMBR 45  
MEMBR 46  
MEMBR 47  
MAY13 31  
OCT16 8  
MEMBR 50  
MEMBR 51  
MEMBR 52  
MEMBR 53  
MEMBR 54  
MEMBR 55  
MEMBR 56



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556 DATA PI /3.1415924
557 C CALCULATE THE PART OF THE SUB-ELEMENT MODAL DENSITY THAT IS
558 C NOT FREQUENCY DEPENDENT
559 PSUMD = A * RHO * T / (2. * PI * S)
560 C IF STIFFNESS REDUCTION IS REQUIRED, MULTIPLY BY SORT(.5)
561 IF (STIFF) PSUMD = SORT(.5) * PSUMD
562 C DO FOR EACH ANALYSIS FREQUENCY
563 DO 20 I = 1,NUMAF
564 OMEGA = 2. * PI * FIAB(I)
565 C CALCULATE THE SUB-ELEMENT MODAL DENSITY
566 SUBMD = PSUMD * OMEGA
567 C SUM THE SUB-ELEMENT MODAL DENSITY TO THE ELEMENT MODAL DENSITY
568 N(ELNUM,I) = N(ELNUM,I) + SUBMD
569 20 CONTINUE
570 C SUM THE SUB-ELEMENT MASS TO THE ELEMENT MASS
571 MSUB = RHO * A * T * M
572 MASS(ELNUM) = MASS(ELNUM) + MSUB
573 C IF THIS IS NOT THE MAIN SUB-ELEMENT, RETURN TO EPROP
574 IF (.NOT. MAIN) RETURN
575 C ELSE PUT THE VALUES OF THE FOLLOWING SUB-ELEMENT VARIABLES INTO
576 C THE CORRESPONDING ELEMENT ARRAYS
577 EE(ELNUM) = E
578 THICK(ELNUM) = T
579 DENS(ELNUM) = RHO
580 AREA(ELNUM) = A
581 VOL(ELNUM) = V
582 EGAMMA(ELNUM) = GAMMA
583 EC(ELNUM) = C
584 C SET MAIN TO INDICATE THAT ANY FOLLOWING SUB-ELEMENTS ARE NOT
585 C THE MAIN SUB-ELEMENT
586 MAIN = .FALSE.
587 RETURN
588 ENC

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```

MEMBR 57
MEMBR 58
MEMBR 59
MEMBR 60
MEMBR 61
MEMBR 62
MEMBR 63
MEMBR 64
MEMBR 65
MEMBR 66
MEMBR 67
MEMBR 68
MEMBR 69
MEMBR 70
MEMBR 71
OCT16 8
OCT16 10
MEMBR 73
MEMBR 74
MEMBR 75
MEMBR 76
MEMBR 77
MEMBR 78
MEMBR 79
MEMBR 80
MEMBR 81
MEMBR 82
MEMBR 83
MEMBR 84
MEMBR 85
MEMBR 86
MEMBR 87
MEMBR 88

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580 SUBROUTINE PLATE
590 C THIS SUBROUTINE CALCULATES THE MODAL DENSITY FOR A SUR-ELEMENT
591 C WHICH IS A PLATE AND SUMS THIS VALUE TO THE ELEMENT MODAL
592 C DENSITY. PLATE IS CALLED FROM EPROP.
593 C
594 C THE FOLLOWING VARIABLES ARE USED IN THIS PROGRAM UNIT:
595 C
596 C NAME TYPE DESCRIPTION
597 C ----
598 C A REAL SUB-ELEMENT AREA
599 C ANP(20) REAL (AREA * MODES / MASS) OF SURFACE EXCITED BY
600 C ACOUSTIC FIELD
601 C AREA(20) REAL ELEMENT AREA
602 C C SPEED OF SOUND IN SUR-ELEMENT ROOM MEDIUM
603 C DENSE(20) REAL ELEMENT DENSITY
604 C EC(20) REAL SUR-ELEMENT MODULUS OF ELASTICITY
605 C FE(20) REAL SPEED OF SOUND IN ELEMENT ROOM MEDIUM
606 C EGAPPA(20) REAL POISSON'S RATIO OF ELASTICITY
607 C ELNLP INT ELEMENT NUMBER
608 C FTAP(40) REAL TABLE OF ANALYSIS FREQUENCIES
609 C GAMMA REAL POISSON'S RATIO FOR SUR-ELEMENT
610 C L REAL LENGTH
611 C P REAL SUR-ELEMENT ADDED MASS (NON-STRUCTURAL)
612 C MASS(20) REAL ELEMENT MASS
613 C MSUR REAL SUB-ELEMENT TOTAL MASS
614 C N(20,40) REAL ELEMENT MODAL DENSITY
615 C NUPF INT NUMBER OF ANALYSIS FREQUENCIES
616 C OMEGA REAL 2*PI*FREQUENCY
617 C PSURPD REAL PARTIAL SUB-ELEMENT MODAL DENSITY
618 C R REAL RADIUS
619 C RMC REAL SUB-ELEMENT DENSITY
620 C S REAL PRESSURE
621 C SUBMD REAL SUB-ELEMENT MODAL DENSITY
622 C T REAL SUB-ELEMENT THICKNESS
623 C THICK(20) REAL ELEMENT THICKNESS
624 C V REAL SUB-ELEMENT VOLUME
625 C VOL(20) REAL ELEMENT VOLUME
626 C
627 C
628 C THE FOLLOWING COMMON BLOCKS ARE USED:
629 C
630 C BLOCK OTHER PROGRAM UNITS USING THIS COMMON BLOCK
631 C
632 C CP1 MAIN,EPROP,BFAM,MEPR,ROOM,CYLIN,ITER,BLOCK DATA
633 C CP4 EPROP,BEAM,MEPR,ROCK,CYLIN
634 C CH5 EPROP,BEAM,MEPR,ROCK,CYLIN,JPROP,EXCITE,ANSWER
635 C CR6 EPROP,EXCITE
636 C CP5 EPROP,BEAM,MEPR,ROCK,CYLIN,JPROP,EXCITE,BLOCK DATA
637 C CF11 BEAM,MEPR,ROCK,CYLIN,ANSWER,PLCK DATA
638 C
639 C
640 C INTEGER ELNUP
641 C CHARACTER ETYP,MTYPE*1
642 C REAL NOL,M,MASS,MSUR
643 C LOGICAL STIFF,MAIN
644 C COMMON /CP1/ FTAP(40),NUPF,FRECI
645 C COMMON /CP4/ ELNUP,T,E,GAMMA,RMC,M,STIFF,L,A,V,C,R,S,MAIN
646 C

```

PLATE 2  
 PLATE 3  
 PLATE 4  
 PLATE 5  
 PLATE 6  
 PLATE 7  
 PLATE 8  
 PLATE 9  
 PLATE 10  
 OCT28 9  
 OCT28 10  
 PLATE 11  
 PLATE 12  
 PLATE 13  
 PLATE 14  
 PLATE 15  
 PLATE 16  
 PLATE 17  
 PLATE 18  
 PLATE 19  
 PLATE 20  
 PLATE 21  
 OCT16 11  
 PLATE 23  
 OCT16 12  
 PLATE 24  
 PLATE 25  
 PLATE 26  
 PLATE 27  
 PLATE 28  
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 PLATE 31  
 PLATE 32  
 PLATE 33  
 PLATE 34  
 PLATE 35  
 PLATE 36  
 PLATE 37  
 MAY13 32  
 PLATE 39  
 PLATE 40  
 PLATE 41  
 PLATE 42  
 PLATE 43  
 OCT28 11  
 PLATE 44  
 PLATE 45  
 PLATE 46  
 PLATE 47  
 PLATE 48  
 OCT28 12  
 OCT16 13  
 PLATE 50  
 PLATE 51  
 PLATE 52

```

645 COMMON /CHE/ THICK(20),AREA(20),DENSE(20),VOL(20),FE(20),
646 * ECGMA(20),EC(20)
647 COMMON /CRF/ SPL(20,40),MECH(20,40),ANM(20),ETYPE(20),MYPE(20)
648 COMMON /CB9/ N(20,40)
649 COMMON /CB11/ MASS(20)
650 DATA PI /3.141592/
651 C CALCULATE THE SUB-ELEMENT MODAL DENSITY
652 * SUFFC = A / (4. * PI * SORT(12. * RHO * (1. - GAMMA ** 2)
653 * / (E * T ** 2)))
654 C IF STIFFNESS REDUCTION IS REQUIRED, MULTIPLY BY SORT(.5)
655 IF (STIFF) SURMD = SORT(.5) * SURMD
656 C DO FOR EACH ANALYSIS FREQUENCY
657 DO 20 I = 1,NUMAF
658 C SUM THE SUB-ELEMENT MODAL DENSITY TO THE ELEMENT MODAL DENSITY
659 * NCELMN(I) = NCELMN(I) + SURMD
660 C 20 CONTINUE
661 C SUM THE SUB-ELEMENT MASS TO THE ELEMENT MASS
662 * MSCB = RHO * A * T * M
663 * MASSCELMN(I) = MASSCELMN(I) + *SUR
664 IF (ETYPECELMN(I) .EQ. TAT .AND. MAIN) ANMCELMN(I) =
665 * ((A * SURMD) / MSCB) * 2. * PI
666 C IF THIS IS NOT THE MAIN SUB-ELEMENT, RETURN TO EPROP
667 IF (.NOT. MAIN) RETURN
668 C ELSE PUT THE VALUES OF THE FOLLOWING SUB-ELEMENT VARIABLES INTO
669 C THE CORRESPONDING ELEMENT ARRAYS
670 * EECELMN(I) = E
671 * THICKCELMN(I) = T
672 * DENSECELMN(I) = RHO
673 * AREACELMN(I) = A
674 * VOLCELMN(I) = V
675 * ECGMACELMN(I) = GAMMA
676 * ECCELMN(I) = C
677 C SET MAIN TO INDICATE THAT ANY FOLLOWING SUB-ELEMENTS ARE NOT
678 C THE MAIN SUB-ELEMENT
679 * MAIN = .FALSE.
680 RETURN
681 END

```

PLATE 53  
 PLATE 54  
 UNVAC2 2  
 MAY13 13  
 PLATE 56  
 PLATE 57  
 PLATE 58  
 PLATE 59  
 PLATE 60  
 PLATE 61  
 JULY16G1  
 PLATE 63  
 PLATE 64  
 PLATE 65  
 MAY13 14  
 PLATE 67  
 PLATE 68  
 OCT16 14  
 OCT16 15  
 NOV2 1  
 NOV2 2  
 PLATE 70  
 PLATE 71  
 PLATE 72  
 PLATE 73  
 PLATE 74  
 PLATE 75  
 PLATE 76  
 PLATE 77  
 PLATE 78  
 PLATE 79  
 PLATE 80  
 PLATE 81  
 PLATE 82  
 PLATE 83  
 PLATE 84  
 PLATE 85

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622      CURRCUTIME ROOM
623      C THIS SUBROUTINE CALCULATES THE MODAL DENSITY FOR A SUB-ELEMENT
624      C WHICH IS A ROOM AND SUMS THIS VALUE TO THE ELEMENT MODAL
625      C DENSITY. ROOM IS CALLED FROM EPROP.
626      C
627      C THE FOLLOWING VARIABLES ARE USED IN THIS PROGRAM UNIT:
628      C NAME TYPE DESCRIPTION
629      C ----
630      C A REAL SUR-ELEMENT AREA
631      C AREA(20) REAL ELEMENT AREA
632      C C REAL SPEED OF SOUND IN SUB-ELEMENT ROOM MEDIUM
633      C DENSE(20) REAL ELEMENT DENSITY
634      C E REAL SUB-ELEMENT MODULUS OF ELASTICITY
635      C EC(20) REAL SPEED OF SOUND IN ELEMENT ROOM MEDIUM
636      C EE(20) REAL ELEMENT MODULUS OF ELASTICITY
637      C EGAPPA(20) REAL POISSON'S RATIO FOR ELEMENT
638      C ELNLP INT ELEMENT NUMBER
639      C FTAR(40) REAL TABLE OF ANALYSIS FREQUENCIES
640      C GAPPA REAL POISSON'S RATIO FOR SUB-ELEMENT
641      C L REAL LENGTH
642      C P REAL SUB-ELEMENT ADDED MASS (A0A-STRUCTURAL)
643      C MASS(20) REAL ELEMENT MASS
644      C MSUR REAL SUB-ELEMENT TOTAL MASS
645      C N(20,40) REAL ELEMENT MODAL DENSITY
646      C NUPIF INT NUMBER OF ANALYSIS FREQUENCIES
647      C OMEGA REAL 2*PI*FREQUENCY
648      C PSURPD REAL PARTIAL SUB-ELEMENT MODAL DENSITY
649      C R REAL RADIUS
650      C RMC REAL SUB-ELEMENT DENSITY
651      C S REAL PRESSURE
652      C SUBMD REAL SUB-ELEMENT MODAL DENSITY
653      C T REAL SUB-ELEMENT THICKNESS
654      C THICK(20) REAL ELEMENT THICKNESS
655      C V REAL SUB-ELEMENT VOLUME
656      C VOL(20) REAL ELEMENT VOLUME
657      C
658      C THE FOLLOWING COMMON BLOCKS ARE USED:
659      C BLOCK OTHER PROGRAM UNITS USING THIS COMMON BLOCK
660      C -----
661      C CB1 MAIN, EPROP, BEAM, MEMBR, PLATE, CYLIN, RITER, BLOCK DATA
662      C CB4 EPROP, BEAM, MEMBR, PLATE, CYLIN
663      C CB5 EPROP, BEAM, MEMBR, PLATE, CYLIN, JPROP, EXCITE, ANSWER
664      C CB6 EPROP, BEAM, MEMBR, PLATE, CYLIN, JPROP, EXCITE, BLOCK DATA
665      C CB11 BEAM, MEMBR, PLATE, CYLIN, ANSWER, BLOCK DATA
666      C
667      C
668      C
669      C
670      C
671      C
672      C
673      C
674      C
675      C
676      C
677      C
678      C
679      C
680      C
681      C
682      C
683      C
684      C
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692      C
693      C
694      C
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820      C
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824      C
825      C
826      C
827      C
828      C
829      C
830      C
831      C
832      C
833      C
834      C
835      C
836      C
837      C
838      C
839      C
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841      C
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850      C
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861      C
862      C
863      C
864      C
865      C
866      C
867      C
868      C
869      C
870      C
871      C
872      C
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897      C
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904      C
905      C
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910      C
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920      C
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930      C
931      C
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942      C
943      C
944      C
945      C
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976      C
977      C
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979      C
980      C
981      C
982      C
983      C
984      C
985      C
986      C
987      C
988      C
989      C
990      C
991      C
992      C
993      C
994      C
995      C
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998      C
999      C
1000     C

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750 DATA PI /3.141592/
751 C CALCULATE THE PART OF THE SUB-ELEMENT MODAL DENSITY THAT IS
752 C NOT FREQUENCY DEPENDENT
753 FSCRPD = V / (2. * PI ** 2 * C ** 4)
754 C DO FOR EACH ANALYSIS FREQUENCY
755 DO 20 I = 1,NUMAF
756 OMEGA = 2. * PI * FTHICK(I)
757 C CALCULATE THE SUB-ELEMENT MODAL DENSITY
758 SUBPD = OMEGA ** 2 * PSUBPD
759 C SUM THE SUB-ELEMENT MODAL DENSITY TO THE ELEMENT MODAL DENSITY
760 NCELMUM(I) = MEELMUM(I) + SUBPD
761 20 CONTINUE
762 C SUM THE SUB-ELEMENT MASS TO THE ELEMENT MASS
763 MSUM = MH0 + V
764 MASS(EELMUM) = MASS(EELMUM) + MSUM
765 C IF THIS IS NOT THE MAIN SUB-ELEMENT, RETURN * FPROP
766 IF (.NOT. PAIN) RETURN
767 C ELSE PUT THE VALUES OF THE FOLLOWING SUB-ELEMENT VARIABLES INTO
768 C THE CORRESPONDING ELEMENT ARRAYS
769 EE(EELMUM) = F
770 THICK(EELMUM) = T
771 DENSE(EELMUM) = RHO
772 AREA(EELMUM) = A
773 VCL(EELMUM) = V
774 GAMPA(EELMUM) = GAMMA
775 EC(EELMUM) = C
776 C SET PAIN TO INDICATE THAT ANY FOLLOWING SUB-ELEMENTS ARE NOT
777 C THE MAIN SUB-ELEMENT
778 PAIN = .FALSE.
779 RETURN
780 ENC
781 ROOM 57
782 ROOM 58
783 ROOM 59
784 ROOM 60
785 ROOM 61
786 ROOM 62
787 ROOM 63
788 ROOM 64
789 ROOM 65
790 ROOM 66
791 ROOM 67
792 ROOM 68
793 ROOM 69
794 ROOM 70
795 ROOM 71
796 ROOM 72
797 ROOM 73
798 ROOM 74
799 ROOM 75
800 ROOM 76
801 ROOM 77
802 ROOM 78
803 ROOM 79
804 ROOM 80
805 ROOM 81
806 ROOM 82
807 ROOM 83
808 ROOM 84
809 ROOM 85
810 ROOM 86
811 ROOM 87
812 ROOM 88
813 ROOM 89
814 ROOM 90
815 ROOM 91
816 ROOM 92
817 ROOM 93
818 ROOM 94
819 ROOM 95
820 ROOM 96
821 ROOM 97
822 ROOM 98
823 ROOM 99

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767 SUBROUTINE CYLIN
770 C THIS SUBROUTINE CALCULATES THE MODAL DENSITY FOR A SUB-ELEMENT
771 C WHICH IS A CYLINDER AND SUMS THIS VALUE TO THE ELEMENT MODAL
772 C DENSITY. CYLIN IS CALLED FROM EPROP.
773
774 C THE FOLLOWING VARIABLES ARE USED IN THIS PROGRAM UNIT:
775 C NAME TYPE DESCRIPTION
776 C ----
777 C A REAL SUB-ELEMENT AREA
778 C AREA(20) REAL ELEMENT AREA
779 C C REAL SPEED OF SOUND IN SUB-ELEMENT ROOM MEDIUM
780 C DENSE(20) REAL ELEMENT DENSITY
781 C E REAL SUB-ELEMENT MODULUS OF ELASTICITY
782 C EL(20) REAL SPEED OF SOUND IN ELEMENT ROOM MEDIUM
783 C EE(20) REAL ELEMENT MODULUS OF ELASTICITY
784 C EGAPMA(20) REAL POISSON'S RATIO FOR ELEMENT
785 C ELNUM INT ELEMENT NUMBER
786 C FTAB(4) REAL TABLE OF ANALYSIS FREQUENCIES
787 C GAPPA REAL POISSON'S RATIO FOR SUB-ELEMENT
788 C L REAL LENGTH
789 C M REAL SUB-ELEMENT ADDED MASS (NON-STRUCTURAL)
790 C MASS(20) REAL ELEMENT MASS
791 C PSUB REAL SUB-ELEMENT TOTAL MASS
792 C N(20,40) REAL ELEMENT MODAL DENSITY
793 C NUMAF INT NUMBER OF ANALYSIS FREQUENCIES
794 C OMEGA REAL 2*PI*FREQUENCY
795 C PSURPD REAL PARTIAL SUB-ELEMENT MODAL DENSITY
796 C R REAL RADIUS
797 C RHO REAL SUB-ELEMENT DENSITY
798 C S REAL PRESSURE
799 C SUPPD REAL SUB-ELEMENT MODAL DENSITY
800 C T REAL SUB-ELEMENT THICKNESS
801 C THICK(20) REAL ELEMENT THICKNESS
802 C V REAL SUB-ELEMENT VOLUME
803 C VOL(20) REAL ELEMENT VOLUME
804 C
805 C THE FOLLOWING COMMON BLOCKS ARE USED:
806 C BLOCK OTHER PROGRAM UNITS USING THIS COMMON BLOCK
807 C ----
808 C CB1 MAIN,EPROP,BEAM,MEMBR,PLATE,ROOM,ITER,BLOCK DATA
809 C CB4 EPROP,TEAM,MEMBR,PLATE,ROOM
810 C CB5 EPROP,TEAM,MEMBR,PLATE,ROOM,JPROP,EXCITE,ANSWER
811 C CB5 EPROP,TEAM,MEMBR,PLATE,ROOM,JPROP,EXCITE,BLOCK DATA
812 C CB11 BEAM,MEMBR,PLATE,ROOM,ANSWER,BLOCK DATA
813 C
814 C
815 C
816 C
817 C
818 C
819 C
820 C
821 C
822 C
823 C
824 C

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CYLIN 2
NOVA 4
CYLIN 4
CYLIN 5
CYLIN 6
CYLIN 7
CYLIN 8
MAY13 42
MAY13 43
CYLIN 11
CYLIN 12
CYLIN 13
CYLIN 14
CYLIN 15
CYLIN 16
MAY13 44
CYLIN 18
CYLIN 19
CYLIN 20
CYLIN 21
OCT16 21
CYLIN 23
OCT16 22
CYLIN 24
CYLIN 25
CYLIN 26
CYLIN 27
CYLIN 28
CYLIN 29
CYLIN 30
CYLIN 31
CYLIN 32
CYLIN 33
CYLIN 34
CYLIN 35
CYLIN 36
CYLIN 37
CYLIN 38
CYLIN 39
CYLIN 40
MAY13 45
CYLIN 42
CYLIN 43
CYLIN 44
MAY13 46
CYLIN 46
CYLIN 47
CYLIN 48
OCT16 23
CYLIN 50
CYLIN 51
CYLIN 52
CYLIN 53
CYLIN 54
MAY13 47
CYLIN 56

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825 DATA PI /3.1415927/
826
827 C CALCULATE THE PART OF THE SUB-ELEMENT MODAL DENSITY THAT IS
828 C NOT FREQUENCY DEPENDENT
829 PSUBMD = R * L / 2 * SQRT(12 * R/C *
830 * (1 - GAMMA ** 2) / (E * T ** 2))
831 C IF STIFFNESS REDUCTION IS REQUIRED, MULTIPLY BY SQRT(.5)
832 IF (STIFF) PSUBMD = SQRT(.5) * PSUBMD
833 C DO FOR EACH ANALYSIS FREQUENCY
834 DO 20 I = 1,NUMAF
835 SUBMD = PSUBMD
836 OMEGA = 2 * PI * FREQ(I)
837 C CALCULATE THE CRITERION FOR THE FREQUENCY DEPENDENT PART
838 C OF THE SUB-ELEMENT MODAL DENSITY
839 CRIT = OMEGA * R * SORT(RHO / E)
840 C IF THE CRITERION IS GREATER THAN 1, ALTER THE MODAL DENSITY
841 C APPROPRIATELY
842 IF (CRIT .LE. 1.) SUBMD = SUBMD * CRIT ** (2./3.)
843 C SUM THE SUB-ELEMENT MODAL DENSITY TO THE ELEMENT MODAL DENSITY
844 N(ELNUM,I) = N(ELNUM,I) + SUBMD
845 20 CONTINUE
846 C SUM THE SUB-ELEMENT MASS TO THE ELEMENT MASS
847 A = 2 * PI * R * L
848 MSUB = A * T * RHO * M
849 MASS(ELNUM) = MASS(ELNUM) + MSUB
850 C IF THIS IS NOT THE MAIN SUB-ELEMENT, RETURN TO EPROP
851 IF (.NOT. MAIN) RETURN
852 C ELSE PUT THE VALUES OF THE FOLLOWING SUB-ELEMENT VARIABLES INTO
853 C THE CORRESPONDING ELEMENT ARRAYS
854 EE(ELNUM) = E
855 THICK(ELNUM) = T
856 GENSE(ELNUM) = RHO
857 AREA(ELNUM) = A
858 VOL(ELNUM) = V
859 GAMMA(ELNUM) = GAMMA
860 ECC(ELNUM) = C
861 C SET MAIN TO INDICATE THAT ANY FOLLOWING SUB-ELEMENTS ARE NOT
862 C THE MAIN SUB-ELEMENT
863 MAIN = .FALSE.
864 RETURN
865 END

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CYLIN 57  
 CYLIN 58  
 CYLIN 59  
 OCT16 24  
 CYLIN 61  
 CYLIN 62  
 MAY13 48  
 CYLIN 64  
 CYLIN 65  
 CYLIN 66  
 CYLIN 67  
 CYLIN 68  
 CYLIN 69  
 CYLIN 70  
 CYLIN 71  
 CYLIN 72  
 AUG13K 2  
 CYLIN 74  
 CYLIN 75  
 CYLIN 76  
 CYLIN 77  
 OCT16 25  
 OCT16 26  
 OCT16 27  
 CYLIN 79  
 CYLIN 80  
 CYLIN 91  
 CYLIN 82  
 CYLIN 83  
 CYLIN 84  
 CYLIN 85  
 CYLIN 86  
 CYLIN 87  
 CYLIN 88  
 CYLIN 89  
 CYLIN 90  
 CYLIN 91  
 CYLIN 92  
 CYLIN 93  
 CYLIN 94

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865      SUBROUTINE JINPUT
866      C THIS SUBROUTINE READS JOINT PROPERTIES FROM UNIT 3 AND CHECKS
867      C THAT ELEMENT PAIRS ARE NOT INPUT MORE THAN ONCE. JINPUT IS
868      C CALLED FROM THE MAIN PROGRAM.
869
870      C THE FOLLOWING VARIABLES ARE USED IN THIS PROGRAM UNIT:
871      C NAME TYPE DESCRIPTION
872      C ----
873      C ASD(150) REAL ACOUSTIC SPACE DENSITY
874      C RL(150) REAL BEAM LENGTH
875      C INPUT INT NUMBER OF INPUT RECORDS READ
876      C JL(190) REAL JOINT LENGTH
877      C JTYPE(190)CHAR TYPE OF JOINT
878      C NE(150) INT FIRST ELEMENT
879      C NE2(190) INT SECOND ELEMENT
880      C NS(190) INT NUMBER OF SIDES
881      C NUNEL INT NUMBER OF ELEMENTS
882      C SP(190) REAL POLY SPACING
883      C T1(190) REAL THICKNESS OF FIRST ELEMENT OF PAIR
884      C T2(190) REAL THICKNESS OF SECOND ELEMENT OF PAIR
885      C TOTAL INT NUMBER OF ELEMENT PAIRS INPUT
886
887      C THE FOLLOWING COMMON BLOCKS ARE USED:
888      C BLOCK OTHER PROGRAM UNITS USING THIS COMMON BLOCK
889      C ----
890      C C82 MAIN,EPROP,BLOCK DATA
891      C C87 EPROP,JPROP,EXCITE,ANSWER,SOLVE,RITER,BLOCK DATA
892      C C810 EPROP,BLOCK DATA
893      C C812 JPROP
894
895      CHARACTER*2 JTYPE
896      INTEGER TOTAL
897      REAL JL
898      LOGICAL ERROR
899      COMMON /C82/ ERROR
900      COMMON /C87/ NUNEL,R
901      COMMON /C810/ INPUT
902      COMMON /C812/ JTYPE(190)
903      * T1(190),T2(190),ASD(190),NS(190),SP(190),JPUT(190),TOTAL
904      110 FORMAT (2I2,A2,I2,A6F10.2)
905      110 FORMAT (10... WARNING ... ON INPUT RECORD 1,14,1, ONE OR 1,
906      1 1 BOTH MEMBERS OF THE 1/1 ELEMENT PAIR 1,12,1 AND 1,12,
907      2 1 WAS EITHER LESS THAN 1 OR GREATER THAN 1,12,1, THE 1/
908      3 1 TOTAL NUMBER OF ELEMENTS. THIS PAIR WILL BE IGNORED. 1/
909      120 FORMAT (10... WARNING ... THE ELEMENT PAIR 1,12,1 AND 1,
910      1 12,1 ON INPUT RECORD 1,14,1 WAS 1/1 PREVIOUSLY READ ON 1,
911      2 1 INPUT RECORD 1,14,1. THE FIRST VALUES WILL BE USED. 1/
912      3 1 THIS ERROR WAS DISCOVERED BY SUBROUTINE JINPUT.1)
913      130 FORMAT (10... ERROR ... WHILE ATTEMPTING TO READ JOINT 1,
914      1 1,1,1,1/
915      2 1,1,1,1/
916      3 1 THIS ERROR WAS DISCOVERED BY SUBROUTINE JINPUT.1)
917
918      MAY13 49
919      JINPUT 3
920      JINPUT 4
921      JINPUT 5
922      JINPUT 6
923      JINPUT 7
924      JINPUT 8
925      JINPUT 9
926      JINPUT 10
927      AUG12J 4
928      JINPUT 12
929      JINPUT 13
930      JINPUT 14
931      JINPUT 15
932      JINPUT 16
933      JINPUT 17
934      JINPUT 18
935      JINPUT 19
936      JINPUT 20
937      JINPUT 21
938      JINPUT 22
939      JINPUT 23
940      JINPUT 24
941      JINPUT 25
942      JINPUT 26
943      JINPUT 27
944      JINPUT 28
945      JINPUT 29
946      JINPUT 30
947      JINPUT 31
948      JINPUT 32
949      JINPUT 33
950      JINPUT 34
951      JINPUT 35
952      JINPUT 36
953      JINPUT 37
954      JINPUT 38
955      JINPUT 39
956      JINPUT 40
957      MAY13 50
958      AUG12J 5
959      JINPUT 42
960      JINPUT 43
961      JINPUT 44
962      MAY13 52
963      MAY13 53
964      JINPUT 47
965      JINPUT 48
966      JINPUT 49
967      MAY13 54
968      JINPUT 51
969      JINPUT 52
970      JINPUT 53
971      MAY13 55
972      JINPUT 55
973      JINPUT 56

```



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921 140 FORMAT (10... ERROR ... THE END OF FILE WAS REACHED REFORM 1,
922 1 1 ANY INFORMATION/1 ABOUT JOINT PROPERTIES WAS READ.1/
923 3 1 THIS ERROR WAS DISCOVERED BY SUBROUTINE JINPUT.1)
924 150 FORMAT (10 BECAUSE OF THE ERRORS LISTED ABOVE, THE SEA PROGRAM,
925 1 1 WILL ABORT.1)
926 150 FORMAT (10... WARNING ... ON INPUT RECORD 1,1,1,
927 1 1 BOTH ELEMENT NUMBERS WERE/1 GIVEN AS 1,12,
928 2 1. THEY MUST BE DIFFERENT. THIS RECORD WILL BE IGNORED.1/
929 3 1 THIS ERROR WAS DISCOVERED BY SUBROUTINE JINPUT.1)
930 170 FORMAT (16, 4X, 1 FIRST ELEMENT = 1, 12 / 10X, 1 SECOND 1,
931 1 ELEMENT = 1, 12 / 10X, 1 TYPE OF JOINT = 1, A2 / 10X,
932 2 1 NUMBER OF SIDES = 1, 12 / 10X, 1 JOINT LENGTH = 1,
933 3 1 PE12.5E2 / 10X, 1 THICKNESS OF FIRST ELEMENT = 1,
934 4 1 PE12.5E2 / 10X, 1 THICKNESS OF SECOND ELEMENT = 1,
935 5 1 PE12.5E2 / 10X, 1 ACOUSTIC SPACE DENSITY = 1, 1 PE12.5E2 /
936 6 10X, 1 BEAM LENGTH = 1, 1 PE12.5E2 / 10X, 1 INSERTION LOSS 1,
937 7 1 FACTOR = 1, 1 PE12.5E2)
938 180 FORMAT (10... 1 RECORD / 1 NUMBER, 1BX, 1 DATA READ 1,
939 2 1 FROM 1, 1 IT 31 / 1X, 6(1H-), 1BX, 2(1H-)) / )
940 C WRITE A HEADING TO CONTINUE PRINTING INPUT DATA
941 WRITE (6,180)
942 C INITIALIZE TOTAL
943 TOTAL = 0
944 C INITIALIZE I
945 I = 0
946 C INCREMENT I
947 5 I = I + 1
948 C INCREMENT INPUT
949 10 INPUT = INPUT + 1
950 C READ THE JOINT PROPERTIES
951 READ (3,108,ERR=60,END=70) NE1(I), NE2(I), JTYPE(I), NS(I), JL(I),
952 1 1(I), 12(I), ASD(1), BL(I), SP(I)
953 C WRITE THE JOINT PROPERTIES TO OUTPUT
954 WRITE (6,170) INPUT, 1(I), NE2(I), JTYPE(I), NS(I), JL(I),
955 1 1(I), 12(I), ASD(1), BL(I), SP(I)
956 C PUT THE INPUT RECORD NUMBER IN JPUT
957 JPUT(I) = INPUT
958 C IF EITHER ELEMENT NUMBER IS OUT OF RANGE,
959 IF (NE1(I) .GE. 1 .AND. NE1(I) .LE. NUMEL .AND. NE2(I) .GE. 1
960 .AND. NE2(I) .LE. NUMEL) GO TO 15
961 C THEN WRITE A WARNING MESSAGE
962 WRITE (6,110) INPUT, NE1(I), NE2(I), NUMEL
963 GO TO 10
964 C END IF
965 C IF BOTH ELEMENTS IN THE PAIR ARE THE SAME,
966 15 IF (NE1(I) .NE. NE2(I)) GO TO 20
967 C THEN WRITE A WARNING MESSAGE
968 WRITE (6,160) INPUT, NE1(I)
969 GO TO 10
970 C END IF
971 C IF THIS IS THE FIRST ELEMENT PAIR, SKIP THE FOLLOWING TEST
972 20 IF (1 .LE. 1) GO TO 50
973 C DO FOR EACH PAIR OF ELEMENTS PREVIOUSLY INPUT
974 10 JC = 1, TOTAL
975 C IF THE CURRENT PAIR OF ELEMENTS MATCHES A PREVIOUS PAIR,
976 IF ((NE1(I) .NE. NE1(J) .OR. NE2(I) .NE. NE2(J)) .AND.

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```

JINPUT57
JINPUT58
JINPUT59
JINPUT60
JINPUT61
JINPUT62
JINPUT63
JINPUT64
JINPUT65
AUG26 51
AUG26 52
AUG26 53
AUG26 54
AUG26 55
AUG26 56
NOVA 5
NOVA 6
AUG26 59
AUG26 60
AUG26 61
AUG26 62
JINPUT66
JINPUT67
JINPUT68
JINPUT69
JINPUT70
JINPUT71
JINPUT72
JINPUT73
JINPUT74
JINPUT75
AUG12J 6
AUG26 63
AUG26 64
AUG26 65
JINPUT77
JINPUT78
JINPUT79
JINPUT80
JINPUT81
JINPUT82
JINPUT83
JINPUT84
JINPUT85
JINPUT86
MAY13 56
JINPUT88
JINPUT89
JINPUT90
JINPUT91
JINPUT92
JINPUT93
JINPUT94
JINPUT95
JINPUT96
JINPUT97

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977      *      (NE1(I),-NF, NE2(J),-OP, NF2(I),-NF, NE1(J))) GO TO 10
978      C IMFA WRITE A WARNING MESSAGE
979      WRITE (6,120) NE1(I),NE2(J),INPUT,JPUT(J)
980      GO TO 10
981      C END IF
982      GO CONTINUE
983      C INCREMENT TOTAL
984      50 TOTAL = I
985      GO TO 5
986      C IF AN ERROR OCCURRED ON READING, WRITE AN ERROR MESSAGE
987      60 WRITE (6,130) INPUT
988      ERROR = .TRUE.
989      GO TO 10
990      C IF THE END OF FILE WAS ENCOUNTERED, PUT NO RECORDS
991      C WERE READ, WRITE A MESSAGE AND SET THE ERROR FLAG
992      70 IF (TOTAL.GE. 1) GO TO 80
993      ERROR = .TRUE.
994      WRITE (6,140)
995      C IF THERE WERE NO ERRORS, RETURN
996      80 IF (.NOT. ERROR) RETURN
997      C ELSE WRITE A MESSAGE AND TERMINATE THE PROGRAM
998      WRITE (6,150)
999      CALL FERR
1000     END

```

```

JINPUT76
JUNE2511
JINPUT00
JINPUT01
JINPUT02
JINPUT03
JINPUT04
JINPUT05
JINPUT06
JINPUT07
JINPUT08
JINPUT09
JINPUT10
AUG26 86
JINPUT12
JINPUT13
JINPUT14
JINPUT15
JINPUT16
JINPUT17
JINPUT18
JINPUT19
JINPUT20
JINPUT21

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1001      SUBROUTINE JPROP
1002      C THIS SUBROUTINE CALCULATES THE ELEMENT-TO-ELEMENT STRUCTURAL
1003      C COUPLING COEFFICIENT (PMI). JPROP IS CALLED FROM THE MAIN
1004      C PROGRAM.
1005      C
1006      C THE FOLLOWING VARIABLES ARE USED IN THIS PROGRAM UNIT:
1007      C NAME TYPE DESCRIPTION
1008      C ----
1009      C A REAL PSEUDO AREA
1010      C AF INT ANALYSIS FREQUENCY ORIGINAL
1011      C AREA(20) REAL SURFACE AREA
1012      C ASD(190) REAL ACOUSTIC SPACE DENSITY
1013      C BL(190) REAL BEAM LENGTH
1014      C CRFREQ REAL CRITICAL FREQUENCY
1015      C DENSE(20) REAL DENSITY
1016      C EC(20) REAL SPEED OF SOUND IN ROOM MEDIUM
1017      C EE(20) REAL MODULUS OF ELASTICITY
1018      C EGAPPA(20) REAL POISSON'S RATIO
1019      C E1 INT FIRST ELEMENT
1020      C E2 INT SECOND ELEMENT
1021      C FREQ REAL FREQUENCY
1022      C JL(190) REAL JOINT LENGTH
1023      C JTYPE(190) CHAR TYPE OF JOINT
1024      C PCODES REAL NUMBER OF MODES IN BANDWIDTH
1025      C M(20,40) REAL MODAL DENSITY
1026      C NE1(190) INT FIRST ELEMENT
1027      C NE2(190) INT SECOND ELEMENT
1028      C NS(150) INT NUMBER OF STORS
1029      C OMEGA REAL 2*PI*FREQUENCY
1030      C PM1(20,20) REAL COUPLING COEFFICIENT
1031      C SIGMA(20) REAL RADIATION EFFICIENCY
1032      C SP(150) REAL ROLT SPACING REDUCTION
1033      C TAU REAL THICKNESS RATIO
1034      C THICK(20) REAL THICKNESS
1035      C TOTAL INT NUMBER OF ELEMENT PAIRS (I.E., JOINTS)
1036      C V(20) REAL VOLUME
1037      C
1038      C THE FOLLOWING COMMON BLOCKS ARE USED:
1039      C BLOCK OTHER PROGRAM UNITS USING THIS COMMON BLOCK
1040      C ----
1041      C CB3 MAIN,EXCITE,ANSWER
1042      C CPE FPROP,BEAM,MEMBR,PLATE,ROOM,CYLIN,EXCITE,ANSWER
1043      C CE7 EPRCP,FINPUT,EXCITE,ANSWER,SOLVE,RIVER,BLOCK DATA
1044      C CB9 EPROP,MEAM,MEMBR,PLATE,ROOM,CYLIN,EXCITE,BLOCK DATA
1045      C CH12 JINPUT
1046      C CR14 EXCITE/ANSWER
1047      C
1048      CHARACTER*2 JTYPE
1049      INTEGER AF,E1,E2,TOTAL
1050      REAL JL,APCODES
1051      LOGICAL ERROR
1052      COMMON /CB3/ FREQ,AF,OMEGA
1053      COMMON /CPE/ THICK(20),AREA(20),DENSE(20),VOL(20),FF(20),
1054      * EGAPPA(20),FC(20)
1055

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JPROP 2  
JPROP 3  
JPROP 4  
JPROP 5  
JPROP 6  
JPROP 7  
MAY13 57  
JPROP 9  
JPROP 10  
JPROP 11  
JPROP 12  
JPROP 13  
AUG12J 7  
JPROP 15  
JPROP 16  
JPROP 17  
JPROP 18  
JPROP 19  
JPROP 20  
JPROP 21  
JPROP 22  
JPROP 23  
JPROP 24  
JPROP 25  
JPROP 26  
JPROP 27  
JPROP 28  
JPROP 29  
JPROP 30  
JPROP 31  
JPROP 32  
JPROP 33  
JPROP 34  
JPROP 35  
JPROP 36  
JPROP 37  
JPROP 38  
JPROP 39  
JPROP 40  
JPROP 41  
JPROP 42  
JPROP 43  
JPROP 44  
JPROP 45  
JPROP 46  
JPROP 47  
JPROP 48  
JPROP 49  
JPROP 50  
JPROP 51  
JPROP 52  
JPROP 53  
JPROP 54  
JPROP 55  
JPROP 56  
JPROP 57

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1057 COMMON /CR7/ NUMFL,G JPROP 58
1058 COMMON /CB5/ N(20,40) JPROP 59
1059 COMMON /CR12C/ JTYPE(190) MAY13 58
1060 COMMON /CR12/ NE1(190),NE2(190),JL(190),BL(190), AUG12J 8
1061 * T1(190),T2(190),A-D(190),NS(190),SP(190),JPUT(190),TOTAL JPROP 61
1062 COMMON /CB14/ PHI(20,20),MODES(20) JPROP 62
1063 DATA PI,ERROR /3.141592, .FALSE./ MAY13 60
1064 200 FORMAT (10,*,ERROR ***, THE TYPE OF JOINT GIVEN FOR I, JPROP 64
1065 1 ELEMENT PAIR I,J,T AND I,J,T ON INPUT RECORD I,J,I,T IST, JPROP 65
1066 2 A2,T. THIS IS NOT A VALID TYPE. THE TYPE MUST I/ JPROP 65
1067 3 I RE PP, BP, BJ, OR PA,I/ JPROP 67
1068 4 I THIS ERROR WAS DISCOVERED BY SUBROUTINE JPROP,T JPROP 68
1069 220 FORMAT (10BECAUSE OF THE ERRORS LISTED ABOVE, THE SEA PROGRAM, JPROP 69
1070 * I WILL ABORT.T) JPROP 70
1071 C INITIALIZE THE COUPLING COEFFICIENTS TO C AND CALCULATE MODES JPROP 74
1072 DO 20 J = 1, 20 JPROP 75
1073 MODES(J) = N(J,AF) * OMEGA / 4.33 JPROP 76
1074 DO 10 I = 1, 20 JPROP 77
1075 PHI(I,J) = 0. JPROP 78
1076 10 CONTINUE JPROP 79
1077 20 CONTINUE JPHOP 80
1078 C DO FOR EACH PAIR OF ELEMENTS FOR WHICH JOINT PROPERTIES JPROP 81
1079 C WERE INPUT JPROP 82
1080 DO 100 I = 1, TOTAL JPROP 83
1081 C PUT THE ELEMENT NUMBERS IN SIMPLE VARIABLES TO AVOID JPROP 84
1082 C DOUBLE SUBSCRIPTS E1 = NE1(I) JPROP 85
1083 E2 = NE2(I) JPROP 86
1084 IF THE TYPE OF JOINT IS PLATE TO PLATE, MAY13 61
1085 IF (JTYPE(I) .NE. TPPI) GO TO 30 JPROP 88
1086 C THEN SET TAU TO A/27 JPROP 89
1087 TAU = A. / 27. JPROP 90
1088 C UNLESS THE RATIO OF THICKNESS IS LESS THAN .5, IN WHICH CASE JPROP 91
1089 C SET TAU TO THE RATIO JPROP 92
1090 IF (T1(I) / T2(I) .LT. .5) JPROP 93
1091 TAU = T1(I) / T2(I) JPROP 94
1092 C CALCULATE THE PSEUDO AREA JPROP 95
1093 A = 2. * PI * N(E1,AF) * 2. * T1(I) * SORT(E(E1) / JPROP 96
1094 * 012. * DENSE(E1) * (1. - EGAMMA(E1) ** 2))) JUNE2512
1095 C CALCULATE THE COUPLING COEFFICIENT FOR PLATE TO PLATE COUPLING JPROP 98
1096 PHI(E1,E2) = 1.07 * JL(I) / (PI * A * MODES(E2)) * JPROP 99
1097 E SORT(OMEGA * THICK(E1) * SORT(E(E1) / (DENSE(E1) * JPROP 100
1098 * 61. - EGAMMA(E1) ** 2))) * TAU JPROP 101
1099 GO TO 50 JPROP 102
1100 C ELSE IF TYPE OF JOINT IS BEAM TO PLATE, JPROP 103
1101 30 IF (JTYPE(I) .NE. TPPI) GO TO 40 JPROP 104
1102 C CALCULATE THE COUPLING COEFFICIENT FOR THAT TYPE JPROP 105
1103 PHI(E1,E2) = 2. * PI * FREQ * JL(I) / (MODES(E2) * JPROP 106
1104 * A * BL(I)) AUG12J 9
1105 GO TO 70 JPROP 107
1106 C ELSE IF THE TYPE OF JOINT IS BOLTED OR RIVETED JOINT, JPROP 109
1107 40 IF (JTYPE(I) .NE. TPPI) GO TO 50 JPROP 110
1108 C THEN SET TAU TO A/27 JPROP 111
1109 TAU = A. / 27. JPROP 112
1110 C UNLESS THE RATIO OF THICKNESS IS LESS THAN .5, IN WHICH CASE JPROP 114
1111 C SET TAU TO THE RATIO JPROP 115
1112

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```

1115 IF (T1(I) / T2(I) .LT. .5)
1116   TAU = T1(I) / T2(I)
1117   C CALCULATE THE PSEUDO AREA
1118   A = 2. * PI * A(E1,AF) * 2. * T1(I) * SORT(E(E1) /
1119     &12. * DENSE(E1) * (1. - EGAMMA(E1) ** 2)))
1120   C CALCULATE THE COUPLING COEFFICIENT FOR THAT TYPE
1121   PHIE1,E2 = 1.07 * JLI(I) / (PI * A * MODES(E2)) *
1122     SORT(OMEGA * THICK(E1) * SORT(E(E1) / (DENSE(E1) *
1123     &1. - EGAMMA(E1) ** 2))) * TAU / SP(I)
1124   GO TO 90
1125 C IF THE TYPE OF JOINT IS PLATE TO ACOUSTIC SPACE,
1126   50 IF (JTYPE(I) .NE. 1) GO TO 60
1127   C CALCULATE THE CRITICAL FREQUENCY
1128   CRFREQ = EC(E2) ** 2 / (1.9 * THICK(E1)) *
1129     SORT(DENSE(E1) / EF(E1))
1130   C CALCULATE THE RADIATION EFFICIENCY USING THE SIGF FUNCTION
1131   SIGMA = 10. ** SIGF(CRFREQ / CRFREQ)
1132   C CALCULATE THE COUPLING COEFFICIENT FOR THAT TYPE
1133   PHIE1,E2 = 4.33 * PI * EC(E2) ** 4 / (CRFREQ * OMEGA ** 2
1134     * VOL(E2) * NS(I) * DENSE(E1) * THICK(E1)
1135     * SIGMA / ASD(I))
1136   GO TO 90
1137 C ELSE IF THE TYPE OF JOINT IS NONE OF THE ABOVE, SET THE ERROR
1138 C FLAG AND WRITE AN ERROR MESSAGE
1139   60 ERROR = .TRUE.
1140   WRITE (0,200) E1,E2,JPUT(I),JTYPE(I)
1141 C END IF
1142   90 PHIE2,E1 = PHIE1,E2
1143   100 CONTINUE
1144 C IF NO ERRORS IN JOINT TYPE WERE ENCOUNTERED, RETURN
1145   IF (.NOT. ERROR) RETURN
1146 C ELSE WRITE A MESSAGE AND TERMINATE THE PROGRAM
1147   WRITE (6,220)
1148   CALL FERR
1149   END

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JPROPI16  
 JPROPI17  
 JPROPI18  
 JUNE2513  
 JPROPI20  
 JPROPI21  
 JPROPI22  
 JPROPI23  
 MAY14 3  
 JPROPI25  
 JPROPI26  
 JPROPI27  
 JPROPI28  
 JPROPI29  
 JPROPI30  
 JPROPI31  
 JPROPI32  
 JPROPI33  
 MAY13 62  
 AUG12J11  
 AUG12J12  
 JPROPI36  
 JPROPI37  
 JPROPI38  
 JPROPI39  
 JPROPI40  
 JPROPI41  
 JPROPI42  
 JPROPI43  
 JPROPI45  
 JPROPI46  
 JPROPI47  
 JPROPI48  
 JPROPI49  
 JPROPI50

```

1148      FUNCTION SIGF(X)
1149      C THIS FUNCTION SUBPROGRAM RETURNS A VALUE FOR THE RADIATION
1150      C EFFICIENCY OF A PANEL BASED ON THE FOLLOWING TABLE:
1151      C
1152      C   X   SIGF // X   SIGF // X   SIGF // X   SIGF
1153      C   ---   --- // ---   --- // ---   --- // ---
1154      C   .00  -1.0 // .41  -1.0 // .77  -0.2 // 1.0  0.0
1155      C   .11  -1.6 // .50  -0.8 // .92  0.0 // 1.5  0.4
1156      C   .22  -1.4 // .60  -0.6 // .88  0.2 // 1.9  0.2
1157      C   .32  -1.2 // .68  -0.4 // .92  0.4 // 2.0  0.0
1158      C
1159      C THE ENTRY IN THE X COLUMN IS THE MAXIMUM VALUE OF X FOR WHICH
1160      C SIGF HAS THE INDICATED VALUE. SIGF IS CALLED FROM JPROP AND
1161      C ENCTE.
1162      C
1163      C THE FOLLOWING VARIABLES ARE USED IN THIS PROGRAM UNIT:
1164      C   NAME   TYPE   DESCRIPTION
1165      C   ----   ---   -----
1166      C   D(16)   REAL   A TABLE OF VALUES TO COMPARE WITH X
1167      C   I       INT    THE ORDINAL OF THE LEAST VALUE OF D GREATER THAN X
1168      C   K       REAL   THE RATIO OF THE ANALYSIS FREQUENCY TO THE
1169      C           CRITICAL FREQUENCY
1170      C
1171      C
1172      DIMENSION D(16)
1173      DATA D /0.0,.11,.22,.32,.41,.50,.60,.68,.77,
1174      I .82,.92,1.0,1.5,1.9,2.0/
1175      C DETERMINE WHERE X LIES IN THE TABLE
1176      DO 20 I = 2, 15
1177      IF (X .LE. D(I)) GO TO 30
1178      20 CONTINUE
1179      I = 16
1180      C SIGF REACHES A MAXIMUM AT I = 13. A DIFFERENT ALGORITHM
1181      C FOR COMPUTING SIGF IS REQUIRED DEPENDING ON WHETHER I IS
1182      C GREATER OR LESS THAN 13.
1183      30 IF (I .LT. 14) GO TO 50
1184      SIGF = -.2 + .2 * FLOAT(I)
1185      C INTERPOLATE
1186      SIGF = -.2 / (D(I) - D(I - 1)) * (D(I) - X) + SIGF
1187      RETURN
1188      50 SIGF = .6 - .2 * FLOAT(I - 13)
1189      C INTERPOLATE IF X IS LESS THAN 4
1190      IF (X .GE. 4.) RETURN
1191      SIGF = .2 / (D(I) - D(I - 1)) * (D(I) - X) + SIGF
1192      RETURN
1193      END

```

```

MAY13 63
SIGF 3
SIGF 4
SIGF 5
SIGF 6
SIGF 7
SIGF 8
SIGF 9
SIGF 10
SIGF 11
SIGF 12
MAY13 64
SIGF 14
SIGF 15
SIGF 16
SIGF 17
SIGF 18
SIGF 19
SIGF 20
SIGF 21
SIGF 22
SIGF 23
SIGF 24
SIGF 25
SIGF 26
SIGF 27
JUNE2514
SIGF 29
SIGF 30
SIGF 31
SIGF 32
SIGF 33
SIGF 34
SIGF 35
SIGF 36
SIGF 37
SIGF 38
SIGF 39
SIGF 40
SIGF 41
SIGF 42
SIGF 43
SIGF 44
SIGF 45
SIGF 46
SIGF 47

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1194      SUPROUTINE EXCITE
1195      C THIS SUBROUTINE DETERMINES ELEMENT ENERGY LEVELS OF ACOUSTIC
1196      C ENERGY INPUTS, DEPENDING ON THE TYPE OF EXCITATION. EXCITE IS
1197      C CALLED FROM THE MAIN PROGRAM.
1198
1199      C THE FOLLOWING VARIABLES ARE USED IN THIS PROGRAM UNIT:
1200      C NAME TYPE DESCRIPTION
1201      C ----
1202      C AF INT ANALYST'S FREQUENCY ORDINAL
1203      C AMM(20) REAL (AREA * MODES / MASS) OF SURFACE EXCITED BY
1204      C ACOUSTIC FIELD
1205      C AREA(20) REAL AREA
1206      C EC(20) REAL ELEMENT ENERGY LEVELS
1207      C EC(20) REAL SPEED OF SOUND IN ROOM MEDIUM
1208      C ETYPE(20) CHAR TYPE OF EXCITATION
1209      C MCOFS REAL NUMBER OF MODES IN BANDWIDTH
1210      C FREQ REAL FREQUENCY
1211      C G REAL GRAVITATIONAL CONSTANT
1212      C PASS(20) REAL MASS
1213      C MECH(20,40) REAL MECHANICAL INPUT
1214      C PTYPE(20) REAL TYPE OF MECHANICAL INPUT
1215      C N(20,40) REAL MODAL DENSITY
1216      C NUMEL INT NUMBER OF ELEMENTS
1217      C NPI(20) REAL 2*PI*FREQUENCY
1218      C S(20) REAL ACOUSTIC ENERGY INPUT
1219      C SIGMA(20) REAL RADIATION EFFICIENCY
1220      C SPL(20,40) REAL SOUND PRESSURE LEVELS
1221
1222      C THE FOLLOWING COMMON BLOCKS ARE USED:
1223      C BLOCK OTHER PROGRAM UNITS USING THIS COMMON BLOCK
1224
1225      C CB2 MAIN,JPROP,ANSWER
1226      C CB5 EPROP,REAM,MEMBR,PLATE,ROOM,CYLIN,JPROP,ANSWER
1227      C CH6 EPROP,PLATE
1228      C CB7 EPROP,INPUT,JPROP,ANSWER,SOLVE,RITER,BLOCK DATA
1229      C CB9 EPROP,REAM,MEMBR,PLATE,ROOM,CYLIN,JPROP,BLOCK DATA
1230      C CB11 REAM,MEMBR,PLATE,ROOM,CYLIN,ANSWER,BLOCK DATA
1231      C CB4 JPROP,ANSWER
1232      C CB15 ANSWER,SOLVE
1233
1234      C CHARACTER ETYPE,NTYPE*3
1235      C INTEGER AF
1236      C REAL M, MASS, MODES, MECH
1237      C COMMON /CB2/ FREQ,AF,OMEGA
1238      C COMMON /CB5/ THICK(20),AREA(20),DENSE(20),VOL(20),EE(20),
1239      C EGAMMA(20),EC(20)
1240      C COMMON /CB6/ SPL(20,40),MECH(20,40),MM(20),ETYPE(20),MYPE(20)
1241      C COMMON /CB7/ NUMEL,G
1242      C COMMON /CB9/ N(20,40)
1243      C COMMON /CB11/ MASS(20)
1244      C COMMON /CB14/ PHI(20,20),MCOFS(20)
1245      C COMMON /CB15/ S(20),F(20)
1246      C DATA PI /3.141592/
1247      C DO FOR EACH ELEMENT
1248      C DO I = 1, NUMEL
1249
1250      EXCITE 2
1251      MAY13 65
1252      EXCITE 4
1253      EXCITE 5
1254      EXCITE 6
1255      EXCITE 7
1256      EXCITE 8
1257      EXCITE 9
1258      EXCITE11
1259      OCT28 17
1260      OCT28 18
1261      EXCITE12
1262      EXCITE13
1263      EXCITE14
1264      EXCITE15
1265      EXCITE16
1266      EXCITE17
1267      EXCITE18
1268      EXCITE19
1269      EXCITE20
1270      JULY12F1
1271      EXCITE22
1272      EXCITE23
1273      JUNE2515
1274      EXCITE25
1275      EXCITE26
1276      EXCITE27
1277      EXCITE28
1278      EXCITE29
1279      EXCITE30
1280      EXCITE31
1281      EXCITE32
1282      EXCITE33
1283      OCT28 19
1284      EXCITE35
1285      EXCITE36
1286      EXCITE37
1287      EXCITE38
1288      EXCITE39
1289      EXCITE40
1290      EXCITE41
1291      EXCITE42
1292      EXCITE43
1293      JULY12F2
1294      EXCITE45
1295      EXCITE46
1296      EXCITE47
1297      UNVAC2 1
1298      EXCITE49
1299      EXCITE50
1300      EXCITE51
1301      EXCITE52
1302      EXCITE53
1303      EXCITE54
1304      EXCITE55
1305      EXCITE57

```

```

1250 C INITIALIZE THE ACOUSTIC ENERGY INPUT AND ELEMENT ENERGY LEVELS
1251 S(I) = 0.
1252 E(I) = 0.
1253 C IF(1) THE TYPE OF EXCITATION IS ACOUSTIC.
1254 IF (ETYPF(I) .NE. 10) GO TO 20
1255 C CALCULATE THE CRITICAL FREQUENCY
1256 CRFREQ = EC(I) ** 2 / (1.9 * THICK(I)) *
1257 SGRY(DENSE(I) / EC(I))
1258 C CALCULATE THE RADIATION EFFICIENCY USING THE SIGF FUNCTION
1259 SIGMA = 10. ** SIGF(CRFREQ / CRFREQ)
1260 C THEN CALCULATE THE ACOUSTIC ENERGY INPUT
1261 S(I) = 4.33 * PI * EC(I) ** 2 * 8.41 * 10. **
1262 (SPL(I, AF) / 10. - 1A.) * SIGMA * ANM(I) /
1263 (OMEGA ** 2)
1264 GO TO 100
1265 C ELSE IF(1) THE TYPE OF EXCITATION IS MECHANICAL,
1266 IF (ETYPF(I) .NE. 10) GO TO 100
1267 C THEN IF(2) THE TYPE OF MECHANICAL INPUT IS RMS,
1268 IF (OTYPF(I) .NE. 10MS) GO TO 30
1269 C THEN CALCULATE THE ELEMENT ENERGY LEVEL
1270 E(I) = MASS(I) / OMEGA ** 2 * (MECH(I, AF) * G) ** 2
1271 GO TO 100
1272 C ELSE IF(2) THE TYPE OF MECHANICAL INPUT IS PSB,
1273 C THEN CALCULATE THE ELEMENT ENERGY LEVEL
1274 E(I) = MASS(I) / (OMEGA ** 2)
1275 E(I) = FTEMP * MECH(I, AF) * (G ** 2) * (FREQ/4.33)
1276 C END IF(2)
1277 C END IF(1)
1278 100 CONTINUE
1279 RETURN
1280 ENC

```

EXCITE5A  
 EXCITE50  
 EXCITE40  
 EXCITE41  
 EXCITE42  
 EXCITE43  
 EXCITE44  
 EXCITE45  
 EXCITE46  
 EXCITE47  
 EXCITE48  
 OCT28 21  
 OCT28 22  
 OCT28 23  
 EXCITE72  
 EXCITE73  
 EXCITE74  
 EXCITE75  
 JULY1011  
 EXCITE77  
 EXCITE78  
 EXCITE79  
 EXCITE80  
 EXCITE81  
 JULY12E1  
 JULY12E2  
 EXCITE84  
 EXCITE85  
 EXCITE86  
 EXCITE88  
 EXCITE89



```

1291 SUBROUTINE ANSWER
1292 C THIS SUBROUTINE SOLVES THE SEA SYSTEM OF EQUATIONS FOR
1293 C ELEMENT ENERGY LEVELS AND STORES THE SOLUTION IN ARRAY
1294 C ARAH TO BE PRINTED OUT BY RITER. ANSWER IS CALLED FROM
1295 C THE MAIN PROGRAM.
1296 C
1297 C
1298 C THE FOLLOWING VARIABLES ARE USED IN THIS PROGRAM UNIT:
1299 C NAME TYPE DESCRIPTION
1300 C ----
1301 C ARAH(20,40) REAL AVERAGE ACCELERATION
1302 C ALPHA(20,20) REAL MATRIX OF COEFFICIENTS
1303 C E(20) REAL ELEMENT ENERGY LEVELS
1304 C ETA(20) REAL DAMPING
1305 C FREQ REAL FREQUENCY
1306 C MASS(20) REAL MASS
1307 C MODES(20) REAL NUMBER OF MODES IN BANDWIDTH
1308 C NUPEL INT MEMBER OF ELEMENTS
1309 C OMEGA REAL 2*PI*FREQUENCY
1310 C CTYPE CHAR TYPE OF OUTPUT
1311 C PHI(20,20) REAL COUPLING COEFFICIENT
1312 C S(20) REAL ACOUSTIC ENERGY INPUT
1313 C SFREQ(20) REAL STARTING FREQUENCY
1314 C SIZE INT SIZE OF REDUCED ALPHA ARRAY
1315 C SLOPE(20) REAL SLOPE
1316 C
1317 C THE FOLLOWING COMMON BLOCKS ARE USED:
1318 C BLOCK OTHER PROGRAM UNITS USING THIS COMMON BLOCK
1319 C ----
1320 C CP2 MAIN,JPROP,EXCITE
1321 C CP5 EFPROP,AREAM,PEMBR,PLATE,ROOM,CYLIN,JPROP,EXCITE
1322 C CB7 EFPROP,JINPUT,JPROP,EXCITE,SOLVE,RITER,BLOCK DATA
1323 C CP8 EFPROP,RITER,BLOCK DATA
1324 C CP11 REAM,PEMBR,PLATE,ROOM,CYLIN,EXCITE,BLOCK DATA
1325 C CP14 JPROP,EXCITE
1326 C CP15 EXCITE,SOLVE
1327 C CP16 SOLVE
1328 C CP17 RITER
1329 C
1330 C CHARACTER*1 OTYPE
1331 C INTEGER AP,SIZE
1332 C REAL MASS,MODES
1333 C DIMENSION ARAH(20,20),SMAT(20)
1334 C COMMON /CB2/ FREQ,AF,OMEGA
1335 C COMMON /CB5/ THICK(20),AREA(20),DENSEF(20),VOL(20),EE(20),
1336 C FEMMA(20),EC(20)
1337 C COMMON /CB7/ NUPEL,G
1338 C COMMON /CB8/ SLOPE(20),SFREQ(20),ETA(20),OTYPE
1339 C COMMON /CP11/ MASS(20)
1340 C COMMON /CP14/ PHI(20,20),PODFS(20)
1341 C COMMON /CP15/ S(20),F(20)
1342 C COMMON /CP16/ ALPHA(20,20)
1343 C COMMON /CP17/ ARAH(20,40)
1344 C DO FOR EACH ELEMENT

```

ANSWER 2  
 ANSWER 3  
 ANSWER 4  
 ANSWER 5  
 ANSWER 6  
 ANSWER 7  
 ANSWER 8  
 ANSWER 9  
 ANSWER10  
 MAY13 68  
 ANSWER12  
 ANSWER13  
 ANSWER14  
 ANSWER15  
 ANSWER16  
 ANSWER17  
 ANSWER18  
 ANSWER19  
 ANSWER20  
 MAY14 4  
 ANSWER21  
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 ANSWER47  
 ANSWER48  
 ANSWER49  
 UNIVAC 3  
 ANSWER51  
 ANSWER52  
 ANSWER53  
 ANSWER54  
 ANSWER55  
 ANSWER56

```

1337 DO 100 I = 1, NUMEL
1338 C SET THE DAMPING FACTOR EQUAL TO THE INPUT VALUE
1339 ETAS = ETAS(I)
1340 SLOPE(I) = SLOPE(I) / 20
1341 C IF DAMPING IS FREQUENCY DEPENDENT AND THE ANALYSIS FREQUENCY
1342 C IS GREATER THAN THE STARTING FREQUENCY OF THE DEPENDENCY,
1343 C CALCULATE THE DAMPING FACTOR
1344 IF (SLOPE(I) .NE. 0. .AND. FREQ .GT. SFREQ(I))
1345 *
1346 ETAS = ETAS * (FREQ / SFREQ(I)) ** SLOPE(I)
1347 C IF THIS IS NOT THE LAST ELEMENT,
1348 IF .NOT. (EQ. NUMEL) GO TO 50
1349 C THEN FOR EACH SUCCEEDING ELEMENT,
1350 IPLUS1 = I + 1
1351 DO 40 J = IPLUS1, NUMEL
1352 C CALCULATE THE I-TH ROW OF ALPHA TO THE RIGHT OF THE MAIN
1353 C DIAGONAL
1354 ALPHA(I,J) = -MODES(I) * PHI(I,J)
1355 ALPHA(J,I) = -MODES(J) * PHI(J,I)
1356 40 CONTINUE
1357 C END IF
1358 C CALCULATE ALPHA ON THE MAIN DIAGONAL
1359 *
1360 ALPHA(I,I) = OMEGA * ETAS
1361 DO 80 J = 1, NUMEL
1362 ALPHA(I,J) = ALPHA(I,I) * MODES(J) * PHI(I,J)
1363 80 CONTINUE
1364 100 CONTINUE
1365 C INITIALIZE SIZE
1366 SIZE = NUMEL
1367 C DO FOR EACH ELEMENT
1368 DO 150 I = 1, NUMEL
1369 C IF THE ENERGY LEVEL IS ALREADY KNOWN,
1370 IF (E(I) .EQ. 0.) GO TO 150
1371 C THEN BC FOR EACH ELEMENT
1372 DO 130 J = 1, NUMEL
1373 C SUBTRACT FROM THE ACOUSTIC ENERGY ARRAY S THAT PORTION OF
1374 C THE VALUE WHICH CAME FROM THE PRODUCT OF THE ALPHA ARRAY
1375 C AND THE KNOWN ENERGY LEVEL
1376 S(IJ) = S(IJ) - ALPHA(I,J) * E(I)
1377 130 CONTINUE
1378 C DECREMENT SIZE TO SHOW THAT THE I-TH ROW AND COLUMN WILL BE
1379 C ELIMINATED FROM THE ALPHA ARRAY WHEN IT BECOMES A MAT AND THE
1380 C I-TH ELEMENT FROM THE S ARRAY WHEN IT BECOMES SPAT
1381 SIZE = SIZE - 1
1382 C END IF
1383 150 CONTINUE
1384 C CALL SOLVE TO CREATE THE REDUCED ARRAYS AND SOLVE THE MATRIX
1385 C EQUATION
1386 CALL SOLVE (AMAT, SMAT, SIZE)
1387 DO 200 I = 1, NUMEL
1388 C PUT THE SOLUTION IN ARAR
1389 ARAR(I,AF) = E(I) * OMEGA ** 2 / MASS(I)
1390 C IF THE DEFAULT OUTPUT TYPE PSD IS SELECTED, PUT ARAR IN THAT FORM
1391 IF (OTYPE .EQ. 1) GO TO 140
1392 ARAR(I,AF) = ARAR(I,AF) / (6 ** 2 * FREQ * 4.33
1393

```

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MAY13 72
ANSWER58
ANSWER59
AUG26 67
ANSWER60
ANSWER61
ANSWER62
MAY13 73
ANSWER64
ANSWER65
ANSWER66
ANSWER67
ANSWER68
ANSWER69
ANSWER70
ANSWER71
ANSWER72
ANSWER73
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ANSWER98
ANSWER99
ANSWER100
ANSWER101
ANSWER102
ANSWER103
ANSWER104
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ANSWER106
ANSWER107
ANSWER108
ANSWER109
ANSWER110
ANSWER111

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1394  
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1399

```

      GO TO 200
C ELSE IF THE OUTPUT TYPE IS RMS, PUT ARRAY IN THAT FORM
  LFC  ARRAY(:JAF) = SORT(ARRAY(I,AF)) / G
C FMC IF
  200 CONTINUE
      RETURN
      ENC

```

ANSWE112  
ANSWE113  
ANSWE114  
ANSWE115  
ANSWE116  
ANSWE117  
ANSWE11A

```

1400 SUBROUTINE SOLVE (AMAT,SMAT,SIZE)
1401 C THIS SUBROUTINE CREATES MATRIX A AT F.GM ALPHA BY ELIMINATING
1402 C THOSE ROWS AND COLUMNS REPRESENTING ELEMENTS FOR WHICH THE
1403 C ENERGY LEVELS ARE ALREADY KNOWN, IN THE SAME WAY, SPAT IS
1404 C CREATED FROM S. SUBROUTINE GASSEN FOR THE BINARY SYSTEMS: A1S
1405 C IS THEN CALLED TO SOLVE THE MATRIX EQUATION ON AMAT-X=SMAT FOR X.
1406 C GASSEN PUTS THE SOLUTION IN SMAT, SO THIS SUBROUTINE PUTS
1407 C THE RESULTS IN E, LEAVING INTACT THOSE ELEMENTS OF E
1408 C WHICH WERE ALREADY KNOWN. SOLVE IS CALLED BY ANSWER.
1409 C
1410 C
1411 C THE FOLLOWING VARIABLES ARE USED IN THIS PROGRAM UNIT:
1412 C NAME TYPE DESCRIPTION
1413 C ----
1414 C ALPHA(20,20) REAL MATRIX OF COEFFICIENTS
1415 C AMAT(SIZE,SIZE) REAL REDUCED ALPHA ARRAY
1416 C DET REAL DETERMINANT OF AMAT
1417 C E(20) REAL ELEMENT ENERGY LEVELS
1418 C NUMEL INT NUMBER OF ELEMENTS
1419 C S(20) REAL ACOUSTIC ENERGY INPUT
1420 C SIZE INT SIZE OF AMAT AND SMAT
1421 C SMAT(SIZE) REAL REDUCED S ARRAY
1422 C
1423 C
1424 C THE FOLLOWING COMMON BLOCKS ARE USED:
1425 C BLOCK OTHER PROGRAM UNITS USING THIS COMMON BLOCK
1426 C -----
1427 C CB7 EPROP, JINPUT, JPROP, EXCITE, ANSWER, RITER, BLOCK DATA
1428 C CB15 EXCITE, ANSWER
1429 C CB16 ANSWER
1430 C
1431 C
1432 C INTEGER SIZE
1433 C DIMENSION AMAT(SIZE,SIZE), SMAT(SIZE)
1434 C COMMON /CB7/ NUMEL,6
1435 C COMMON /CB15/ S(20),E(20)
1436 C COMMON /CB16/ ALPHA(20,20)
1437 C
1438 C 300 FORMAT (19.999, ERROR ***) THE DETERMINANT OF THE SEA IS
1439 C 1 EQUATION MATRIX IS 0. IF HENCE THERE IS NO SOLUTION.
1440 C 2 IF THIS ERROR WILL CAUSE THE PROGRAM TO ABORT.
1441 C 3 IF THIS ERROR WAS DISCOVERED BY SUBROUTINE SOLVE.
1442 C INITIALIZE I1
1443 C I1 = 0
1444 C DO FOR EACH ELEMENT FOR WHICH ENERGY LEVELS ARE UNKNOWN
1445 C DO 100 I = 1, SIZE
1446 C INCREMENT I1
1447 C I1 = I1 + 1
1448 C IF THE ENERGY LEVEL FOR ELEMENT I1 IS KNOWN, GO TO NEXT ELEMENT
1449 C IF (E(I1) .NE. 0.) GO TO 10
1450 C ELSE INITIALIZE J1
1451 C J1 = 0
1452 C DO FOR EACH ELEMENT FOR WHICH ENERGY LEVELS ARE UNKNOWN
1453 C DO 200 J = 1, SIZE
1454 C INCREMENT J1
1455 C J1 = J1 + 1
1456 C IF ENERGY LEVEL FOR ELEMENT J1 IS KNOWN, GO TO NEXT ELEMENT

```

SOLVE 2  
 SOLVE 3  
 SOLVE 4  
 SOLVE 5  
 SOLVE 6  
 SOLVE 7  
 MAY13 75  
 SOLVE 9  
 SOLVE 10  
 SOLVE 11  
 SOLVE 12  
 SOLVE 13  
 SOLVE 14  
 SOLVE 15  
 SOLVE 16  
 SOLVE 17  
 SOLVE 18  
 SOLVE 19  
 SOLVE 20  
 SOLVE 21  
 SOLVE 22  
 MAY13 76  
 SOLVE 24  
 SOLVE 25  
 SOLVE 26  
 SOLVE 27  
 SOLVE 28  
 SOLVE 29  
 SOLVE 30  
 SOLVE 31  
 SOLVE 32  
 SOLVE 33  
 MAY13 77  
 SOLVE 35  
 SOLVE 36  
 SOLVE 37  
 SOLVE 38  
 SOLVE 39  
 SOLVE 40  
 MAY13 79  
 SOLVE 42  
 SOLVE 43  
 SOLVE 44  
 SOLVE 45  
 SOLVE 46  
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 SOLVE 57

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1485

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      IF (E(J1) .NE. 0.) GO TO 30
      C ELSE PUT THE VALUE OF ALPHA FOR ELEMENT PAIR I1,J1 IN AMAT
      AMAT(I1,J1) = ALPHA(I1,J1)
      C END IF
      DO CONTINUE
      C PUT THE VALUE OF S FOR ELEMENT I1 IN SPAT
      SPAT(I1) = S(I1)
      DO CONTINUE
      C SOLVE THE MATRIX EQUATION
      CALL GASSEP (AMAT,SIZE,1,DET,SMAT)
      C IF THE DETERMINANT OF AMAT IS 0+
      IF (DET .NE. 0.) GO TO 152
      C THEN WRITE AN ERROR MESSAGE AND TERMINATE THE PROGRAM
      WRITE (6,380)
      CALL FERR
      C ELSE INITIALIZE I1 TO 1, REPRESENTING THE FIRST VALUE
      C RETURNED BY GASSEP TO ARRAY SMAT
      I1 = 1
      C DO FOR EACH ELEMENT
      DO 200 I = 1, NMUEL
      C IF THE ELEMENT ENERGY LEVEL WAS UNKNOWN,
      IF (E(I) .NE. 0.) GO TO 200
      C THEN PLACE ITS VALUE IN THE ARRAY E
      E(I) = SMAT(I1)
      C INCREMENT I1
      I1 = I1 + 1
      C END IF
      DO CONTINUE
      RETURN
      END

```

SOLVE 5A  
SOLVE 5B  
SOLVE 60  
SOLVE 61  
SOLVE 62  
SOLVE 63  
SOLVE 64  
SOLVE 65  
SOLVE 66  
UNIVAC 4  
SOLVE 68  
UNIVAC 5  
SOLVE 70  
SOLVE 71  
MAY13 82  
SOLVE 73  
SOLVE 74  
SOLVE 75  
SOLVE 76  
SOLVE 77  
SOLVE 78  
SOLVE 79  
SOLVE 80  
SOLVE 81  
SOLVE 82  
SOLVE 83  
SOLVE 84  
SOLVE 85  
SOLVE 86  
SOLVE 87

```

1486 SUBROUTINE RITER
1487 C THIS SUBROUTINE WRITES OUT THE RESULTS THAT ARE STORED IN
1488 C ARRAY ARAR. RITER IS CALLED FROM THE MAIN PROGRAM.
1489 C
1490 C
1491 C THE FOLLOWING VARIABLES ARE USED IN THIS PROGRAM UNIT:
1492 C NAME TYPE DESCRIPTION
1493 C ----
1494 C ARAR(20,20) REAL AVERAGE ACCELERATION
1495 C FTAB(40) REAL TABLE OF ANALYSIS FREQUENCIES
1496 C NUMAF INT NUMBER OF ANALYSIS FREQUENCIES
1497 C ANUPEL INT NUMBER OF ELEMENTS
1498 C OTYPE CHAR TYPE OF OUTPUT
1499 C
1500 C
1501 C THE FOLLOWING COMMON BLOCKS ARE USED:
1502 C BLOCK OTHER PROGRAM UNITS USING THIS COMMON BLOCK
1503 C
1504 C CB1 MAIN,EPROP,BEAM,MEMBR,PLATE,ROOM,CYLIN,BLOCK DATA
1505 C CB7 EPROP,INPUT,JPROP,EXCITE,ANSWER,SOLVE,BLOCK DATA
1506 C CB8 EPROP,ANSWER,BLOCK DATA
1507 C CB17 ANSWER
1508 C
1509 C
1510 CHARACTER*20 TITLE / IPSD LEVELS (G**2/MZ)** /,
1511 * RMS / 9 GRMS
1512 COMMON /CB1/ FTAB(40),NUMAF,FREQ1
1513 COMMON /CB7/ NUMEL,6
1514 COMMON /CB8/ SLOPE(20),SFREQ(20),ETA(20),OTYPE
1515 COMMON /CB17/ ABAR(20,40)
1516 100 FORMAT (//1//5X,10CENTER1,24X,A20/
1517 1 4X,9FREQ(20),4(5X,1ELEMENT 1,12:))
1518 110 FORMAT (F12.2,4(1PE15.5E2))
1519 C IF THE TYPE OF OUTPUT IS RMS, CHANGE THE HEADING ACCORDINGLY
1520 IF (OTYPE.EQ.'RMS') TITLE = RMS
1521 C DO FOR EACH ELEMENT IN GROUPS OF 4
1522 DO 20 I = 1, NUMEL, 4
1523 C IPLUS3 WILL BE THE TERMINAL VARIABLE FOR IMPLIED DO LOOPS TO
1524 C PRODUCE 4 COLUMNS UNLESS THERE ARE FEWER THAN 4 ELEMENTS
1525 C REPAIRING TO BE LISTED
1526 IPLUS3 = MIN(NUMEL, I + 3)
1527 C WRITE THE HEADING FOR THESE 4 ELEMENTS
1528 WRITE (6,100) TITLE,'J,J=I,IPLUS3)
1529 C DO FOR EACH ANALYSIS FREQUENCY
1530 DO 10 J = 1, NUMAF
1531 C WRITE THE RESULTS
1532 WRITE (6,110) FTAB(J), (ARAR(K,J),K=I,IPLUS3)
1533 10 CONTINUE
1534 20 CONTINUE
1535 RETURN
1536 ENC

```

```

RITER 2
RITER 3
RITER 4
RITER 5
RITER 6
RITER 7
RITER 8
RITER 9
RITER 10
RITER 11
RITER 12
RITER 13
MAY14 5
RITER 14
RITER 15
RITER 16
RITER 17
RITER 18
RITER 19
RITER 20
RITER 21
RITER 22
RITER 23
RITER 24
UNIVAC 6
UNIVAC 7
RITER 27
RITER 28
UNIVAC 8
RITER 30
RITER 31
RITER 32
RITER 33
RITER 34
RITER 35
RITER 36
RITER 37
RITER 38
RITER 39
RITER 40
RITER 41
RITER 42
MAY13 88
RITER 44
RITER 45
RITER 46
RITER 47
RITER 48
RITER 49
RITER 50
RITER 51

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1537      PLCKC DATA
1538
1539      C THE FOLLOWING VARIABLES ARE USED IN THIS PROGRAM UNIT:
1540
1541      C NAME TYPE DESCRIPTION
1542      C ----
1543      C ERROR LOG TRUE IF A FATAL ERROR HAS OCCURRED
1544      C FTAB(40) REAL TABLE OF ANALYSIS FREQUENCIES
1545      C G REAL GRAVITATIONAL CONSTANT
1546      C INPUT INT NUMBER OF INPUT RECORDS READ
1547      C MASS(20) REAL MASS
1548      C M(20,40) REAL POBAL DENSITY
1549      C SLOPE REAL SLOPE
1550
1551      C THE FOLLOWING COMMON BLOCKS ARE USED:
1552
1553      C BLOCK OTHER PROGRAM UNITS USING THIS COMMON BLOCK
1554      C ----
1555      C CB1 MAIN,EPROP,BEAM,PLATE,ROOM,CYLIN,RITER
1556      C CB2 MAIN,EPROP,JINPUT
1557      C CB7 EPROP,JINPUT,JPROP,EXCITE,ANSWER,SOLVE,RITER
1558      C CB9 EPROP,ANSWER,LITER
1559      C CB10 EPROP,BEAM,MEMBR,PLATE,ROOM,CYLIN,JPROP,EXCITE
1560      C CB11 BEAM,MEMBR,PLATE,ROOM,CYLIN,ANSWER
1561
1562      REAL M,MASS
1563      LOGICAL ERROR
1564      COMMON /CB1/ FTAB(40),NUMMF,FREQ1
1565      COMMON /CB2/ ERROR
1566      COMMON /CB7/ NUMEL,G
1567      COMMON /CB8/ SLOPE(20),SFREQ(20),ETA(20),OTYPE
1568      COMMON /CB9/ M(20,40)
1569      COMMON /CB10/ INPUT
1570      COMMON /CB11/ MASS(20)
1571      DATA FTAB /1.0,1.25,1.6,2.0,2.5,3.15,4.0,5.0,6.3,8.0,
1572      1 10.0,12.5,16.0,20.0,25.0,31.5,40.0,50.0,63.0,80.0,
1573      2 100.0,125.0,160.0,200.0,250.0,315.0,400.0,500.0,
1574      3 630.0,800.0,1000.0,1250.0,1600.0,2000.0,2500.0,
1575      4 3150.0,4000.0,5000.0,6300.0,8000.0/
1576      END
1577
1578      RIVER 52
1579      RIVER 53
1580      RIVER 54
1581      RIVER 55
1582      RIVER 56
1583      RIVER 57
1584      RIVER 58
1585      RIVER 59
1586      RIVER 60
1587      RIVER 61
1588      RIVER 62
1589      RIVER 63
1590      RIVER 64
1591      RIVER 65
1592      RIVER 66
1593      RIVER 67
1594      RIVER 68
1595      RIVER 69
1596      RIVER 70
1597      RIVER 71
1598      RIVER 72
1599      RIVER 73
1600      RIVER 74
1601      RIVER 75
1602      RIVER 76
1603      RIVER 77
1604      RIVER 78
1605      RIVER 79
1606      RIVER 80
1607      RIVER 81
1608      UNIVAC 9
1609      RIVER 83
1610      RIVER 84
1611      RIVER 85
1612      RIVER 86
1613      RIVER 87
1614      RIVER 88
1615      RIVER 89
1616      RIVER 90
1617      RIVER 91
1618      RIVER 92

```

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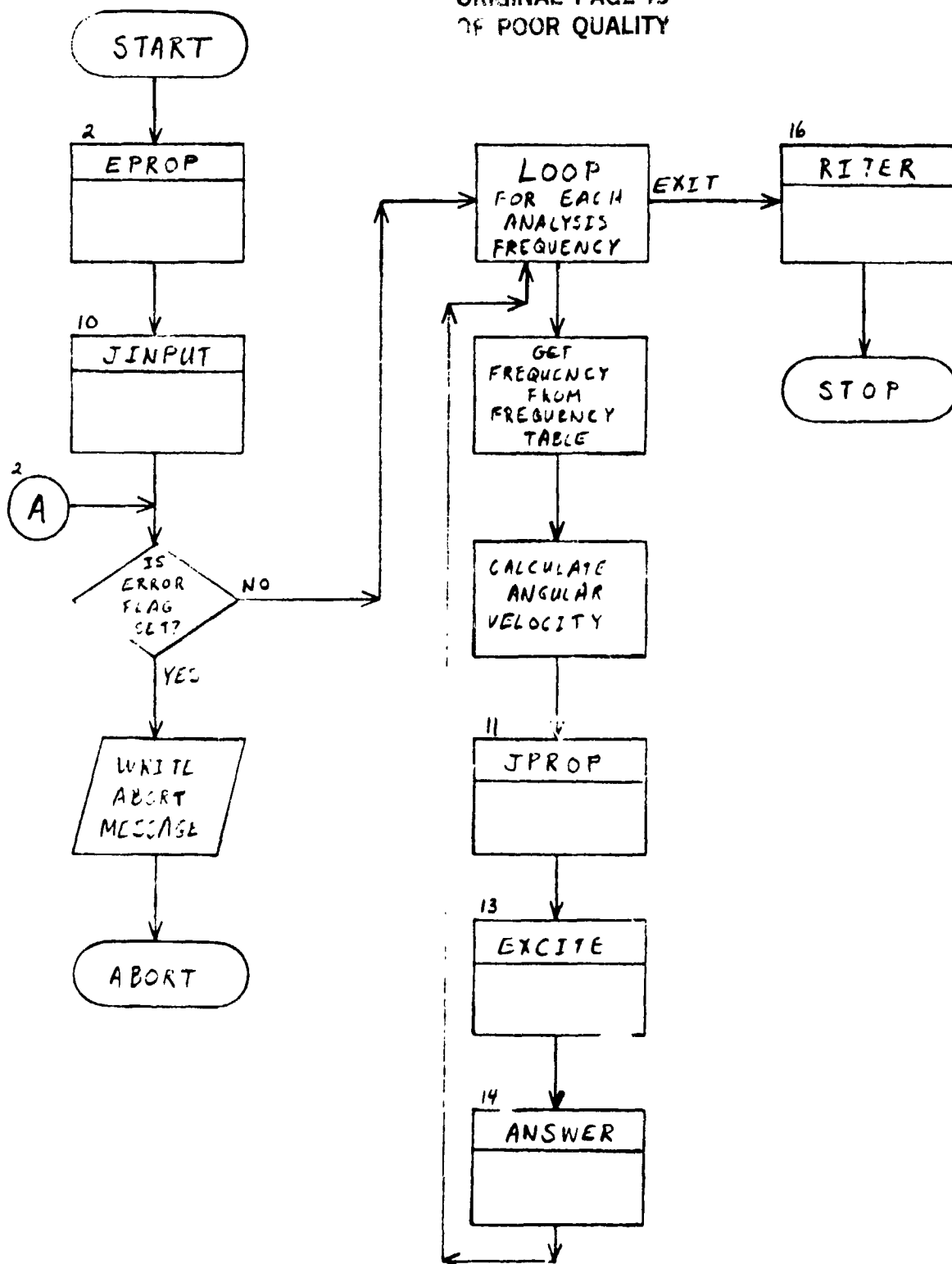
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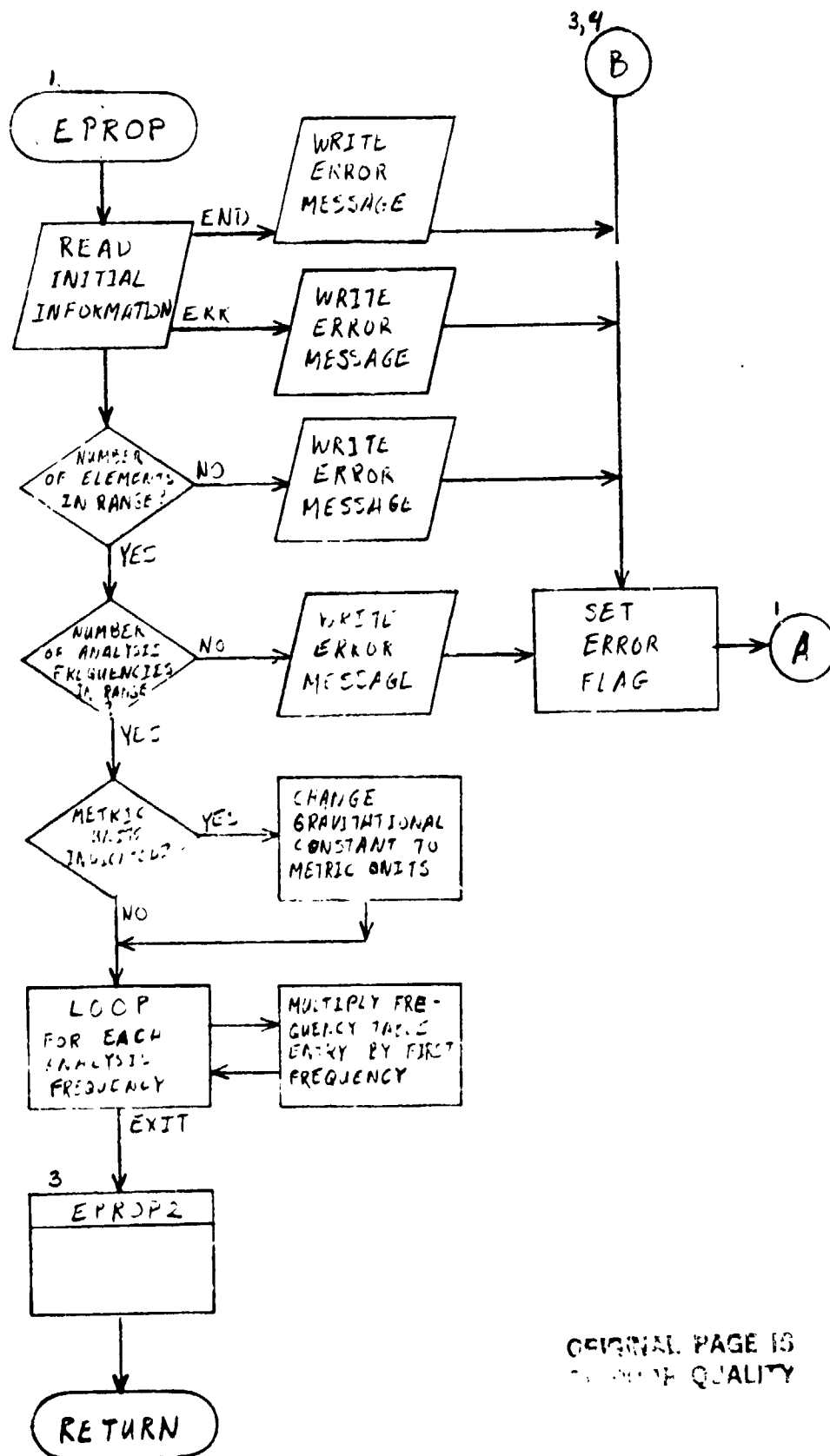


## Appendix II

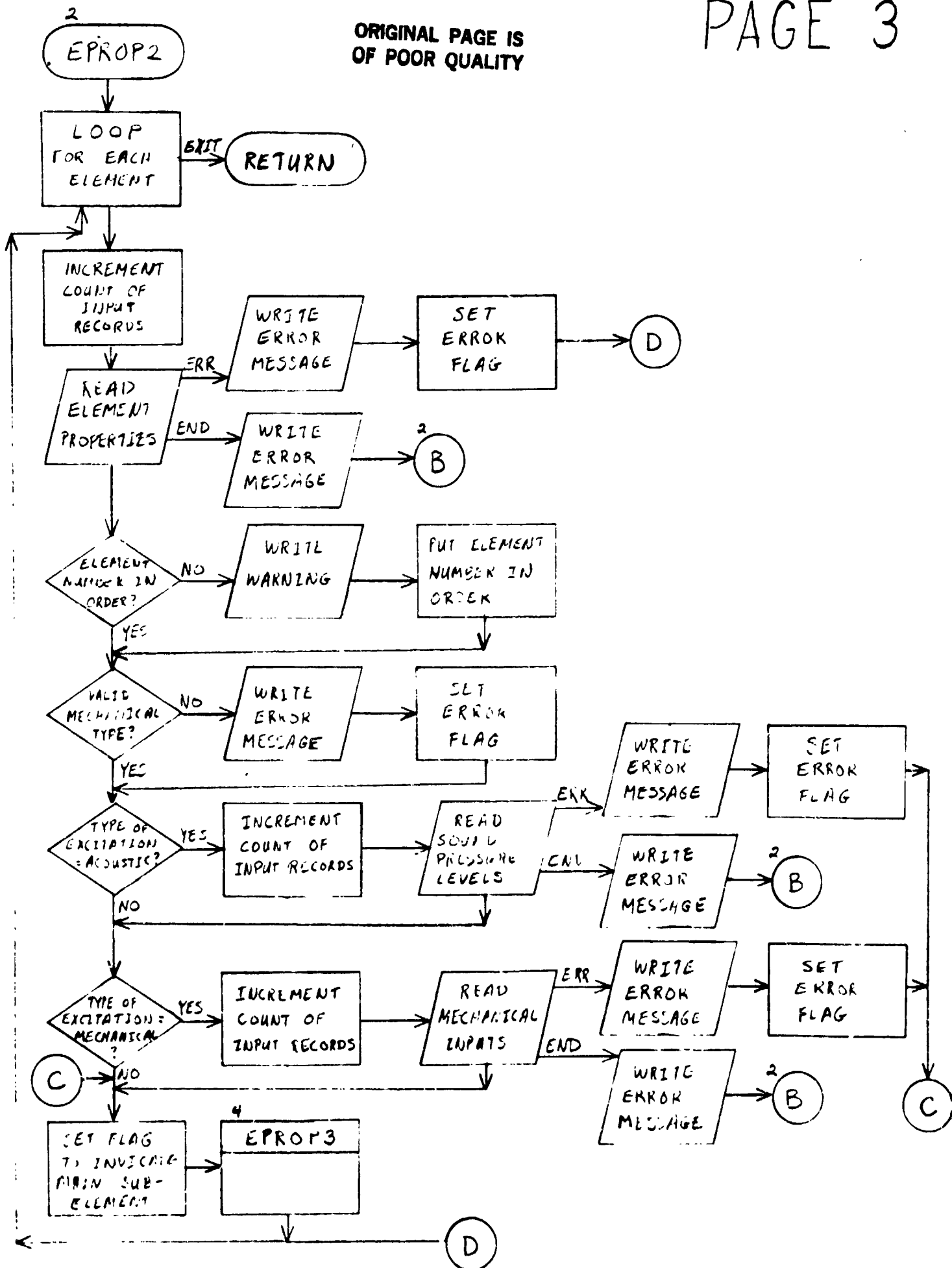
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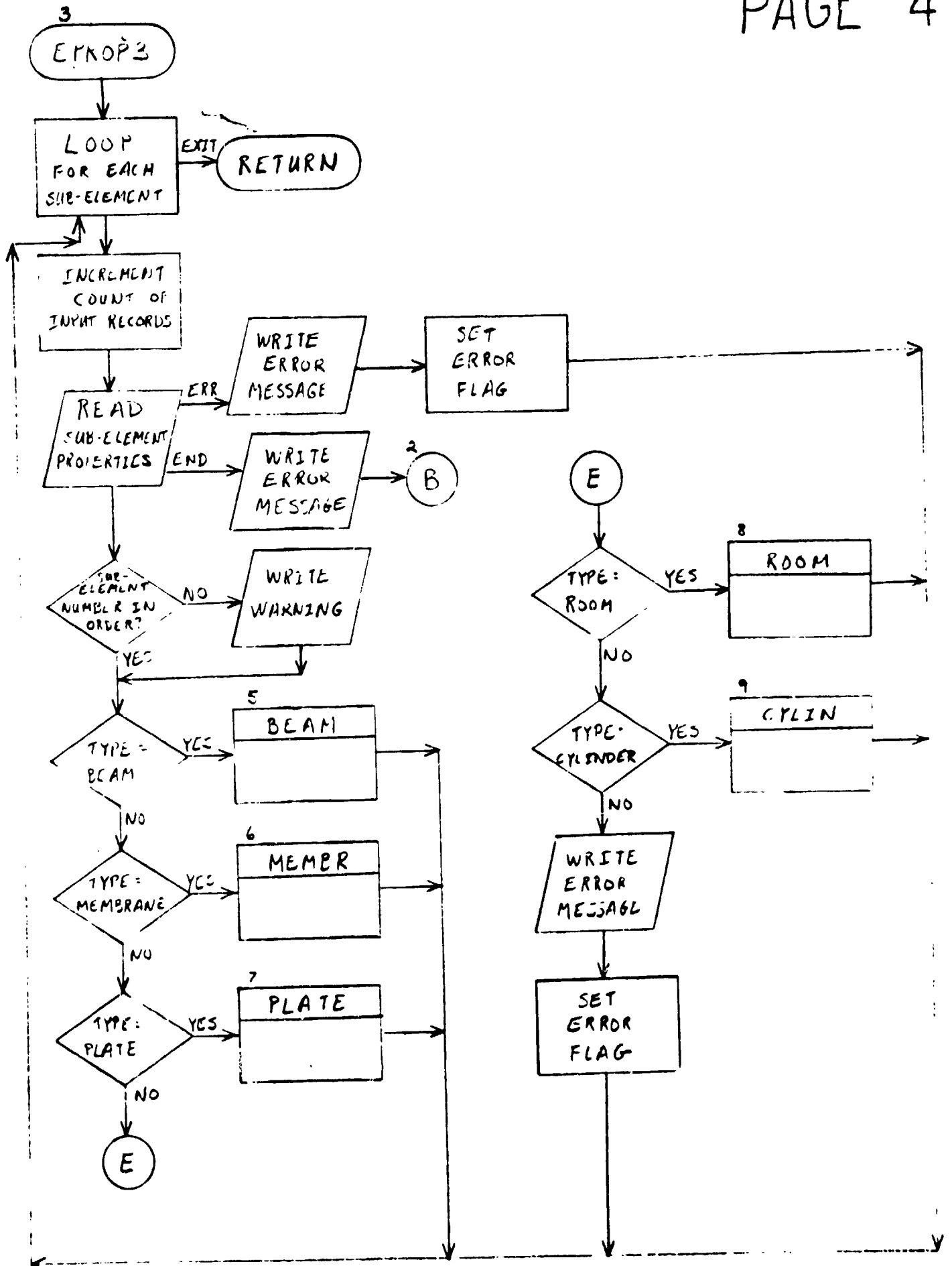
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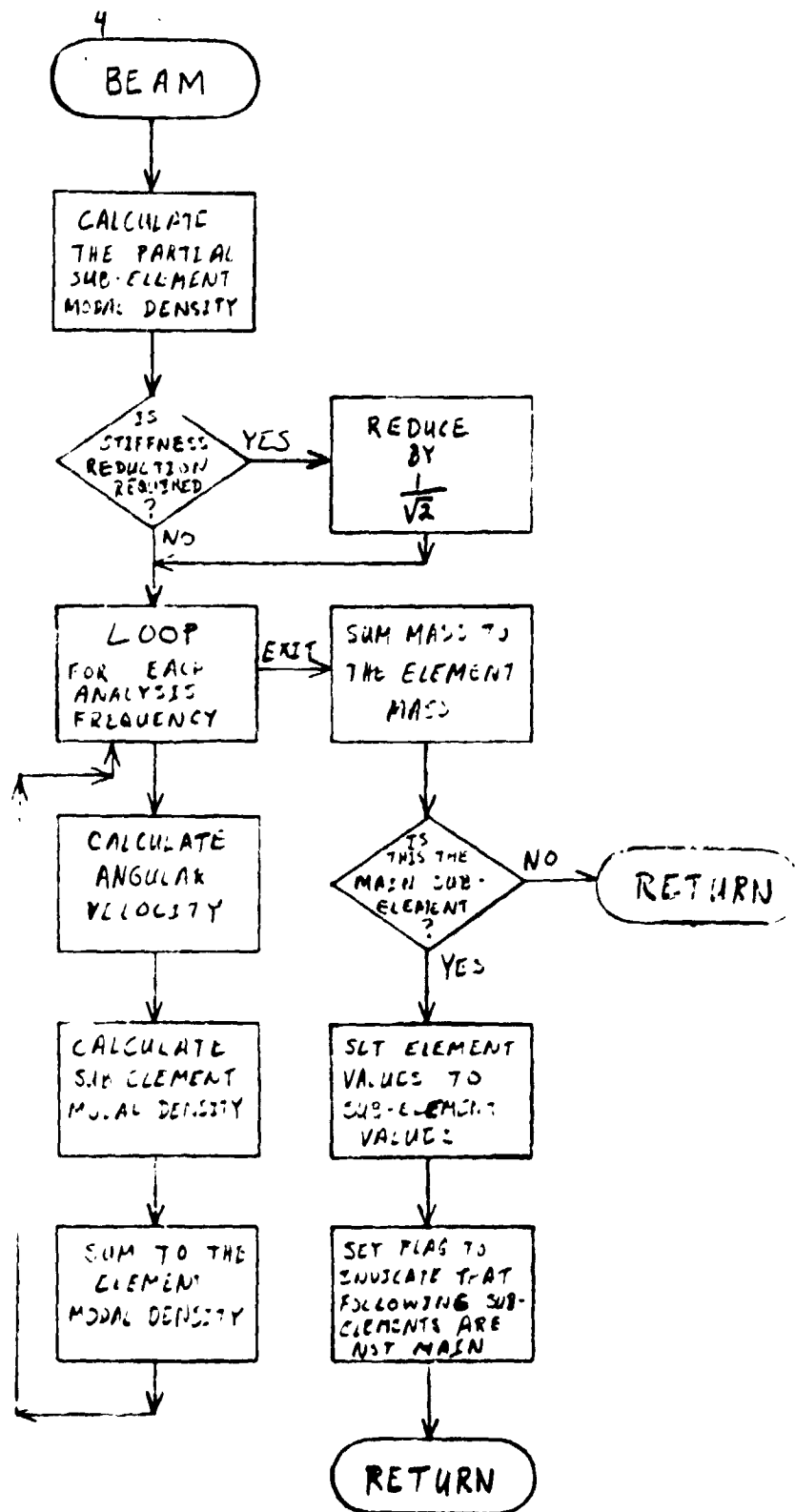


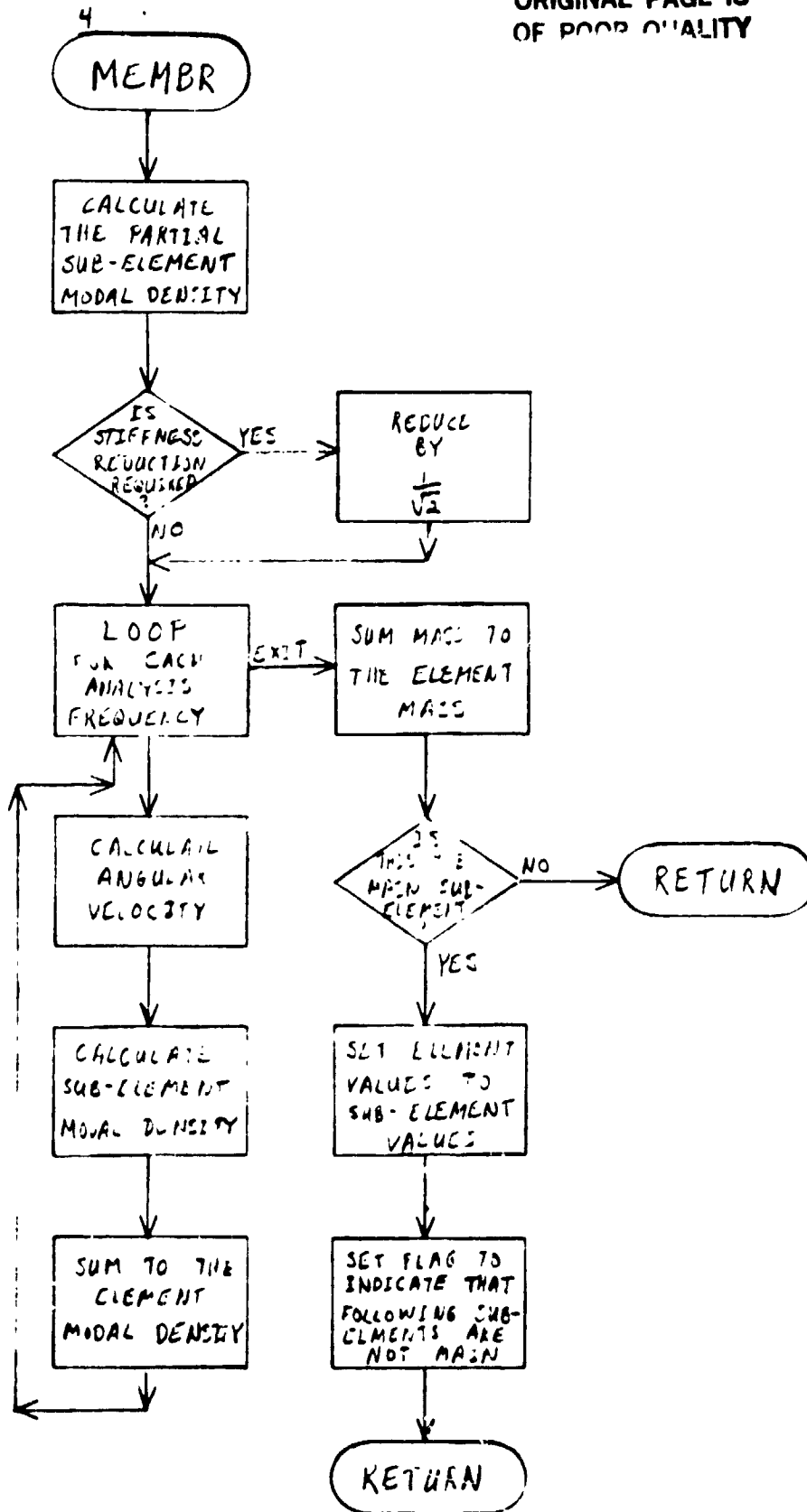


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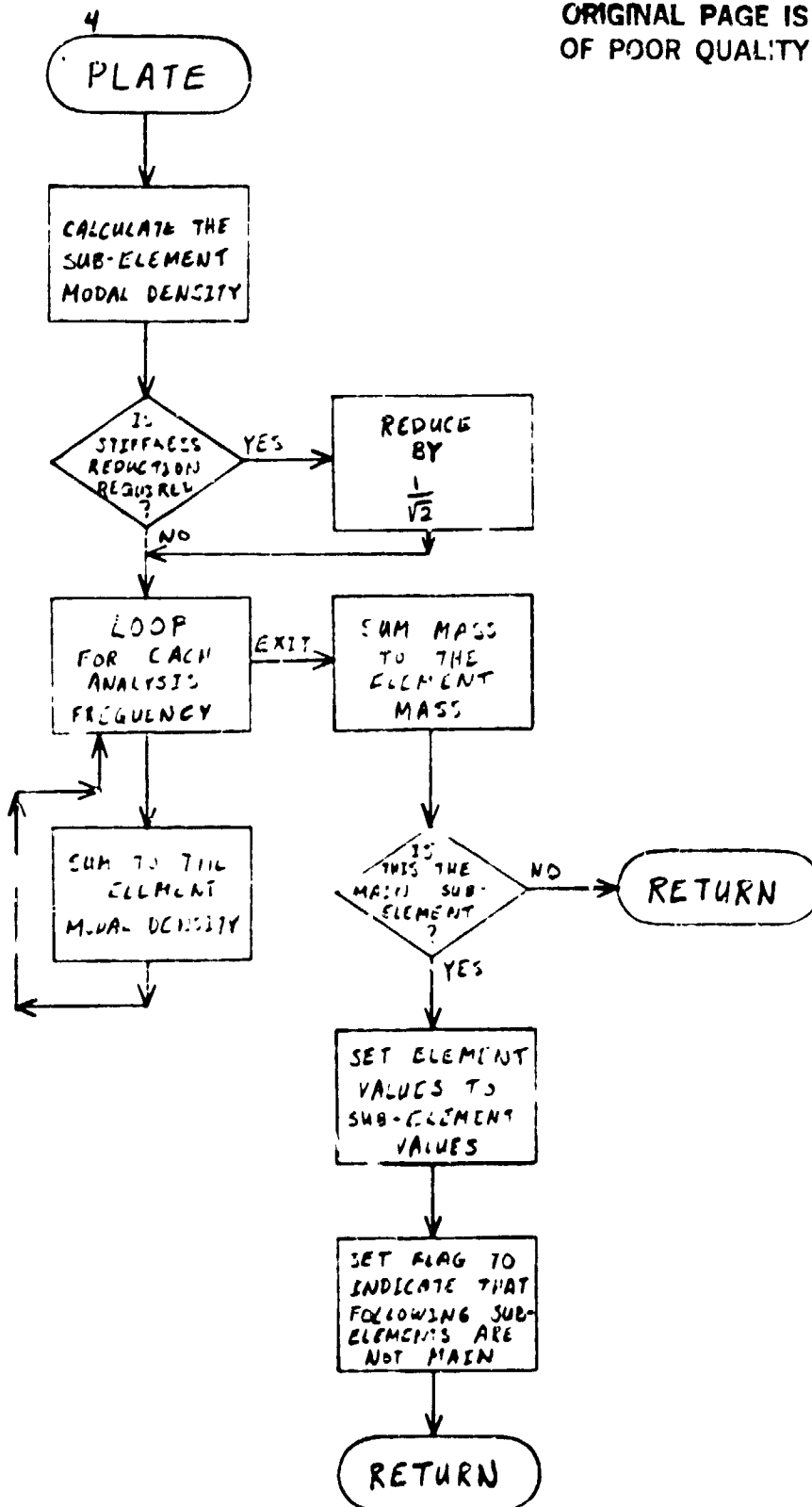




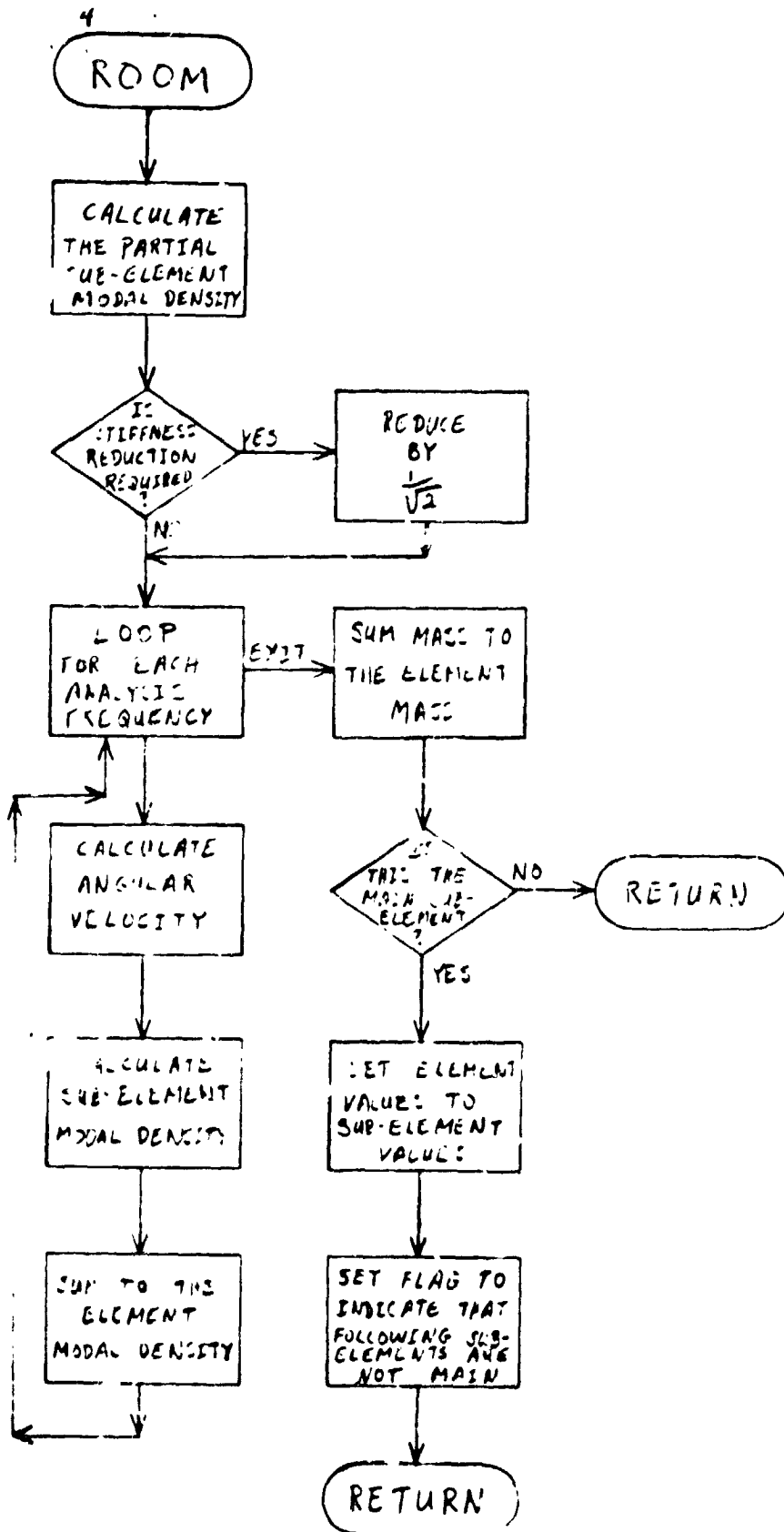




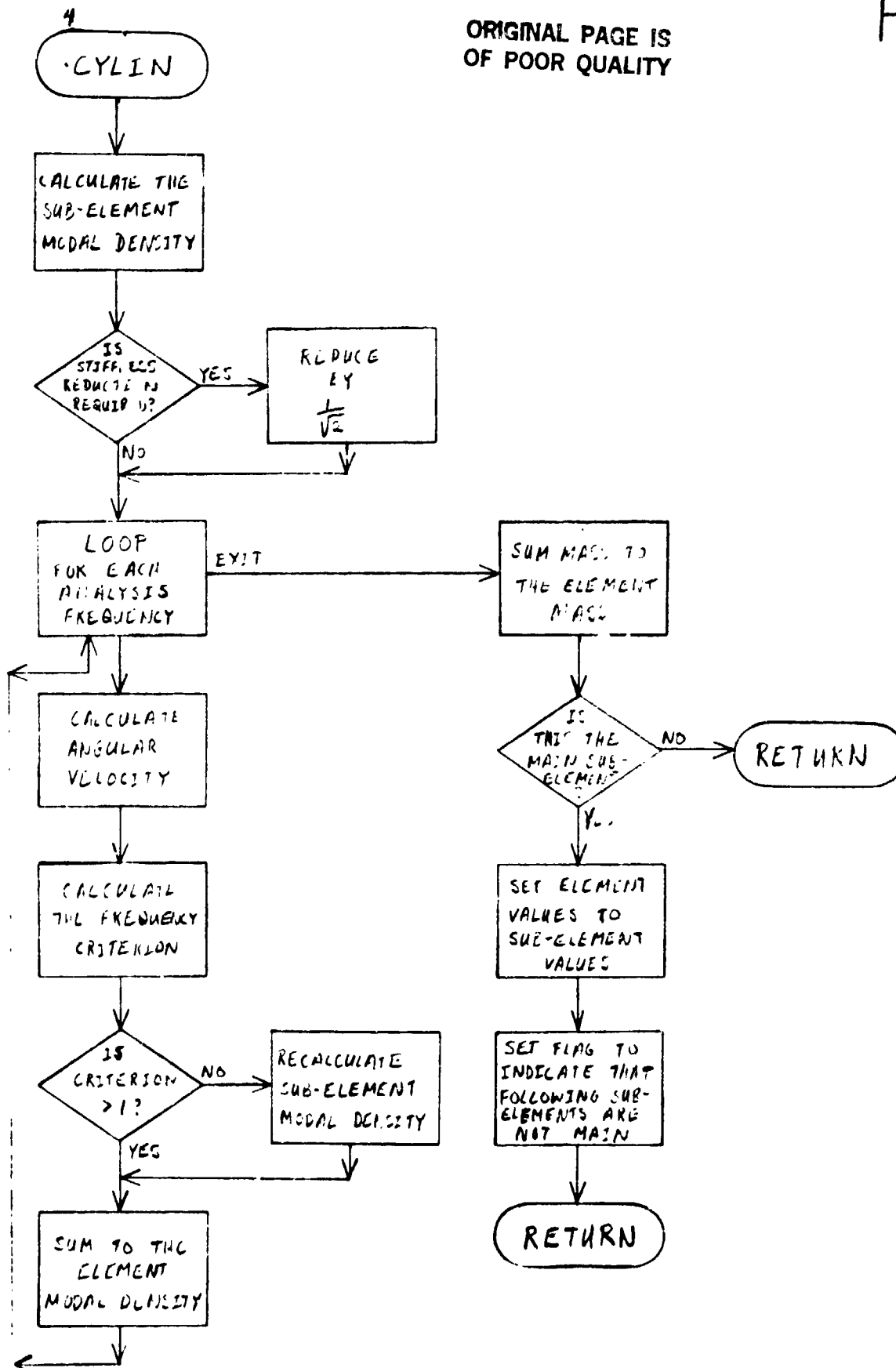
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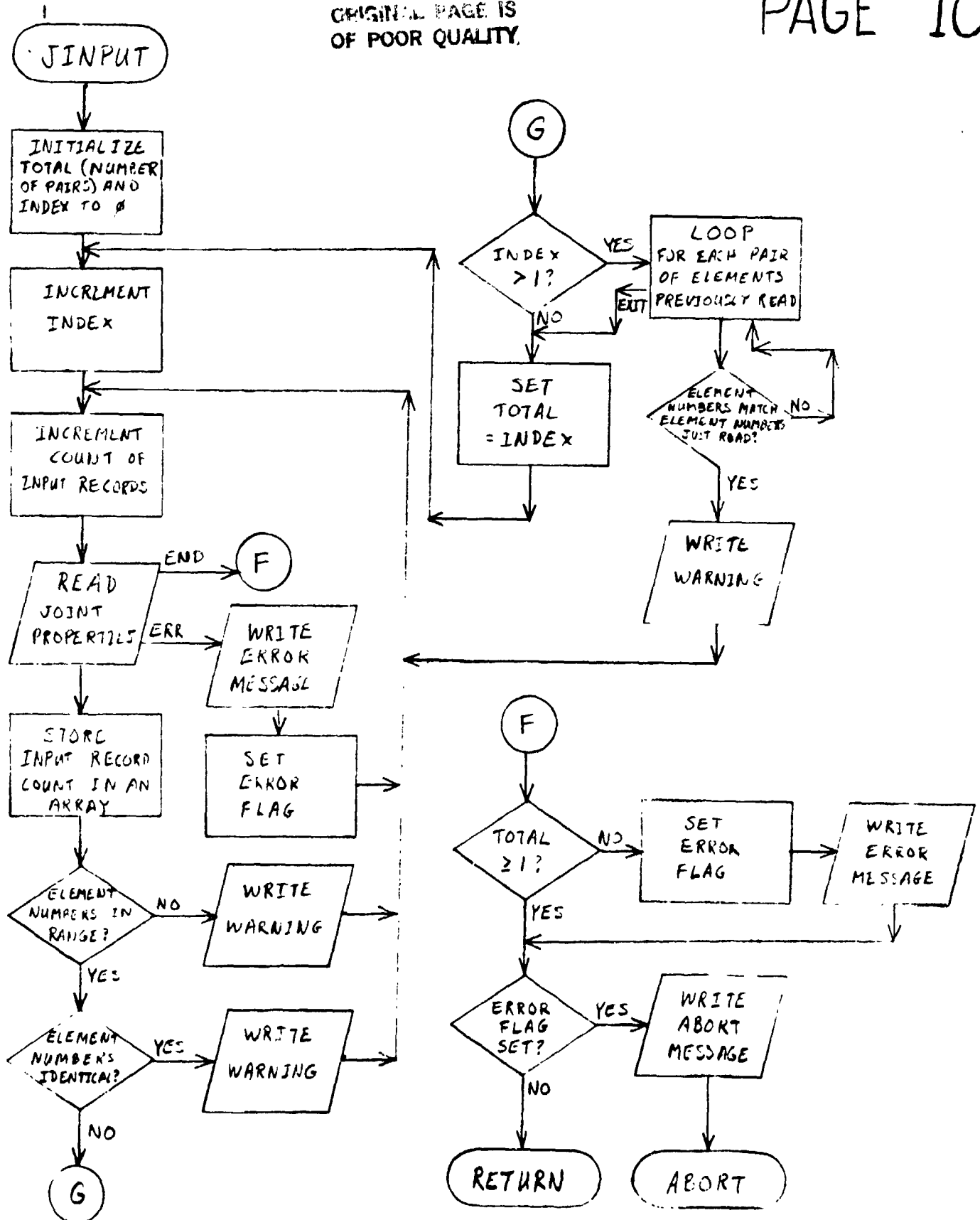


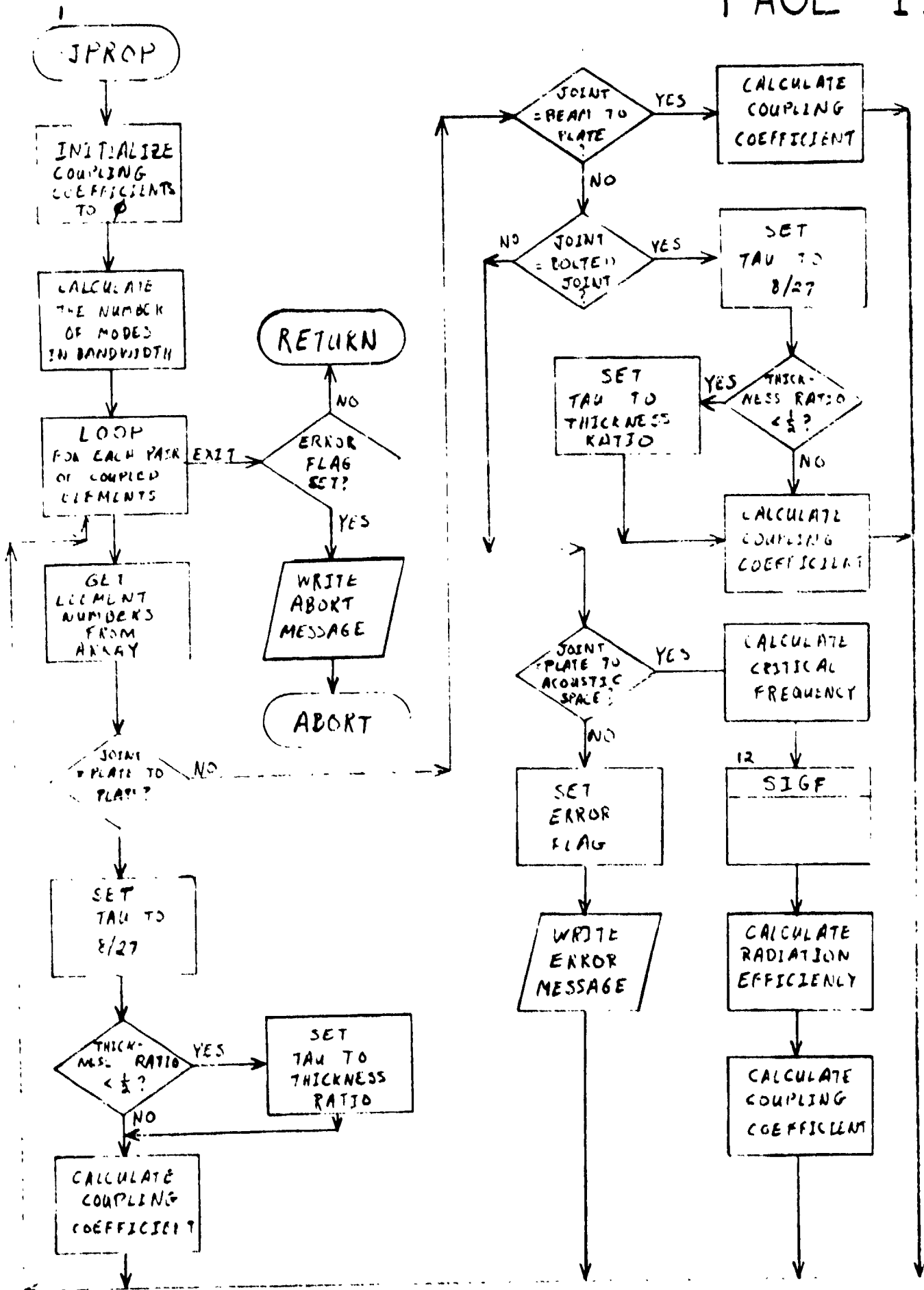


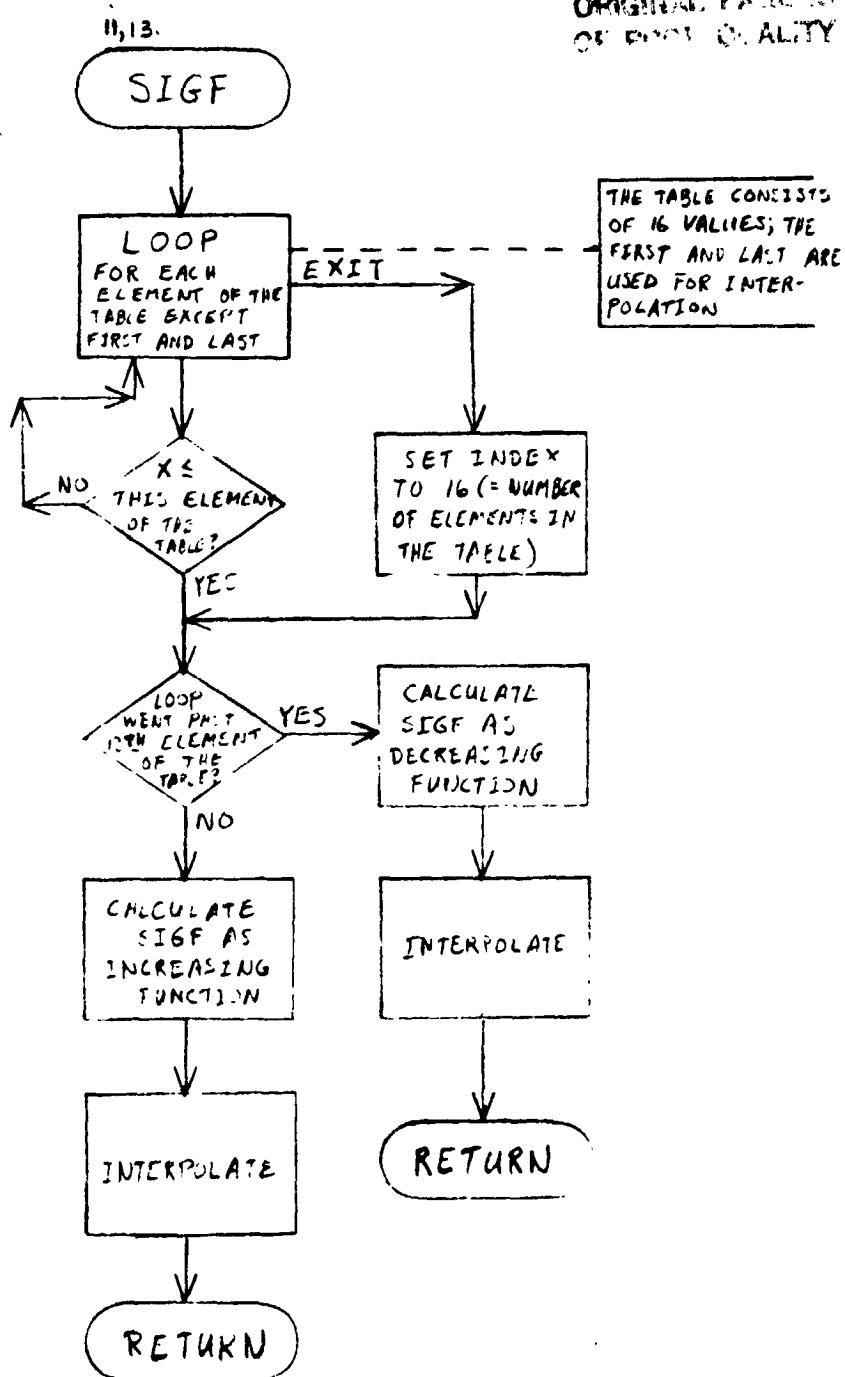


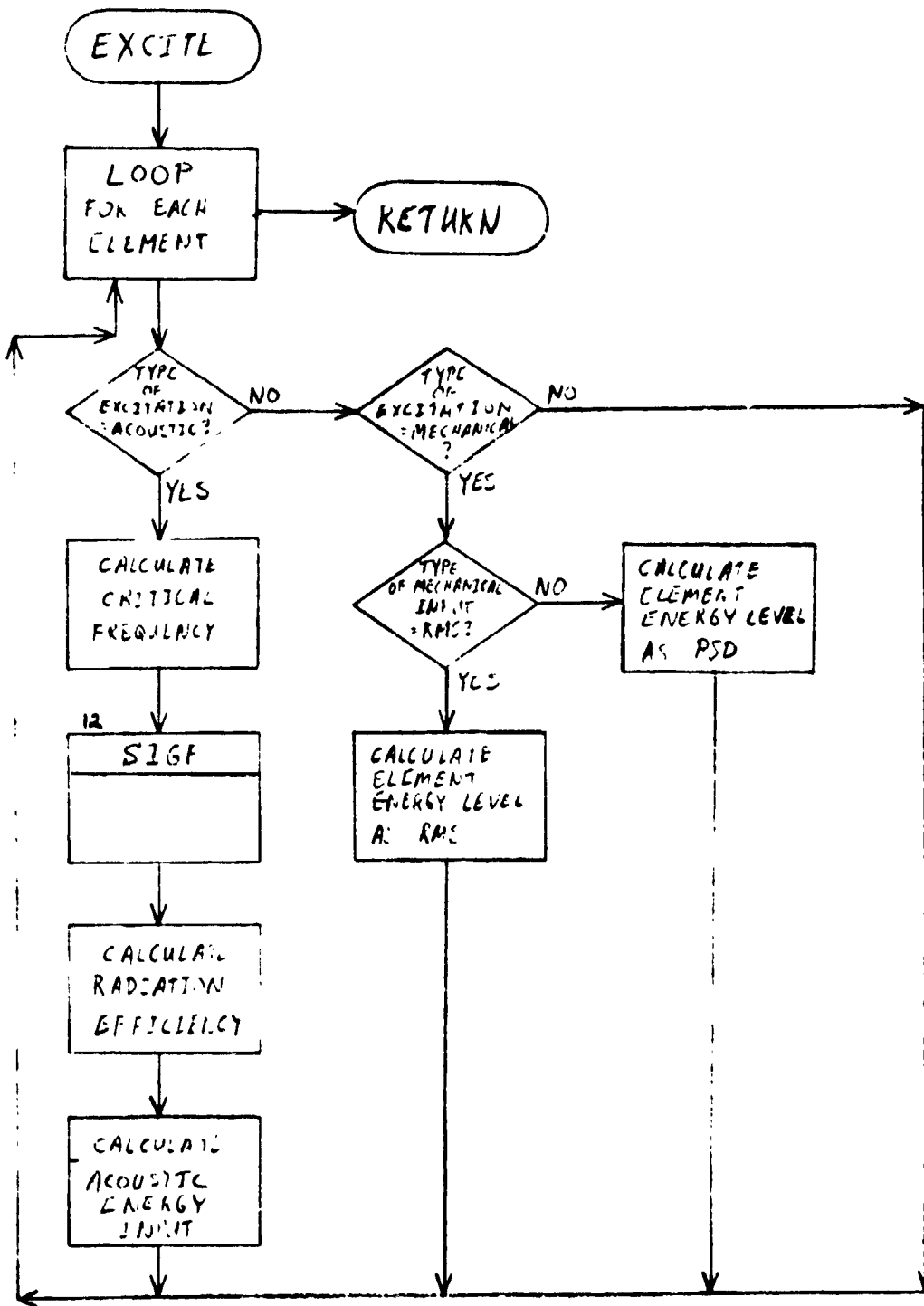
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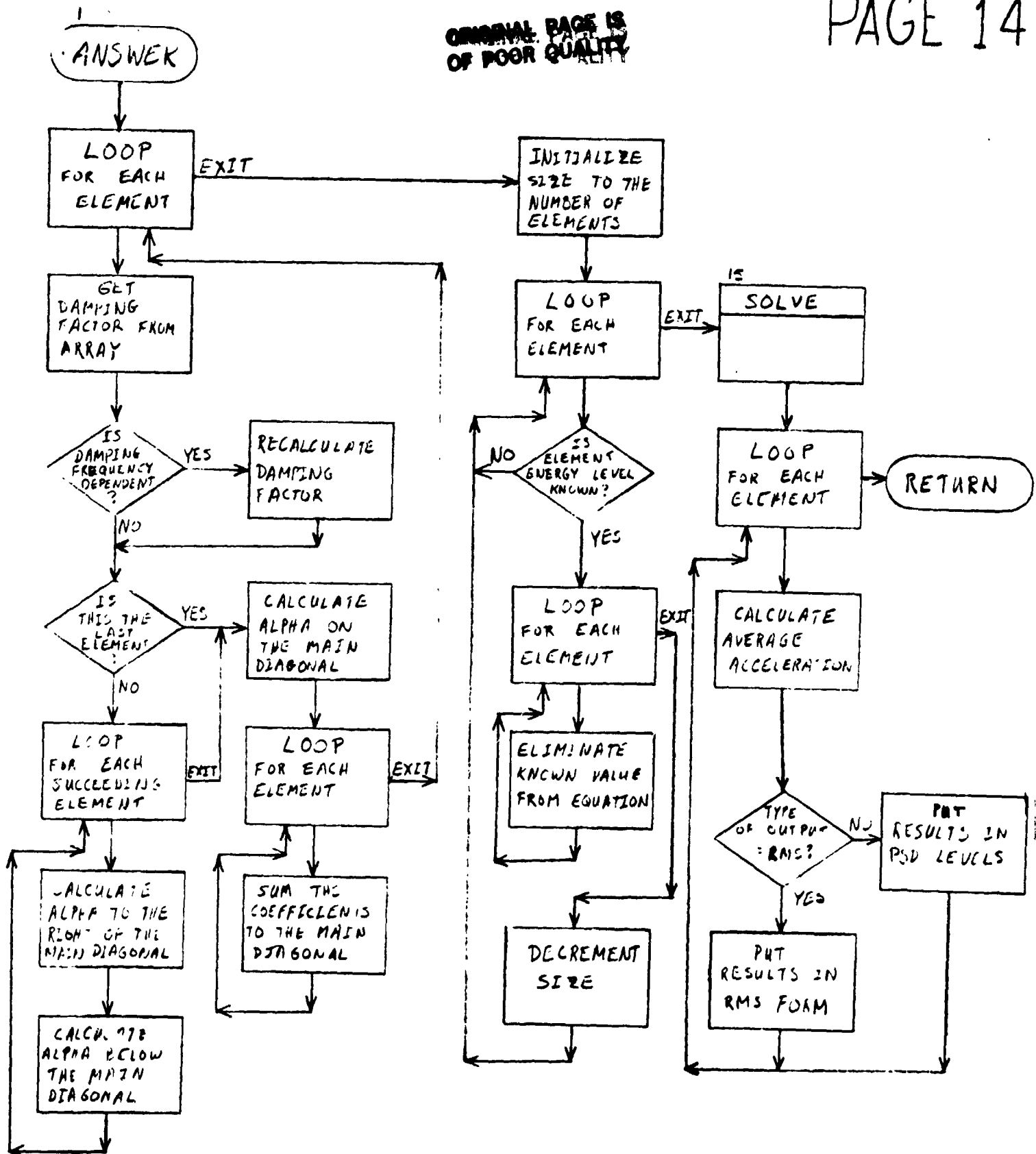


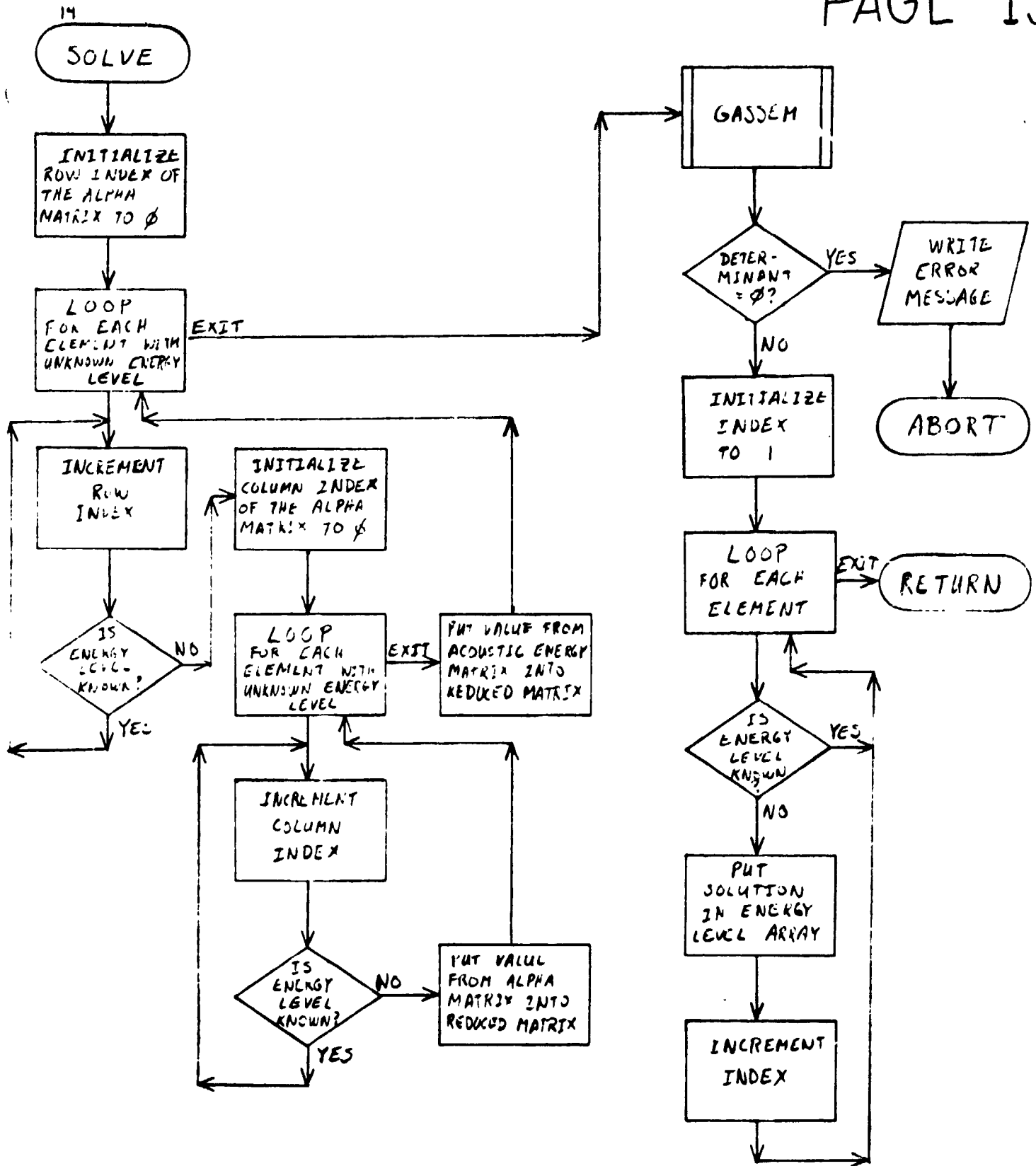




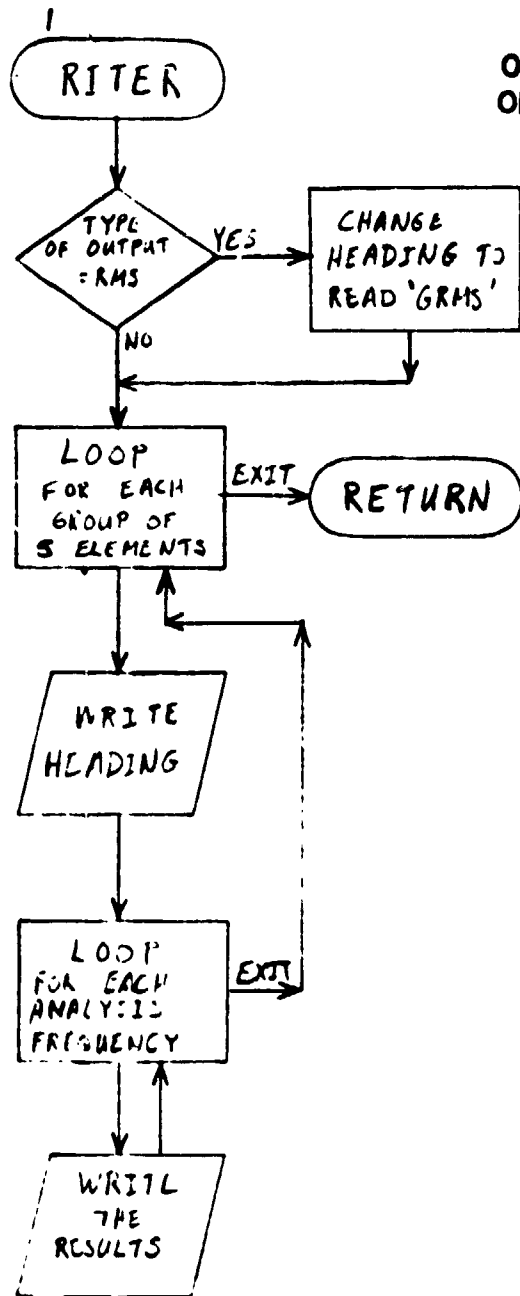


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Appendix III

SEA PROGRAM INPUT LIST

MATERIALS EXPERIMENT ASSEMBLY - EXAMPLE 2

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STATISTICAL ENERGY ANALYSIS OF COMPLEX STRUCTURES

RECORD  
NUMBER

DATA READ FROM UNIT 3

1 NUMBER OF ELEMENTS = 6  
NUMBER OF ANALYSIS FREQUENCIES = 23  
FIRST ANALYSIS FREQUENCY = 3.15000E+01  
TYPE OF UNITS =  
TYPE OF OUTPUT = PSD  
2 ELEMENT NUMBER = 1  
NUMBER OF SUB-ELEMENTS = 20  
TYPE OF EXCITATION = A  
TYPE OF MECHANICAL INPUT =  
DAMPING = 1.00000E-02  
SLOPE = 0.  
STARTING FREQUENCY = 0.  
SOUND PRESSURE LEVELS  
3 1.20000E+02 1.22000E+02 1.24000E+02 1.26000E+02  
1.28000E+02 1.30000E+02 1.32000E+02 1.33000E+02  
4 1.34500E+02 1.35000E+02 1.35000E+02 1.35000E+02  
1.35000E+02 1.33000E+02 1.31000E+02 1.27000E+02  
5 1.26000E+02 1.25000E+02 1.23000E+02 1.21000E+02  
1.19000E+02 1.17000E+02 1.15000E+02 0.  
6 SUB-ELEMENT NUMBER = 1  
TYPE OF SUB-ELEMENT = P  
DENSITY = 2.61700E-04  
MODULUS OF ELASTICITY = 1.00000E+07  
THICKNESS = 3.70600E-02  
AREA = 2.35670E+04  
POISSONS RATIO = 3.30000E-01  
LENGTH = 0.  
PRESSURE = 0.  
7 STIFFNESS REDUCTION REQUIRED = F  
RADIUS = 0.  
VOLUME = 0.  
SPEED OF SOUND IN ROOM MEDIUM = 1.34000E+04  
ADDED MASS = 6.49000E-01  
8 SUB-ELEMENT NUMBER = 2  
TYPE OF SUB-ELEMENT = P  
DENSITY = 2.61700E-04  
MODULUS OF ELASTICITY = 1.00000E+07  
THICKNESS = 1.90000E-01  
AREA = 2.16500E+02  
POISSONS RATIO = 3.30000E-01  
LENGTH = 0.  
PRESSURE = 0.  
9 STIFFNESS REDUCTION REQUIRED = F  
RADIUS = 0.  
VOLUME = 0.  
SPEED OF SOUND IN ROOM MEDIUM = 0.  
ADDED MASS = 0.  
10 SUB-ELEMENT NUMBER = 3  
TYPE OF SUB-ELEMENT = P  
DENSITY = 2.61700E-04  
MODULUS OF ELASTICITY = 1.00000E+07  
THICKNESS = 9.50000E-02  
AREA = 3.31200E+03  
POISSONS RATIO = 3.30000E-01

LENGTH = 0.  
 PRESSURE = 0.  
 11 STIFFNESS REDUCTION REQUIRED = F  
 RADIUS = 0.  
 VOLUME = 0.  
 SPEED OF SOUND IN ROOM MEDIUM = C.  
 ADDED MASS = 0.  
 12 SUB-ELEMENT NUMBER = 4  
 TYPE OF SUB-ELEMENT = P  
 DENSITY = 2.61700E-04  
 MODULUS OF ELASTICITY = 1.00000E+07  
 THICKNESS = 2.50000E-01  
 AREA = 6.90000E+02  
 POISSONS RATIO = 3.30000E-01  
 LENGTH = 0.  
 PRESSURE = 0.  
 13 STIFFNESS REDUCTION REQUIRED = F  
 RADIUS = 0.  
 VOLUME = 0.  
 SPEED OF SOUND IN ROOM MEDIUM = C.  
 ADDED MASS = 0.  
 14 SUB-ELEMENT NUMBER = 5  
 TYPE OF SUB-ELEMENT = P  
 DENSITY = 2.61700E-04  
 MODULUS OF ELASTICITY = 1.00000E+07  
 THICKNESS = 9.50000E-02  
 AREA = 3.93000E+03  
 POISSONS RATIO = 3.30000E-01  
 LENGTH = 0.  
 PRESSURE = 0.  
 15 STIFFNESS REDUCTION REQUIRED = F  
 RADIUS = 0.  
 VOLUME = 0.  
 SPEED OF SOUND IN ROOM MEDIUM = C.  
 ADDED MASS = 0.  
 16 SUB-ELEMENT NUMBER = 6  
 TYPE OF SUB-ELEMENT = P  
 DENSITY = 2.61700E-04  
 MODULUS OF ELASTICITY = 1.00000E+07  
 THICKNESS = 2.50000E-01  
 AREA = 1.22813E+03  
 POISSONS RATIO = 3.30000E-01  
 LENGTH = 0.  
 PRESSURE = 0.  
 17 STIFFNESS REDUCTION REQUIRED = F  
 RADIUS = 0.  
 VOLUME = 0.  
 SPEED OF SOUND IN ROOM MEDIUM = C.  
 ADDED MASS = 0.  
 18 SUB-ELEMENT NUMBER = 7  
 TYPE OF SUB-ELEMENT = P  
 DENSITY = 2.61700E-04  
 MODULUS OF ELASTICITY = 1.00000E+07  
 THICKNESS = 2.00000E-01  
 AREA = 1.74720E+03  
 POISSONS RATIO = 3.30000E-01  
 LENGTH = 0.  
 PRESSURE = 0.  
 19 STIFFNESS REDUCTION REQUIRED = F  
 RADIUS = 0.  
 VOLUME = 0.  
 SPEED OF SOUND IN ROOM MEDIUM = C.  
 ADDED MASS = 0.  
 20 SUB-ELEMENT NUMBER = 8  
 TYPE OF SUB-ELEMENT = P  
 DENSITY = 2.61700E-04

MODULUS OF ELASTICITY = 1.00000E+07  
 THICKNESS = 4.64000E-02  
 AREA = 2.84637E+03  
 POISSONS RATIC = 3.30000E-01  
 LENGTH = 0.  
 PRESSURE = 0.  
 STIFFNESS REDUCTION REQUIRED = F  
 RADIUS = 0.  
 VOLUME = 0.  
 SPEED OF SOUND IN ROOM MEDIUM = C.  
 ADDED MASS = 0.  
 SUB-ELEMENT NUMBER = 9  
 TYPE OF SUB-ELEMENT = P  
 DENSITY = 2.61700E-04  
 MODULUS OF ELASTICITY = 1.00000E+07  
 THICKNESS = 1.25000E-01  
 AREA = 7.47320E+02  
 POISSONS RATIC = 3.30000E-01  
 LENGTH = 0.  
 PRESSURE = 0.  
 STIFFNESS REDUCTION REQUIRED = F  
 RADIUS = 0.  
 VOLUME = 0.  
 SPEED OF SOUND IN ROOM MEDIUM = C.  
 ADDED MASS = 0.  
 SUB-ELEMENT NUMBER = 10  
 TYPE OF SUB-ELEMENT = P  
 DENSITY = 2.61700E-04  
 MODULUS OF ELASTICITY = 1.00000E+07  
 THICKNESS = 5.62000E-01  
 AREA = 6.00000E+01  
 POISSONS RATIC = 3.30000E-01  
 LENGTH = 0.  
 PRESSURE = 0.  
 STIFFNESS REDUCTION REQUIRED = F  
 RADIUS = 0.  
 VOLUME = 0.  
 SPEED OF SOUND IN ROOM MEDIUM = C.  
 ADDED MASS = 0.  
 SUB-ELEMENT NUMBER = 11  
 TYPE OF SUB-ELEMENT = P  
 DENSITY = 2.61700E-04  
 MODULUS OF ELASTICITY = 1.00000E+07  
 THICKNESS = 2.50000E-01  
 AREA = 1.65920E+03  
 POISSONS RATIC = 3.30000E-01  
 LENGTH = 0.  
 PRESSURE = 0.  
 STIFFNESS REDUCTION REQUIRED = F  
 RADIUS = 0.  
 VOLUME = 0.  
 SPEED OF SOUND IN ROOM MEDIUM = C.  
 ADDED MASS = 0.  
 SUB-ELEMENT NUMBER = 12  
 TYPE OF SUB-ELEMENT = P  
 DENSITY = 2.61700E-04  
 MODULUS OF ELASTICITY = 1.00000E+07  
 THICKNESS = 3.75000E-01  
 AREA = 5.03000E+01  
 POISSONS RATIC = 3.30000E-01  
 LENGTH = 0.  
 PRESSURE = 0.  
 STIFFNESS REDUCTION REQUIRED = F  
 RADIUS = 0.  
 VOLUME = 0.  
 SPEED OF SOUND IN ROOM MEDIUM = C.

30 ADDED MASS = 0.  
 SUB-ELEMENT NUMBER = 13  
 TYPE OF SUB-ELEMENT = P  
 DENSITY = 2.61700E-04  
 MODULUS OF ELASTICITY = 1.00000E+07  
 THICKNESS = 1.90000E-01  
 AREA = 3.54160E+02  
 POISSONS RATIO = 3.30000E-01  
 LENGTH = 0.  
 PRESSURE = 0.  
 31 STIFFNESS REDUCTION REQUIRED = F  
 RADIUS = 0.  
 VOLUME = 0.  
 SPEED OF SOUND IN ROOM MEDIUM = C.  
 32 ADDED MASS = 0.  
 SUB-ELEMENT NUMBER = 14  
 TYPE OF SUB-ELEMENT = P  
 DENSITY = 2.61700E-04  
 MODULUS OF ELASTICITY = 1.00000E+07  
 THICKNESS = 1.50000E-01  
 AREA = 7.36200E+02  
 POISSONS RATIO = 3.30000E-01  
 LENGTH = 0.  
 PRESSURE = 0.  
 33 STIFFNESS REDUCTION REQUIRED = F  
 RADIUS = 0.  
 VOLUME = 0.  
 SPEED OF SOUND IN ROOM MEDIUM = C.  
 34 ADDED MASS = 0.  
 SUB-ELEMENT NUMBER = 15  
 TYPE OF SUB-ELEMENT = P  
 DENSITY = 2.61700E-04  
 MODULUS OF ELASTICITY = 1.00000E+07  
 THICKNESS = 2.00000E-01  
 AREA = 1.98000E+02  
 POISSONS RATIO = 3.30000E-01  
 LENGTH = 0.  
 PRESSURE = 0.  
 35 STIFFNESS REDUCTION REQUIRED = F  
 RADIUS = 0.  
 VOLUME = 0.  
 SPEED OF SOUND IN ROOM MEDIUM = C.  
 36 ADDED MASS = 0.  
 SUB-ELEMENT NUMBER = 16  
 TYPE OF SUB-ELEMENT = P  
 DENSITY = 2.61700E-04  
 MODULUS OF ELASTICITY = 1.00000E+07  
 THICKNESS = 2.50000E-01  
 AREA = 1.07100E+02  
 POISSONS RATIO = 3.30000E-01  
 LENGTH = 0.  
 PRESSURE = 0.  
 37 STIFFNESS REDUCTION REQUIRED = F  
 RADIUS = 0.  
 VOLUME = 0.  
 SPEED OF SOUND IN ROOM MEDIUM = C.  
 38 ADDED MASS = 0.  
 SUB-ELEMENT NUMBER = 17  
 TYPE OF SUB-ELEMENT = P  
 DENSITY = 2.61700E-04  
 MODULUS OF ELASTICITY = 1.00000E+07  
 THICKNESS = 1.90000E-01  
 AREA = 7.91500E+01  
 POISSONS RATIO = 3.30000E-01  
 LENGTH = 0.  
 PRESSURE = 0.

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39 STIFFNESS REDUCTION REQUIRED = F  
RADIUS = 0.  
VOLUME = 0.  
SPEED OF SOUND IN ROOM MEDIUM = C.  
ADDED MASS = 0.

40 SUB-ELEMENT NUMBER = 18  
TYPE OF SUB-ELEMENT = P  
DENSITY = 2.61700E-04  
MODULUS OF ELASTICITY = 1.00000E+07  
THICKNESS = 2.50000E-01  
AREA = 1.71800E+01  
POISSONS RATIO = 3.30000E-01  
LENGTH = 0.  
PRESSURE = 0.

41 STIFFNESS REDUCTION REQUIRED = F  
RADIUS = 0.  
VOLUME = 0.  
SPEED OF SOUND IN ROOM MEDIUM = C.  
ADDED MASS = 0.

42 SUB-ELEMENT NUMBER = 19  
TYPE OF SUB-ELEMENT = P  
DENSITY = 2.61700E-04  
MODULUS OF ELASTICITY = 1.00000E+07  
THICKNESS = 2.00000E-01  
AREA = 1.55250E+01  
POISSONS RATIO = 3.30000E-01  
LENGTH = 0.  
PRESSURE = 0.

43 STIFFNESS REDUCTION REQUIRED = F  
RADIUS = 0.  
VOLUME = 0.  
SPEED OF SOUND IN ROOM MEDIUM = C.  
ADDED MASS = 0.

44 SUB-ELEMENT NUMBER = 20  
TYPE OF SUB-ELEMENT = P  
DENSITY = 2.61700E-04  
MODULUS OF ELASTICITY = 1.00000E+07  
THICKNESS = 1.90000E-01  
AREA = 1.16476E+03  
POISSONS RATIO = 3.30000E-01  
LENGTH = 0.  
PRESSURE = 0.

45 STIFFNESS REDUCTION REQUIRED = T  
RADIUS = 0.  
VOLUME = 0.  
SPEED OF SOUND IN ROOM MEDIUM = C.  
ADDED MASS = 0.

46 ELEMENT NUMBER = 2  
NUMBER OF SUB-ELEMENTS = 4  
TYPE OF EXCITATION =  
TYPE OF MECHANICAL INPUT =  
DAMPING = 1.00000E-02  
SLOPE = -8.30480E-01  
STARTING FREQUENCY = 2.50000E+02

47 SUB-ELEMENT NUMBER = 1  
TYPE OF SUB-ELEMENT = P  
DENSITY = 2.61700E-04  
MODULUS OF ELASTICITY = 1.00000E+07  
THICKNESS = 1.90000E-01  
AREA = 3.54160E+02  
POISSONS RATIO = 3.30000E-01  
LENGTH = 0.  
PRESSURE = 0.

48 STIFFNESS REDUCTION REQUIRED = F  
RADIUS = 0.  
VOLUME = 0.

SPEED OF SOUND IN ROOM MEDIUM = C.  
 ADDED MASS = 8.60000E-03  
 SUB-ELEMENT NUMBER = 2  
 TYPE OF SUB-ELEMENT = P  
 DENSITY = 2.61700E-04  
 MODULUS OF ELASTICITY = 1.00000E+07  
 THICKNESS = 1.50000E-01  
 AREA = 7.36000E+02  
 POISSONS RATIO = 3.30000E-01  
 LENGTH = 0.  
 PRESSURE = 0.  
 STIFFNESS REDUCTION REQUIRED = F  
 RADIUS = 0.  
 VOLUME = 0.  
 SPEED OF SOUND IN ROOM MEDIUM = C.  
 ADDED MASS = 0.  
 SUB-ELEMENT NUMBER = 3  
 TYPE OF SUB-ELEMENT = P  
 DENSITY = 2.61700E-04  
 MODULUS OF ELASTICITY = 1.00000E+07  
 THICKNESS = 2.00000E-01  
 AREA = 1.98000E+02  
 POISSONS RATIO = 3.30000E-01  
 LENGTH = 0.  
 PRESSURE = 0.  
 STIFFNESS REDUCTION REQUIRED = F  
 RADIUS = 0.  
 VOLUME = 0.  
 SPEED OF SOUND IN ROOM MEDIUM = C.  
 ADDED MASS = 0.  
 SUB-ELEMENT NUMBER = 4  
 TYPE OF SUB-ELEMENT = P  
 DENSITY = 2.61700E-04  
 MODULUS OF ELASTICITY = 1.00000E+07  
 THICKNESS = 2.50000E-01  
 AREA = 1.07100E+02  
 POISSONS RATIO = 3.30000E-01  
 LENGTH = 0.  
 PRESSURE = 0.  
 STIFFNESS REDUCTION REQUIRED = F  
 RADIUS = 0.  
 VOLUME = 0.  
 SPEED OF SOUND IN ROOM MEDIUM = C.  
 ADDED MASS = 0.  
 ELEMENT NUMBER = 3  
 NUMBER OF SUB-ELEMENTS = 3  
 TYPE OF EXCITATION =  
 TYPE OF MECHANICAL INPUT =  
 DAMPING = 1.00000E-02  
 SLOPE = -8.30480E-01  
 STARTING FREQUENCY = 2.50000E+02  
 SUB-ELEMENT NUMBER = 1  
 TYPE OF SUB-ELEMENT = P  
 DENSITY = 2.61700E-04  
 MODULUS OF ELASTICITY = 1.00000E+07  
 THICKNESS = 6.50000E-01  
 AREA = 8.76300E+01  
 POISSONS RATIO = 3.30000E-01  
 LENGTH = 0.  
 PRESSURE = 0.  
 STIFFNESS REDUCTION REQUIRED = T  
 RADIUS = 0.  
 VOLUME = 0.  
 SPEED OF SOUND IN ROOM MEDIUM = C.  
 ADDED MASS = 1.44800E-01  
 SUB-ELEMENT NUMBER = 2



TYPE OF SUB-ELEMENT = P  
 DENSITY = 2.61700E-04  
 MODULUS OF ELASTICITY = 1.00000E+07  
 THICKNESS = 3.50000E-01  
 AREA = 9.16700E+01  
 POISSONS RATIO = 3.30000E-01  
 LENGTH = 0.  
 PRESSURE = 0.  
 STIFFNESS REDUCTION REQUIRED = T  
 RADIUS = 0.  
 VOLUME = 0.  
 SPEED OF SOUND IN ROOM MEDIUM = C.  
 ADDED MASS = 0.

SUB-ELEMENT NUMBER = 3  
 TYPE OF SUB-ELEMENT = P  
 DENSITY = 2.61700E-04  
 MODULUS OF ELASTICITY = 1.00000E+07  
 THICKNESS = 7.00000E-01  
 AREA = 6.21920E+02  
 POISSONS RATIO = 3.30000E-01  
 LENGTH = 0.  
 PRESSURE = 0.  
 STIFFNESS REDUCTION REQUIRED = T  
 RADIUS = 0.  
 VOLUME = 0.  
 SPEED OF SOUND IN ROOM MEDIUM = C.  
 ADDED MASS = 0.

ELEMENT NUMBER = 4  
 NUMBER OF SUB-ELEMENTS = 1  
 TYPE OF EXCITATION =  
 TYPE OF MECHANICAL INPUT =  
 DAMPING = 1.00000E-02  
 SLOPE = -8.30480E-01  
 STARTING FREQUENCY = 2.50000E+02

SUB-ELEMENT NUMBER = 1  
 TYPE OF SUB-ELEMENT = P  
 DENSITY = 2.61700E-04  
 MODULUS OF ELASTICITY = 1.00000E+07  
 THICKNESS = 2.50000E-01  
 AREA = 7.05440E+02  
 POISSONS RATIO = 3.30000E-01  
 LENGTH = 0.  
 PRESSURE = 0.  
 STIFFNESS REDUCTION REQUIRED = T  
 RADIUS = 0.  
 VOLUME = 0.  
 SPEED OF SOUND IN ROOM MEDIUM = C.  
 ADDED MASS = 2.27000E-02

ELEMENT NUMBER = 5  
 NUMBER OF SUB-ELEMENTS = 3  
 TYPE OF EXCITATION =  
 TYPE OF MECHANICAL INPUT =  
 DAMPING = 1.00000E-02  
 SLOPE = -8.30480E-01  
 STARTING FREQUENCY = 2.50000E+02

SUB-ELEMENT NUMBER = 1  
 TYPE OF SUB-ELEMENT = P  
 DENSITY = 2.61700E-04  
 MODULUS OF ELASTICITY = 1.00000E+07  
 THICKNESS = 6.50000E-01  
 AREA = 8.76300E+01  
 POISSONS RATIO = 3.30000E-01  
 LENGTH = 0.  
 PRESSURE = 0.  
 STIFFNESS REDUCTION REQUIRED = T  
 RADIUS = 0.

ORIGINAL PAGE IS  
OF POOR QUALITY

VOLUME = 0.  
SPEED OF SOUND IN ROOM MEDIUM = C.  
ADDED MASS = 1.58790E+00  
68 SUB-ELEMENT NUMBER = 2  
TYPE OF SUB-ELEMENT = P  
DENSITY = 2.61700E-04  
MODULUS OF ELASTICITY = 1.00000E+07  
THICKNESS = 3.50000E-01  
AREA = 1.09960E+02  
POISSONS RATIO = 3.30000E-01  
LENGTH = 0.  
PRESSURE = 0.  
69 STIFFNESS REDUCTION REQUIRED = T  
RADIUS = 0.  
VOLUME = 0.  
SPEED OF SOUND IN ROOM MEDIUM = C.  
ADDED MASS = 0.  
70 SUB-ELEMENT NUMBER = 3  
TYPE OF SUB-ELEMENT = P  
DENSITY = 2.61700E-04  
MODULUS OF ELASTICITY = 1.00000E+07  
THICKNESS = 7.00000E-01  
AREA = 6.71110E+02  
POISSONS RATIO = 3.30000E-01  
LENGTH = 0.  
PRESSURE = 0.  
71 STIFFNESS REDUCTION REQUIRED = T  
RADIUS = 0.  
VOLUME = 0.  
SPEED OF SOUND IN ROOM MEDIUM = C.  
ADDED MASS = 0.  
72 ELEMENT NUMBER = 6  
NUMBER OF SUB-ELEMENTS = 2  
TYPE OF EXCITATION =  
TYPE OF MECHANICAL INPUT =  
DAMPING = 1.00000E-02  
SLOPE = -8.30480E-01  
STARTING FREQUENCY = 2.50000E+02  
73 SUB-ELEMENT NUMBER = 1  
TYPE OF SUB-ELEMENT = P  
DENSITY = 2.61700E-04  
MODULUS OF ELASTICITY = 1.00000E+07  
THICKNESS = 5.00000E-01  
AREA = 7.26800E+02  
POISSONS RATIO = 3.30000E-01  
LENGTH = 0.  
PRESSURE = 0.  
74 STIFFNESS REDUCTION REQUIRED = T  
RADIUS = 0.  
VOLUME = 0.  
SPEED OF SOUND IN ROOM MEDIUM = C.  
ADDED MASS = 1.30320E+00  
75 SUB-ELEMENT NUMBER = 2  
TYPE OF SUB-ELEMENT = P  
DENSITY = 2.61700E-04  
MODULUS OF ELASTICITY = 1.00000E+07  
THICKNESS = 3.75000E-01  
AREA = 7.26800E+02  
POISSONS RATIO = 3.30000E-01  
LENGTH = 0.  
PRESSURE = 0.  
76 STIFFNESS REDUCTION REQUIRED = T  
RADIUS = 0.  
VOLUME = 0.  
SPEED OF SOUND IN ROOM MEDIUM = C.  
ADDED MASS = 0.

CENTER FREQ(HZ)	MODAL DENSITY - MODES/(RAD/SEC)			
	ELEMENT 1	ELEMENT 2	ELEMENT 3	ELEMENT 4
31.50	1.08892E+00	1.09014E-02	1.20975E-03	2.65614E-03
39.38	1.08892E+00	1.09014E-02	1.20975E-03	2.65614E-03
50.40	1.08892E+00	1.09014E-02	1.20975E-03	2.65614E-03
63.00	1.08892E+00	1.09014E-02	1.20975E-03	2.65614E-03
78.75	1.08892E+00	1.09014E-02	1.20975E-03	2.65614E-03
99.23	1.08892E+00	1.09014E-02	1.20975E-03	2.65614E-03
126.00	1.08892E+00	1.09014E-02	1.20975E-03	2.65614E-03
157.50	1.08892E+00	1.09014E-02	1.20975E-03	2.65614E-03
196.45	1.08892E+00	1.09014E-02	1.20975E-03	2.65614E-03
252.00	1.08892E+00	1.09014E-02	1.20975E-03	2.65614E-03
315.00	1.08892E+00	1.09014E-02	1.20975E-03	2.65614E-03
393.75	1.08892E+00	1.09014E-02	1.20975E-03	2.65614E-03
504.00	1.08892E+00	1.09014E-02	1.20975E-03	2.65614E-03
630.00	1.08892E+00	1.09014E-02	1.20975E-03	2.65614E-03
787.50	1.08892E+00	1.09014E-02	1.20975E-03	2.65614E-03
992.25	1.08892E+00	1.09014E-02	1.20975E-03	2.65614E-03
1260.00	1.08892E+00	1.09014E-02	1.20975E-03	2.65614E-03
1575.00	1.08892E+00	1.09014E-02	1.20975E-03	2.65614E-03
1964.50	1.08892E+00	1.09014E-02	1.20975E-03	2.65614E-03
2520.00	1.08892E+00	1.09014E-02	1.20975E-03	2.65614E-03
3150.00	1.08892E+00	1.09014E-02	1.20975E-03	2.65614E-03
3937.50	1.08892E+00	1.09014E-02	1.20975E-03	2.65614E-03
5040.00	1.08892E+00	1.09014E-02	1.20975E-03	2.65614E-03

CENTER FREQ(HZ)	MODAL DENSITY - MODES/(RAD/SEC)	
	ELEMENT 5	ELEMENT 6
31.50	1.32509E-03	3.19266E-03
39.38	1.32509E-03	3.19266E-03
50.40	1.32509E-03	3.19266E-03
63.00	1.32509E-03	3.19266E-03
78.75	1.32509E-03	3.19266E-03
99.23	1.32509E-03	3.19266E-03
126.00	1.32509E-03	3.19266E-03
157.50	1.32509E-03	3.19266E-03
196.45	1.32509E-03	3.19266E-03
252.00	1.32509E-03	3.19266E-03
315.00	1.32509E-03	3.19266E-03
393.75	1.32509E-03	3.19266E-03
504.00	1.32509E-03	3.19266E-03
630.00	1.32509E-03	3.19266E-03
787.50	1.32509E-03	3.19266E-03
992.25	1.32509E-03	3.19266E-03
1260.00	1.32509E-03	3.19266E-03
1575.00	1.32509E-03	3.19266E-03
1964.50	1.32509E-03	3.19266E-03
2520.00	1.32509E-03	3.19266E-03
3150.00	1.32509E-03	3.19266E-03
3937.50	1.32509E-03	3.19266E-03
5040.00	1.32509E-03	3.19266E-03

RECORD  
NUMBER

DATA READ FROM UNIT 3

77 FIRST ELEMENT = 1  
SECOND ELEMENT = 2  
TYPE OF JOINT = BJ  
NUMBER OF STOPS = 0

JOINT LENGTH = 7.06700E+01  
 THICKNESS OF FIRST ELEMENT = 2.50000E-01  
 THICKNESS OF SECOND ELEMENT = 1.90000E-01  
 ACOUSTIC SPACE DENSITY = 0.  
 BEAM LENGTH = 0.  
 INSERTION LOSS FACTOR = 1.00000E+02  
 76 FIRST ELEMENT = 1  
 SECOND ELEMENT = 3  
 TYPE OF JOINT = BJ  
 NUMBER OF SIDES = 0  
 JOINT LENGTH = 1.23840E+02  
 THICKNESS OF FIRST ELEMENT = 2.50000E-01  
 THICKNESS OF SECOND ELEMENT = 7.00000E-01  
 ACOUSTIC SPACE DENSITY = 0.  
 BEAM LENGTH = 0.  
 INSERTION LOSS FACTOR = 1.00000E+02  
 79 FIRST ELEMENT = 1  
 SECOND ELEMENT = 4  
 TYPE OF JOINT = BJ  
 NUMBER OF SIDES = 0  
 JOINT LENGTH = 4.00000E+00  
 THICKNESS OF FIRST ELEMENT = 2.50000E-01  
 THICKNESS OF SECOND ELEMENT = 2.50000E-01  
 ACOUSTIC SPACE DENSITY = 0.  
 BEAM LENGTH = 0.  
 INSERTION LOSS FACTOR = 1.00000E+02  
 80 FIRST ELEMENT = 4  
 SECOND ELEMENT = 3  
 TYPE OF JOINT = PP  
 NUMBER OF SIDES = 0  
 JOINT LENGTH = 1.87600E+01  
 THICKNESS OF FIRST ELEMENT = 7.00000E-01  
 THICKNESS OF SECOND ELEMENT = 2.50000E-01  
 ACOUSTIC SPACE DENSITY = 0.  
 BEAM LENGTH = 0.  
 INSERTION LOSS FACTOR = 0.  
 81 FIRST ELEMENT = 1  
 SECOND ELEMENT = 5  
 TYPE OF JOINT = BJ  
 NUMBER OF SIDES = 0  
 JOINT LENGTH = 1.08840E+02  
 THICKNESS OF FIRST ELEMENT = 2.50000E-01  
 THICKNESS OF SECOND ELEMENT = 7.00000E-01  
 ACOUSTIC SPACE DENSITY = 0.  
 BEAM LENGTH = 0.  
 INSERTION LOSS FACTOR = 1.00000E+02  
 82 FIRST ELEMENT = 1  
 SECOND ELEMENT = 6  
 TYPE OF JOINT = BJ  
 NUMBER OF SIDES = 0  
 JOINT LENGTH = 2.67500E+02  
 THICKNESS OF FIRST ELEMENT = 2.50000E-01  
 THICKNESS OF SECOND ELEMENT = 3.45000E-01  
 ACOUSTIC SPACE DENSITY = 0.  
 BEAM LENGTH = 0.  
 INSERTION LOSS FACTOR = 1.00000E+02